

any helicopter unless it has been inspected as required by this AD.

**(f) Alternative Methods of Compliance (AMOCs)**

(1) The Manager, Safety Management Group, FAA, may approve AMOCs for this AD. Send your proposal to: Rao Edupuganti, Aviation Safety Engineer, Regulations and Policy Group, Rotorcraft Directorate, FAA, 2601 Meacham Blvd., Fort Worth, Texas 76137; telephone (817) 222-5110; email [rao.edupuganti@faa.gov](mailto:rao.edupuganti@faa.gov).

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office, before operating any aircraft complying with this AD through an AMOC.

**(g) Additional Information**

(1) Eurocopter Emergency Alert Service Bulletin No. 63A011, Revision 0, dated August 1, 2012, which is not incorporated by reference, contains additional information about the subject of this AD. For service information identified in this AD, contact American Eurocopter Corporation, 2701 N. Forum Drive, Grand Prairie, TX 75052; telephone (972) 641-0000 or (800) 232-0323; fax (972) 641-3775; or at <http://www.eurocopter.com/techpub>.

You may review a copy of the service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth Texas 76137.

(2) The subject of this AD is addressed in European Aviation Safety Agency AD No. 2012-0144-E, dated August 1, 2012.

**(h) Subject**

Joint Aircraft Service Component (JASC)  
Code: 6320: Main Rotor Gearbox.

Issued in Fort Worth, Texas, on October 15, 2012.

**Lance T. Gant,**

*Acting Manager, Rotorcraft Directorate,  
Aircraft Certification Service.*

[FR Doc. 2012-25894 Filed 10-22-12; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

**[Docket No. FAA-2012-0342; Directorate Identifier 2011-SW-028-AD; Amendment 39-17216; AD 2012-21-01]**

**RIN 2120-AA64**

**Airworthiness Directives; MD Helicopters, Inc.**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) to supersede an existing airworthiness directive for MD Helicopters, Inc. (MDHI), Model MD900 helicopters. The existing AD requires a visual inspection, and if necessary, an eddy current inspection of the main rotor lower hub assembly (lower hub) for a crack. If a crack exists, the AD requires replacing the lower hub with an airworthy lower hub before further flight. Because that AD was immediately effective, we declined to require certain long-term actions prior to public comment. This superseding AD will require the same inspections as the existing AD but will also require long-term recurring inspections and replacing the lower hub with an airworthy lower hub. We are issuing this AD to detect a crack in the lower hub and prevent failure of the lower hub and subsequent loss of control of the helicopter.

**DATES:** This AD is effective November 27, 2012.

The Director of the Federal Register approved the incorporation by reference of certain documents listed in this AD as of November 27, 2012.

**ADDRESSES:** For service information identified in this AD, contact MD Helicopters Inc., Attn: Customer Support Division, 4555 E. McDowell Rd., Mail Stop M615, Mesa, AZ 85215-9734, telephone 1-800-388-3378, fax 480-346-6813, or at <http://www.mdhelicopters.com>. You may review a copy of the referenced service information at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth Texas 76137.

**Examining the AD Docket:** You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the Docket Operations Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, any incorporated-by-reference service information, the economic evaluation, any comments received, and other information. The street address for the Docket Operations Office (phone: 800-647-5527) is U.S. Department of Transportation, Docket Operations Office, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Eric Schrieber, Aviation Safety Engineer, Los Angeles Aircraft Certification Office, Transport Airplane Directorate, FAA, 3960 Paramount Blvd., Lakewood, CA 90712; telephone (562) 627-5348; email [eric.schrieber@faa.gov](mailto:eric.schrieber@faa.gov).

