## **Rules and Regulations**

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## OFFICE OF MANAGEMENT AND BUDGET

### 2 CFR Part 200

#### Guidance for Reporting and Use of Information Concerning Recipient Integrity and Performance; Corrections

**AGENCY:** Executive Office of the President, Office of Management and Budget.

**ACTION:** Correcting amendments.

**SUMMARY:** The Office of Management and Budget (OMB) is correcting the final guidance that appeared in the Federal Register on July 22, 2015 (80 FR 43301). OMB is amending the guidance to make technical corrections where necessary. The final guidance is revised to reflect that due to the 14 day delay of the publically available information entered into the OMB-designated integrity and performance system accessible through SAM (currently the Federal Awardee Performance and Integrity Information System (FAPIIS)), Federal awarding agencies should review the non-public segment of the system when conducting their risk review as described in the guidance. This will ensure that Federal awarding agencies have the most current information available when completing the review.

**DATES:** *Effective date:* November 9, 2015.

The applicability date of the final guidance remains January 1, 2016. **FOR FURTHER INFORMATION CONTACT:** Rhea Hubbard, Office of Federal Financial Management, *rhubbard*@

omb.eop.gov, or via telephone at (202) 395–3993.

**SUPPLEMENTARY INFORMATION:** On July 22, 2015 (80 FR 43301), OMB issued a number of changes to Title 2 of the Code of Federal Regulations (2 CFR part 180 and 2 CFR part 200). These changes provided guidance to Federal agencies to implement Section 872 of the Duncan

Hunter National Defense Authorization Act for Fiscal Year 2009. As Section 872 required, OMB and the General Services Administration (GSA) have established an integrity and performance system that includes governmentwide data with specified information related to the integrity and performance of entities awarded Federal grants and contracts. This is the second set of corrections. The first set of corrections was published in the Federal Register on July 30, 2015 (80 FR 45395) to advance the effective date to July 30, 2015 and reflect that the applicability date is January 1, 2016. This set of corrections reflects that Federal awarding agencies should check the non-public segment of the OMB-designated integrity and performance system accessible through SAM when conducting their risk review.

#### List of Subjects in 2 CFR Part 200

Accounting, Auditing, Colleges and universities, State and local governments, Grant programs, Grants administration, Hospitals, Indians, Nonprofit organizations, Reporting and recordkeeping requirements.

#### Mark Reger,

Deputy Controller.

Under the authority of the Chief Financial Officer Act of 1990 (31 U.S.C. 503), the Office of Management and Budget amends 2 CFR part 200 by making the following correcting amendments:

## PART 200—UNIFORM ADMINISTRATIVE REQUIREMENTS, COST PRINCIPLES, AND AUDIT REQUIREMENTS FOR FEDERAL AWARDS

■ 1. The authority citation for part 200 continues to read as follows:

Authority: 31 U.S.C. 503.

## §200.205 [Amended]

■ 2. Amend § 200.205 paragraph (a)(2) by removing "publicly available information in" and adding, in its place "non-public segment of".

[FR Doc. 2015–28441 Filed 11–6–15; 8:45 am] BILLING CODE 3110–01–P Federal Register Vol. 80, No. 216 Monday, November 9, 2015

## DEPARTMENT OF TRANSPORTATION

#### **Federal Aviation Administration**

## 14 CFR Part 39

[Docket No. FAA-2015-3969; Directorate Identifier 2014-SW-010-AD; Amendment 39-18318; AD 2015-23-02]

#### RIN 2120-AA64

# Airworthiness Directives; Agusta S.p.A. Helicopters

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT). **ACTION:** Final rule; request for comments.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for Agusta S.p.A. (Agusta) Model AB412 helicopters. This AD requires inspecting the filters installed on the pressure lines of utility hydraulic systems for metal particles. This AD is prompted by a report of a pump failure on the hydraulic external hoist caused by metal particles. These actions are intended to detect metal particles in the filter of the pressure line and prevent the pumps' failure, which could lead to a hoisting accident and injury to persons.

**DATES:** This AD becomes effective November 24, 2015.

We must receive comments on this AD by January 8, 2016.

**ADDRESSES:** You may send comments by any of the following methods:

• *Federal eRulemaking Docket:* Go to *http://www.regulations.gov.* Follow the online instructions for sending your comments electronically.

• Fax: 202-493-2251.

• *Mail:* Send comments to the U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12–140, 1200 New Jersey Avenue SE., Washington, DC 20590–0001.

