

by adding the following new airworthiness directive (AD):

2005–20–41 Boeing: Amendment 39–14338. Docket No. FAA–2005–20137; Directorate Identifier 2004–NM–96–AD.

Effective Date

(a) This AD becomes effective November 17, 2005.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Boeing Model 757–200, –200PF, and –300 series airplanes; certificated in any category; powered by Pratt & Whitney PW2000 series engines.

Unsafe Condition

(d) This AD was prompted by reports of damage and subsequent failure of the support brackets and associated fasteners for the hydraulic lines located internal to the upper fairing cavity of the nacelle struts. We are issuing this AD to prevent such failure, which, in conjunction with sparking of electrical wires, failure of seals that would allow flammable fluids to migrate to compartments with ignition sources, or overheating of the pneumatic ducts beyond auto-ignition temperatures, could result in an uncontained fire.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Repetitive Inspections

(f) Within 6,000 flight hours or 24 months after the effective date of this AD, whichever is first: Do a detailed inspection for loose or damaged components of the support brackets and associated fasteners for the hydraulic lines located in the nacelle struts by accomplishing all of the actions specified in Part 1, Part 2, and Part 3 of the Accomplishment Instructions of Boeing Service Bulletin 757–29–0064 (for Model 757–200 and –200PF series airplanes) or Boeing Service Bulletin 757–29–0065 (for Model 757–300 series airplanes), both dated February 29, 2004; as applicable. Repeat the inspection thereafter at intervals not to exceed 6,000 flight hours or 24 months, whichever is first.

Note 1: For the purposes of this AD, a detailed inspection is defined as: “An intensive visual examination of a specific structural area, system, installation, or assembly to detect damage, failure, or irregularity. Available lighting is normally supplemented with a direct source of good lighting at intensity deemed appropriate by the inspector. Inspection aids such as mirror, magnifying lenses, etc., may be used. Surface cleaning and elaborate access procedures may be required.”

Concurrent Service Bulletin

(g) Prior to or concurrently with the accomplishment of paragraph (f) of this AD: Accomplish all of the actions specified in the Accomplishment Instructions of Boeing

Service Bulletin 757–29–0043, dated June 21, 1990.

Note 2: The part numbers identified in Boeing Service Bulletins 757–29–0064 or 757–29–0065, both dated February 29, 2004; or Boeing Service Bulletin 757–29–0043, dated June 21, 1990; are acceptable configurations and fully comply with the AD requirements for the actions required by paragraphs (f) and (g) of this AD.

Related Investigative and Corrective Actions

(h) Except as required by paragraph (i) of this AD: If any loose or damaged part is found during any inspection required by paragraph (f) of this AD, before further flight, do all of the related investigative and corrective actions specified in Part 1 and Part 2 of the Accomplishment Instructions of Boeing Service Bulletin 757–29–0064, or Boeing Service Bulletin 757–29–0065, both dated February 29, 2004; as applicable.

Repair Information

(i) If any damage is found during any inspection required by this AD, and the service bulletin specifies contacting Boeing for appropriate action: Before further flight, repair per a method approved by the Manager, Seattle Aircraft Certification Office (ACO), FAA. For a repair method to be approved, the approval letter must specifically refer to this AD.

Note 3: There is no terminating action currently available for the repetitive inspections required by paragraph (f) of this AD.

Alternative Methods of Compliance (AMOCs)

(j)(1) The Manager, Seattle ACO, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

Material Incorporated by Reference

(k) You must use Boeing Service Bulletin 757–29–0064, dated February 29, 2004, or Boeing Service Bulletin 757–29–0065, dated February 29, 2004; and Boeing Service Bulletin 757–29–0043, dated June 21, 1990; as applicable, to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of these documents in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124–2207, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., Room PL–401, Nassif Building, Washington, DC; on the Internet at <http://dms.dot.gov>; or at the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741–6030, or go to http://www.archives.gov/federal_register/

[code_of_federal_regulations/ibr_locations.html](#).

Issued in Renton, Washington, on September 30, 2005.

Ali Bahrami,

Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 05–20264 Filed 10–12–05; 8:45 am]

BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2001–NE–50–AD; Amendment 39–14306; AD 2005–20–12]

RIN 2120–AA64

Airworthiness Directives; Dowty Aerospace Propellers Type R321/4–82–F/8, R324/4–82–F/9, R333/4–82–F/12, and R334/4–82–F/13 Propeller Assemblies

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; request for comments.