**SUPPLEMENTARY INFORMATION:**

**Discussion**

On March 29, 2012, at 77 FR 18963, the **Federal Register** published our notice of proposed rulemaking (NPRM), which proposed to amend 14 CFR part 39 to include an AD that would apply to MDHI Model MD900 helicopters. That NPRM proposed to supersede an existing AD to require the same inspections as the existing AD but also require recurring inspections and replacing the lower hub with an airworthy lower hub. The proposed requirements were intended to detect a crack in the lower hub and prevent failure of the lower hub and subsequent loss of control of the helicopter.

**Comments**

We gave the public the opportunity to participate in developing this AD, but we received no comments on the NPRM.

**FAA's Determination**

We have reviewed the relevant information and determined that an unsafe condition exists and is likely to exist or develop on other products of these same type designs and that air safety and the public interest require adopting the AD requirements as proposed.

**Related Service Information**

We reviewed MDHI Service Bulletin SB900-117, dated January 14, 2011 (SB). The SB specifies an initial 100-hour and recurring 300-hour visual and eddy current inspections of the lower hub for a crack and, if a crack exists, replacement of the lower hub with an airworthy lower hub. The SB requires the inspections at the stated intervals or during the next annual inspection, whichever occurs first. The SB also specifies replacing the lower hub within three years.

**Costs of Compliance**

We estimate that this AD will affect 12 helicopters of U.S. registry.

We estimate the following costs to comply with this AD:

- Visually inspecting the hub. We estimate that will take one work-hour at \$85 an hour, for a total cost per helicopter of \$85 and a total cost of \$1,020 for the fleet.

- Eddy current inspecting the lower hub. We estimate that will take one work-hour at \$85 an hour, for a total cost per helicopter of \$85 and a total cost of \$1,020 for the fleet.

- Replacing the lower hub. We estimate that will take 11 work-hours at \$85 an hour for a total labor cost of \$935, and that parts will cost \$12,480 per hub, for a total cost of \$13,415 per

helicopter and a total cost of \$160,980 for the fleet.

- The costs for U.S. operators will total \$163,020, assuming that the lower hubs for the entire fleet get replaced.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

#### Regulatory Findings

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

*For the reasons discussed above, I certify that this AD:*

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979);
- (3) Will not affect intrastate aviation in Alaska to the extent that it justifies making a regulatory distinction; and
- (4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared an economic evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by removing airworthiness directive (AD) 2011–14–05, Amendment 39–16740 (76 FR 41662, July 15, 2011), and adding the following new AD:

#### 2012–21–01 MD HELICOPTERS, Inc.

(MDHI): Amendment 39–17216; Docket No. FAA–2012–0342; Directorate Identifier 2011–SW–028–AD.

#### (a) Applicability

This AD applies to MDHI Model MD900 helicopters with main rotor lower hub assembly (lower hub) part number 900R2101008–107, with serial numbers beginning with 5009, certificated in any category.

#### (b) Unsafe Condition

This AD defines the unsafe condition as a crack in the main rotor lower hub assembly (lower hub). This condition could result in failure of the lower hub and subsequent loss of control of the helicopter.

#### (c) Other Affected ADs

This AD supersedes AD 2011–14–05, amendment 39–16740 (76 FR 41662, July 15, 2011).

#### (d) Effective Date

This AD becomes effective November 27, 2012.

#### (e) Compliance

You are responsible for performing each action required by this AD within the specified compliance time unless it has already been accomplished prior to that time.

#### (f) Required Actions

(1) Within 100 hours time-in-service (TIS) or during the next annual inspection, whichever occurs first, unless done within the last 200 hours TIS, and thereafter at intervals not to exceed 300 hours TIS or during the next annual inspection, whichever occurs first:

(i) Visually inspect the sides and bottom of the area between the arms for the centering bearing and the areas adjacent to the bushings of the lower hub assembly for a crack. If there is a crack, before further flight, replace the lower hub with an airworthy lower hub.

