

PART 1083—CIVIL PENALTY ADJUSTMENTS

■ 1. The authority citation for part 1083 continues to read as follows:

Authority: 12 U.S.C. 2609(d); 12 U.S.C. 5113(d)(2); 12 U.S.C. 5565(c); 15 U.S.C. 1639e(k); 15 U.S.C. 1717a(a); 28 U.S.C. 2461 note.

■ 2. Section 1083.1 is revised to read as follows:

§ 1083.1 Adjustments of civil penalty amounts.

(a) The maximum amount of each civil penalty within the jurisdiction of the Consumer Financial Protection Bureau to impose is adjusted in

accordance with the Federal Civil Penalties Inflation Adjustment Act of 1990, as amended by the Debt Collection Improvement Act of 1996 and further amended by the Federal Civil Penalties Inflation Adjustment Act Improvements Act of 2015 (28 U.S.C. 2461 note), as follows:

U.S. Code citation	Civil penalty description	Adjusted maximum civil penalty amount
12 U.S.C. 5565(c)(2)(A)	Tier 1 penalty	\$5,639
12 U.S.C. 5565(c)(2)(B)	Tier 2 penalty	28,195
12 U.S.C. 5565(c)(2)(C)	Tier 3 penalty	1,127,799
15 U.S.C. 1717a(a)(2)	Per violation	1,964
15 U.S.C. 1717a(a)(2)	Annual cap	1,963,870
12 U.S.C. 2609(d)(1)	Per failure	92
12 U.S.C. 2609(d)(1)	Annual cap	184,767
12 U.S.C. 2609(d)(2)(A)	Per failure, where intentional	185
12 U.S.C. 5113(d)(2)	Per violation	28,474
15 U.S.C. 1639e(k)(1)	First violation	11,279
15 U.S.C. 1639e(k)(2)	Subsequent violations	22,556

(b) The adjustments in paragraph (a) of this section shall apply to civil penalties assessed after January 15, 2018, regardless of when the violation for which the penalty is assessed occurred.

Dated: January 4, 2018.

Mick Mulvaney,

Acting Director, Bureau of Consumer Financial Protection.

[FR Doc. 2018-00399 Filed 1-11-18; 8:45 am]

BILLING CODE 4810-AM-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 39**

[Docket No. FAA-2017-1244; Product Identifier 2013-NM-145-AD; Amendment 39-19152; AD 2018-01-11]

RIN 2120-AA64

Airworthiness Directives; Airbus Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: We are adopting a new airworthiness directive (AD) for certain Airbus Model A319-115 and A319-133 airplanes. This AD requires contacting the FAA to obtain instructions for addressing the unsafe condition on these products, and doing the actions specified in those instructions. This AD was prompted by a fire during a flight,

in the vicinity of the gaseous oxygen system (GOS) for passengers. We are issuing this AD to address the unsafe condition on these products.

DATES: This AD becomes effective January 29, 2018.

We must receive comments on this AD by February 26, 2018.

ADDRESSES: You may send comments, using the procedures found in 14 CFR 11.43 and 11.45, by any of the following methods:

- *Federal eRulemaking Portal:* Go to <http://www.regulations.gov>. Follow the instructions for submitting comments.

- *Fax:* 202-493-2251.

- *Mail:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

- *Hand Delivery:* U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Examining the AD Docket

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-1244; or in person at the Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The street address for

the Docket Operations office (telephone: 800-647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

FOR FURTHER INFORMATION CONTACT:

Sanjay Ralhan, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 1601 Lind Avenue SW, Renton, WA 98057-3356; telephone: 425-227-1405; fax: 425-227-1149.