• *Hand Delivery:* Deliver to the "Mail" address between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

#### **Examining the AD Docket**

You may examine the AD docket on the Internet at *http:// www.regulations.gov* by searching for locating Docket No. FAA–2015–3969 or in person at the Docket Operations Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the European Aviation Safety Agency (EASA) AD, and other information. The street address for the Docket Operations Office (telephone 800–647–5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

For service information identified in this AD, contact AgustaWestland, Product Support Engineering, Via del Gregge, 100, 21015 Lonate Pozzolo (VA) Italy, ATTN: Maurizio D'Angelo; telephone 39–0331–664757; fax 39– 0331–664680; or at *http:// www.agustawestland.com/technicalbulletins.* You may review the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N–321, Fort Worth, TX 76177.

FOR FURTHER INFORMATION CONTACT: Matt Wilbanks, Aviation Safety Engineer, Regulations and Policy Group, Rotorcraft Directorate, FAA, 10101 Hillwood Pkwy., Fort Worth, TX 76177; email *matt.wilbanks@faa.gov.* 

## SUPPLEMENTARY INFORMATION:

#### **Comments Invited**

This AD is a final rule that involves requirements affecting flight safety, and we did not provide you with notice and an opportunity to provide your comments prior to it becoming effective. However, we invite you to participate in this rulemaking by submitting written comments, data, or views. We also invite comments relating to the economic, environmental, energy, or federalism impacts that resulted from adopting this AD. The most helpful comments reference a specific portion of the AD, explain the reason for any recommended change, and include supporting data. To ensure the docket does not contain duplicate comments, commenters should send only one copy of written comments, or if comments are filed electronically, commenters should submit them only one time. We will file in the docket all comments that we receive, as well as a report summarizing each substantive public contact with FAA personnel concerning this rulemaking during the comment period. We will consider all the comments we receive and may conduct additional rulemaking based on those comments.

### Discussion

EASA, which is the Technical Agent for the Member States of the European Union, issued EASA AD No. 2014– 0063–E, dated March 12, 2014, to correct an unsafe condition for Agusta Model AB412 helicopters. EASA advises that during a hoist operation, a pressure line pump part number (P/N) 1–8072 Rev. A failed on a Model AB412 helicopter. Preliminary investigations reveal that unusual wear of an internal subcomponent generated metal particles, which caused the pump to fail, EASA advises. The root cause of this wear has not been determined.

This condition, if not detected and corrected, could lead to future pump failures, resulting in hoisting accidents and injuries. As a result, EASA requires repetitive inspections of the pump's filter for metal particles and replacement of the pressure line pump if particles exist. EASA advises that its AD is an interim action and that further AD action may follow, depending on the outcome of the investigations.

## **FAA's Determination**

These helicopters have been approved by the aviation authority of Italy and are approved for operation in the United States. Pursuant to our bilateral agreement with Italy, EASA, its technical representative, has notified us of the unsafe condition described in the EASA AD. We are issuing this AD because we evaluated all information provided by EASA and determined the unsafe condition exists and is likely to exist or develop on other helicopters of these same type designs.

#### **Related Service Information**

AgustaWestland has issued Bollettino Tecnico No. 412–140, dated March 11, 2014 (BT), for Model AB412 helicopters with a hydraulic external hoist P/N BL10300–60 installed. The BT notes that Agusta received a report that pump P/N 1-8072 Rev. A failed during a hoist operation on a Model AB412 helicopter due to metal particles generated by unusual wear of an internal subcomponent. The BT calls for inspecting the filter installed on the external hoist's pressure line for metal particles. The BT notes that an investigation is underway to determine the failure's root causes and that the BT could be updated.

### **AD Requirements**

This AD requires, within 10 hours time-in-service (TIS) and thereafter at intervals not to exceed 25 hours TIS, inspecting to determine whether metal particles are in the filter installed on the pressure line of the utility hydraulic system. If there are any metal particles, this AD requires, before the next flight, flushing the utility hydraulic system, replacing the filter with an airworthy filter, and replacing the pressure line pump with an airworthy pressure line pump.

#### **Interim Action**

We consider this AD to be an interim action. The design approval holder is investigating the root cause for the unsafe condition identified in this AD. Once the investigation is completed, we might consider additional rulemaking.

#### **Costs of Compliance**

There are no costs of compliance with this AD because there are no helicopters with this type certificate on the U.S. Registry.

## FAA's Justification and Determination of the Effective Date

There are no helicopters with this type certificate are on the U.S. Registry. Therefore, we believe it is unlikely that we will receive any adverse comments or useful information about this AD from U.S. Operators.