(ii) If the lower hub is not replaced as a result of the visual inspection required by paragraph (f)(1)(i) of this AD, eddy current inspect the lower hub for a crack by following the Accomplishment Instructions, paragraphs 2.A(2) through 2.A.(10), of MD Helicopters Inc. Service Bulletin SB900–117, dated January 14, 2011. If there is a crack, before further flight, replace the lower hub with an airworthy hub.

(2) The eddy current inspection required by paragraph (f)(1)(ii) of this AD must be

done by a Level II technician with ASNT–TC–1A, CEN EN 4179, MIL–STD–410, NAS410, or equivalent certification in eddy current inspections. The technician must have done an eddy current inspection in the last 12 months.

(3) Within 3 years, replace the lower hub with an airworthy lower hub not included in the Applicability section of this AD. This replacement is terminating action for the requirements of this AD.

#### (g) Alternative Methods of Compliance (AMOCs)

(1) The Manager, Los Angeles Aircraft Certification Office (LAACO), FAA, may approve AMOCs for this AD. Send your proposal to: Eric Schrieber, Aviation Safety Engineer, Los Angeles Aircraft Certification Office, Transport Airplane Directorate, FAA, 3960 Paramount Blvd., Lakewood, CA 90712; telephone (562) 627–5348; email [eric.schrieber@faa.gov](mailto:eric.schrieber@faa.gov).

(2) For operations conducted under a 14 CFR part 119 operating certificate or under 14 CFR part 91, subpart K, we suggest that you notify your principal inspector, or lacking a principal inspector, the manager of the local flight standards district office or certificate holding district office before operating any aircraft complying with this AD through an AMOC.

#### (h) Subject

Joint Aircraft Service Component (JASC) Code: 2597, Equipment/furnishing system wiring.

#### (i) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.

(i) MD Helicopters Inc. Service Bulletin SB900–117, dated January 14, 2011.

(ii) Reserved.

(3) For service information identified in this AD, contact MD Helicopters Inc., Attn: Customer Support Division, 4555 E. McDowell Rd., Mail Stop M615, Mesa, AZ 85215–9734, telephone 1–800–388–3378, fax 480–346–6813, or at <http://www.mdhelicopters.com>.

(4) You may view this service information at FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas 76137. For information on the availability of this material at the FAA, call (817) 222–5110.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741–6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Fort Worth, Texas, on October 5, 2012.

**Kim Smith,**

*Manager, Rotorcraft Directorate, Aircraft Certification Service.*

[FR Doc. 2012-25387 Filed 10-22-12; 8:45 am]

BILLING CODE 4910-13-P

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2010-0856; Directorate Identifier 2010-NM-117-AD; Amendment 39-17224; AD 2012-21-08]

RIN 2120-AA64

#### Airworthiness Directives; The Boeing Company Airplanes

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** We are superseding an existing airworthiness directive (AD) for certain The Boeing Company Model 737-600, -700, -700C, -800, and -900 series airplanes. That AD currently requires installing and testing an updated version of the operational program software (OPS) of the flight control computers (FCCs). This new AD requires an inspection for part numbers of the operational program software of the flight control computers, and corrective actions if necessary. This AD was prompted by reports of undetected erroneous output from a single radio altimeter channel, which resulted in premature autothrottle retard during approach. We are issuing this AD to detect and correct an unsafe condition associated with erroneous output from a radio altimeter channel, which could result in premature autothrottle landing flare retard and the loss of automatic speed control, and consequent loss of control of the airplane.

**DATES:** This AD is effective November 27, 2012.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of November 27, 2012.

The Director of the Federal Register approved the incorporation by reference of a certain other publication listed in this AD as of May 12, 2005 (70 FR 17603, April 7, 2005).

**ADDRESSES:** For service information identified in this AD, contact Boeing Commercial Airplanes, Attention: Data & Services Management, P.O. Box 3707, MC 2H-65, Seattle, WA 98124-2207; telephone 206-544-5000, extension 1;

fax 206-766-5680; Internet <https://www.myboeingfleet.com>. You may review copies of the referenced service information at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington. For information on the availability of this material at the FAA, call 425-227-1221.

#### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this AD, the regulatory evaluation, any comments received, and other information. The address for the Docket Office (phone: 800-647-5527) is Document Management Facility, U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Gregg Nesemeier, Senior Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office (ACO), 1601 Lind Avenue SW., Renton, Washington 98057-3356; phone: (425) 917-6479; fax: (425) 917-6590; email: [gregg.nesemeier@faa.gov](mailto:gregg.nesemeier@faa.gov).

#### SUPPLEMENTARY INFORMATION:

##### Discussion

We issued a supplemental Notice of Proposed Rulemaking (SNPRM) to amend 14 CFR part 39 to supersede airworthiness directive (AD) 2005-07-20, Amendment 39-14045 (70 FR 17603, April 7, 2005). That AD applied to the specified products. That SNPRM published in the **Federal Register** on May 30, 2012 (77 FR 31758). The original NPRM (75 FR 57885, September 23, 2010) proposed to require inspecting for part numbers of the OPS of the FCCs, and doing corrective actions if necessary. The SNPRM contained the same requirements as the original NPRM, but also proposed to supersede an existing AD to require new software.

##### Comments

We gave the public the opportunity to participate in developing this AD. The following presents the comments received on the SNPRM (77 FR 31758, May 30, 2012) and the FAA's response to each comment.

#### Support for the SNPRM (77 FR 31758, May 30, 2012)

Boeing supports the SNPRM (77 FR 31758, May 30, 2012).

#### Requests To Use Alternative Service Information

Europe Airpost and Southwest Airlines requested that we revise the SNPRM (77 FR 31758, May 30, 2012) to include Boeing Alert Service Bulletin 737-22A1224, dated May 18, 2012, as an alternative method of compliance (AMOC) in the SNPRM.

Europe Airpost explained that Boeing Alert Service Bulletin 737-22A1224, dated May 18, 2012, takes into account the latest version (and later versions) of the OPS part number (P/N) 2274-COL-AC1-07 (P6.0 version number) and OPS software of the FCC.

Southwest Airlines explained that Boeing Alert Service Bulletin 737-22A1224, dated May 18, 2012, requires installation of FCC software P/N 2274-COL-AC1-07, and that its latest Model 737-800 deliveries have FCC software P/N 2274-COL-AC1-07 installed. Southwest Airlines stated that it has begun installing this version of the FCC software into all its airplanes, using Boeing Alert Service Bulletin 737-22A1224, dated May 18, 2012. Southwest Airlines also reasoned that Boeing Alert Service Bulletin 737-22A1224, dated May 18, 2012, has already been approved as an AMOC for AD 2005-07-20, Amendment 39-14045 (70 FR 17603, April 7, 2005).

We agree with the commenters' requests. We agree to allow use of Boeing Alert Service Bulletin 737-22A1224, dated May 18, 2012, for the installation actions specified in paragraphs (h)(1)(ii) and (h)(2) of this AD, although the compliance time for this final rule remains the same as proposed: Within 3 months after the effective date of this AD. We have reformatted and revised paragraphs (h)(1) and (h)(2) of this final rule accordingly.

#### Request To Remove the Phrase "Fully Interchangeable"

Paragraph (h)(1) of the SNPRM (77 FR 31758, May 30, 2012) specified installation of certain software that is "fully interchangeable" with the software specified in table 2 of the Accomplishment Instructions of Boeing Alert Service Bulletin 737-22A1211, dated April 13, 2010. Southwest Airlines requested that we revise the SNPRM by removing the phrase "fully interchangeable" from paragraph (h)(1) of the SNPRM and allowing, as acceptable in paragraphs (h)(1) and (h)(2) of the SNPRM, "subsequent software versions that have been approved by the Boeing Commercial Airplanes Organization Designation Authorization (ODA) after April 13,