SUPPLEMENTARY INFORMATION:**Discussion**

The European Aviation Safety Agency (EASA), which is the Technical Agent for the Member States of the European Union, has issued EASA AD 2014-0045, dated February 25, 2014; corrected March 4, 2014 (referred to after this as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for certain Airbus Model A319-115 and A319-133 airplanes. The MCAI states:

Following an ECAM [electronic centralized aircraft monitor] warning “CARGO SMOKE” during flight, the flight crew elected to divert and the aeroplane made an uneventful landing. The post-flight inspection evidenced a heavy fire in the vicinity of the Gaseous Oxygen System (GOS) for passengers, located close to the cargo area. The origin of the fire has not been clearly identified. After more investigation, Airbus determined that the current optional passenger GOS design, specific to A319 aeroplanes, is not robust enough to prevent further events of this kind.

This condition, if not detected and corrected, could lead to an uncontrolled fire, possibly resulting in loss of the aeroplane.

To address this potential unsafe condition, Airbus developed mod 153555 to improve

the (optional) A319 GOS for passengers and published Service Bulletin (SB) A320–35–1062 to provide that modification for in-service application.

Consequently, EASA issued AD 2013–0153 to require modification of the passenger GOS installation.

Since that [EASA] AD was issued, a mistake was identified in Airbus SB A320–35–1062 concerning the pressure relief valve installation and the SB has been corrected and revised accordingly with mod 155860.

For the reason described above, this [EASA] AD retains the requirements of EASA AD 2013–0153, which is superseded, but requires those actions to be done in accordance with the instructions of Airbus SB 35–1062 Revision 01.

This [EASA] AD has been republished to correct an error that referenced the wrong effective date for EASA AD 2013–0153.

You may examine the MCAI on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA–2017–1244.

FAA’s Determination and Requirements of This AD

This product has been approved by the aviation authority of another country, and is approved for operation in the United States. Pursuant to our

bilateral agreement with the State of Design Authority, we have been notified of the unsafe condition described in the MCAI. We are issuing this AD because we evaluated all pertinent information and determined the unsafe condition exists and is likely to exist or develop on other products of the same type design.

FAA’s Determination of the Effective Date

Since there are currently no domestic operators of this product, we find good cause that notice and opportunity for prior public comment are unnecessary. In addition, for the reason(s) stated above, we find that good cause exists for making this amendment effective in less than 30 days.

Comments Invited

This AD is a final rule that involves requirements affecting flight safety, and we did not precede it by notice and opportunity for public comment. We invite you to send any written relevant data, views, or arguments about this AD. Send your comments to an address listed under the **ADDRESSES** section.

ESTIMATED COSTS

Action	Labor cost	Parts cost	Cost per product
Passenger GOS modification	Up to 33 work-hours × \$85 per hour = \$2,805.	Up to \$30,782	Up to \$33,587
Placard modification	5 work-hours × \$85 per hour = \$425.	\$0	\$425
Pressure hose modification	8 work-hours × \$85 per hour = \$680.	\$9,690	\$10,370

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

This AD is issued in accordance with authority delegated by the Executive

Director, Aircraft Certification Service, as authorized by FAA Order 8000.51C. In accordance with that order, issuance of ADs is normally a function of the Compliance and Airworthiness Division, but during this transition period, the Executive Director has delegated the authority to issue ADs applicable to transport category airplanes to the Director of the System Oversight Division.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;

2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
3. Will not affect intrastate aviation in Alaska; and
4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

2018-01-11 Airbus: Amendment 39-19152; Docket No. FAA-2017-1244; Product Identifier 2013-NM-145-AD.

(a) Effective Date

This AD becomes effective January 29, 2018.

(b) Affected ADs

None.

(c) Applicability

This AD applies to Airbus Model A319-115 and A319-133 airplanes, certificated in any category, all manufacturer serial numbers, having received in production Airbus modification 33125 (installation of Gaseous Oxygen System (GOS) for passengers), except those on which Airbus modification 153555 and 155860 have been embodied in production.

(d) Subject

Air Transport Association (ATA) of America Code 35, Oxygen.