Since an unsafe condition exists that requires the immediate adoption of this AD, we determined that notice and opportunity for public comment before issuing this AD are unnecessary because there are none of these helicopters on the U.S. Registry and that good cause exists for making this amendment effective in less than 30 days.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. "Subtitle VII: Aviation Programs," describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in "Subtitle VII, Part A, Subpart III, Section 44701: General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

## **Regulatory Findings**

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed, I certify that this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;

2. Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);

3. Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction; and

4. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### §39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):

**2015–23–02** Agusta S.p.A.: Amendment 39– 18318; Docket No. FAA–2015–3969; Directorate Identifier 2014–SW–010–AD.

## (a) Applicability

This AD applies to Model AB412 helicopters with a hydraulic external hoist part number BL10300–60 installed, certificated in any category.

#### (b) Unsafe Condition

This AD defines the unsafe condition as failure of a hydraulic external hoist pressure line pump. This condition, if not detected and prevented, could result in loss of hydraulic pressure and subsequent injury to persons being lifted in the hoist.

#### (c) Effective Date

This AD becomes effective November 24, 2015.

#### (d) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

### (e) Required Actions

Within 10 hours time-in-service (TIS) and thereafter at intervals not to exceed 25 hours TIS: (1) Inspect the hydraulic external hoist pressure line filter for metal particles.

(2) If there are any metal particles, before further flight, flush the utility hydraulic system, replace the pressure line pump, and replace the filter.

### (f) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Safety Management Group, FAA, may approve AMOCs for this AD. Send your proposal to: Matt Wilbanks, Aviation Safety Engineer, Regulations and Policy Group, Rotorcraft Directorate, FAA, 10101 Hillwood Pkwy., Fort Worth, TX 76177; email 9-asw-ftw-amoc-requests@ faa.gov.

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office, before operating any aircraft complying with this AD through an AMOC.

#### (g) Additional Information

(1) AgustaWestland Bollettino Tecnico No. 412-140, dated March 11, 2014, which is not incorporated by reference, contains additional information about the subject of this AD. For service information identified in this AD, contact AgustaWestland, Product Support Engineering, Via del Gregge, 100, 21015 Lonate Pozzolo (VA) Italy, ATTN: Maurizio D'Angelo; telephone 39-0331-664757; fax 39-0331-664680; or at http:// www.agustawestland.com/technicalbulletins. You may review a copy of the service information at the FAA, Office of the Regional Counsel, Southwest Region, 10101 Hillwood Pkwy., Room 6N-321, Fort Worth, TX 76177.

(2) The subject of this AD is addressed in European Aviation Safety Agency (EASA) AD No. 2014–0063–E, dated March 12, 2014. You may view the EASA AD on the Internet at *http://www.regulations.gov* by searching for and locating it in Docket No. FAA–2015– 3969.

#### (h) Subject

Joint Aircraft Service Component (JASC) Code: 2550, Cargo Compartments.

Issued in Fort Worth, Texas, on October 30, 2015.

#### James A. Grigg,

Acting Assistant Manager, Rotorcraft Directorate, Aircraft Certification Service. [FR Doc. 2015–28314 Filed 11–6–15; 8:45 am]

## BILLING CODE 4910-13-P

**DEPARTMENT OF TRANSPORTATION** 

#### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. FAA-2015-3224; Directorate Identifier 2015-CE-026-AD; Amendment 39-18290; AD 2015-20-11]

## RIN 2120-AA64

#### Airworthiness Directives; Schempp-Hirth Flugzeugbau GmbH Gliders

**AGENCY:** Federal Aviation Administration (FAA), DOT. **ACTION:** Final rule; correction.

**SUMMARY:** The FAA is correcting an airworthiness directive (AD) that published in the **Federal Register**. That AD applies to certain Schempp-Hirth Flugzeugbau GmbH Models Duo Discus and Duo Discus T gliders. The Model Duo Discus gliders were incorrectly referenced as powered sailplanes in the Applicability section. This document corrects that error and refers to both models as just gliders as in previous ADs. In all other respects, the original document remains the same; however we are publishing the entire rule in the **Federal Register**.

**DATES:** This final rule is effective November 18, 2015.

ADDRESSES: You may examine the AD docket on the Internet at http:// www.regulations.gov by searching for and locating Docket No. FAA-2015-3224; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (phone: 800–647–5527) is Document Management Facility, U.S. Department of Transportation, Docket Operations, M–30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

FOR FURTHER INFORMATION CONTACT: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329–4165; fax: (816) 329–4090; email: *jim.rutherford*@ *faa.gov.* 

### SUPPLEMENTARY INFORMATION:

Airworthiness Directive 2015–20–11, Amendment 39–18290 (80 FR 61722, October 14, 2015), currently requires inspecting and replacing the airbrake bell crank and the airbrake drive funnels and inspecting the airbrake control system for proper clearance and making necessary adjustments.