(e) Reason

This AD was prompted by a fire during a flight, in the vicinity of the GOS for passengers. We are issuing this AD to prevent an uncontrolled fire in the vicinity of the GOS for passengers, near the cargo area, which could result in loss of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Required Action(s)

Within 30 days after the effective date of this AD, request instructions from the Manager, International Section, Transport Standards Branch, FAA, to address the unsafe condition specified in paragraph (e) of this AD; and accomplish the actions at the times specified in, and in accordance with, those instructions. Guidance can be found in Mandatory Continuing Airworthiness Information (MCAI) European Aviation Safety Agency (EASA) AD 2014-0045, dated February 25, 2014; corrected March 4, 2014.

(h) Alternative Methods of Compliance (AMOCs)

The Manager, International Section, Transport Standards Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the International Section, send it to the attention of the person identified in paragraph (i)(2) of this AD. Information may be emailed to: 9-ANM-116-AMOC-REQUESTS@faa.gov. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal

inspector, the manager of the local flight standards district office/certificate holding district office.

(i) Related Information

(1) Refer to MCAI EASA AD 2014-0045, dated February 25, 2014; corrected March 4, 2014, for related information. You may examine the MCAI on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-1244.

(2) For more information about this AD, contact Sanjay Ralhan, Aerospace Engineer, International Section, Transport Standards Branch, FAA, 1601 Lind Avenue SW, Renton, WA 98057-3356; telephone: 425-227-1405; fax: 425-227-1149.

(j) Material Incorporated by Reference

None.

Issued in Renton, Washington, on January 2, 2018.

Michael Kaszycki,

Acting Director, System Oversight Division, Aircraft Certification Service.

[FR Doc. 2018-00343 Filed 1-11-18; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2017-0629; Product Identifier 2016-NM-184-AD; Amendment 39-19149; AD 2018-01-08]

RIN 2120-AA64

Airworthiness Directives; The Boeing Company Airplanes

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: We are adopting a new airworthiness directive (AD) for all The Boeing Company Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. This AD was prompted by reports of fatigue cracking in the frame outboard chord and in the radius of the auxiliary chord at a certain area. This AD requires inspections to detect this cracking, and corrective action if necessary. We are issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective February 16, 2018.

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in this AD as of December 26, 2012 (77 FR 69747, November 21, 2012).

ADDRESSES: For service information identified in this final rule, contact Boeing Commercial Airplanes, Attention: Contractual & Data Services

(C&DS), 2600 Westminister Blvd., MC 110-SK57, Seal Beach, CA 90740; telephone 562-797-1717; internet <https://www.myboeingfleet.com>. You may view this service information at the FAA, Transport Standards Branch, 1601 Lind Avenue SW, Renton, WA. For information on the availability of this material at the FAA, call 425-227-1221. It is also available on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0629.

Examining the AD Docket

You may examine the AD docket on the internet at <http://www.regulations.gov> by searching for and locating Docket No. FAA-2017-0629; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (phone: 800-647-5527) is Docket Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE, Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT:

George Garrido, Aerospace Engineer, Airframe Section, FAA, Los Angeles ACO Branch, 3960 Paramount Boulevard, Lakewood, CA 90712-4137; phone: 562-627-5232; fax: 562-627-5210; email: george.garrido@faa.gov.

SUPPLEMENTARY INFORMATION:

Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to all The Boeing Company Model 737-100, -200, -200C, -300, -400, and -500 series airplanes. The NPRM published in the **Federal Register** on June 30, 2017 (82 FR 29792). The NPRM was prompted by reports of fatigue cracking in the frame outboard chord and in the radius of the auxiliary chord at a certain area. The NPRM proposed to require inspections to detect this cracking, and corrective action if necessary. We are issuing this AD to detect and correct fatigue cracking of the outboard and auxiliary chords, which could result in reduced structural integrity of the outboard chord and consequent rapid decompression of the airplane.

Comments

We gave the public the opportunity to participate in developing this final rule. The following presents the comments