SUMMARY: The FAA is superseding an existing airworthiness directive (AD) for Type R321/4–82–F/8, R324/4–82–F/9, R333/4–82–F/12, and R334/4–82–F/13 propeller assemblies. That AD currently requires initial and repetitive ultrasonic inspections of propeller hubs, part number (P/N) 660709201. This AD requires the same initial and repetitive ultrasonic inspections, but reduces the initial and repetitive compliance times for Type R334/4–82–F/13 propeller assemblies when used on Construcciones Aeronauticas, S.A. (CASA) 212 airplanes. This AD results from a report of a hub separation on a CASA 212 airplane. We are issuing this AD to prevent propeller hub failure due to cracks in the hub, which could result in loss of control of the airplane.

DATES: Effective October 28, 2005. The Director of the Federal Register approved the incorporation by reference of certain publications listed in the regulations as of October 28, 2005. The Director of the Federal Register previously approved the incorporation by reference of certain publications as listed in the regulations as of July 27, 2004 (69 FR 34560, June 22, 2004).

We must receive any comments on this AD by December 12, 2005.

ADDRESSES:

Use one of the following addresses to comment on this AD:

- By mail: Federal Aviation Administration (FAA), New England

Region, Office of the Regional Counsel, Attention: Rules Docket No. 2001-NE-50-AD, 12 New England Executive Park, Burlington, MA 01803-5299.

- By fax: (781) 238-7055.

- By e-mail: 9-ane-adcomment@faa.gov.

You can get the service information referenced in this AD from Dowty Aerospace Propellers, Anson Business Park, Cheltenham Road East, Gloucester GL 29QN, UK; telephone 44 (0) 1452 716000; fax 44 (0) 1452 716001.

You may examine the AD docket at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA.

FOR FURTHER INFORMATION CONTACT:

Terry Fahr, Aerospace Engineer, Boston Aircraft Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone (781) 238-7116; fax (781) 238-7155.

SUPPLEMENTARY INFORMATION: June 10, 2004, the FAA issued AD 2004-13-01, Amendment 39-13681 (69 FR 34560, June 22, 2004). That AD requires initial and repetitive ultrasonic inspections of propeller hubs, P/N 660709201, installed on airplanes, and for hubs and propellers in storage, initial ultrasonic inspection of propeller hubs before placing in service. That AD was the result of the manufacturer's reevaluation of potential hub failure on Type R321/4-82-F/8, R324/4-82-F/9, R333/4-82-F/12, and R334/4-82-F/13 propeller assemblies. That condition, if not corrected, could result in propeller hub failure due to cracks in the hub, which could result in loss of control of the airplane.

Actions Since AD 2004-13-01 Was Issued

Since we issued AD 2004-13-01, the Civil Aviation Authority (CAA), which is the airworthiness authority for the United Kingdom (U.K.), recently notified the FAA that an unsafe condition might exist on Dowty Aerospace Propellers Type R334/4-82-F/13 propeller assemblies installed on CASA 212 airplanes. The CAA advises that they have received a report of a hub separation of a Type R334/4-82-F/13 propeller assembly installed on a CASA 212 airplane. This AD requires the same initial and repetitive inspections as specified in AD 2004-13-01, but reduces the compliance intervals for the initial and repetitive inspections on Type R334/4-82-F/13 propeller assemblies installed on CASA 212 airplanes. We intend the actions specified in this AD to prevent propeller hub failure due to cracks in the hub,

which could result in loss of control of the airplane.

Relevant Service Information

We have reviewed and approved the technical contents of Dowty Aerospace Propellers Alert Mandatory Service Bulletin (MSB) No. 61-1119, Revision 4, dated September 14, 2005, that specifies initial and repetitive ultrasonic inspections of the rear wall of the rear half of the propeller hub for cracks on Type R334/4-82-F/13 propeller assemblies. The CAA classified this service bulletin as mandatory and issued CAA UK AD No. G-2005-0027, dated September 8, 2005, to assure the airworthiness of these Dowty Aerospace Propellers in the U.K.

Differences Between This AD and the Manufacturer's Service Information

Although Appendix A of Alert MSB No. 61-1119, Revision 4, dated September 14, 2005, requires reporting the inspection data to Dowty Aerospace Propellers, this AD requires that you report the data to the Boston Aircraft Certification Office of the FAA. Also, the Accomplishment Instructions 3.A.(1) of Alert MSB No. 61-1119, Revision 4, dated September 14, 2005, allows you to use Appendix A or Appendix D of that MSB, this AD requires that you use Appendix A of that MSB.

Bilateral Airworthiness Agreement

This propeller model is manufactured in the U.K. and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Under this bilateral airworthiness agreement, the CAA has kept the FAA informed of the situation described above. We have examined the findings of the CAA, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

FAA's Determination and Requirements of This AD

The unsafe condition described previously is likely to exist or develop on other Dowty Aerospace Propellers Type R321/4-82-F/8, R324/4-82-F/9, R333/4-82-F/12, and R334/4-82-F/13 propeller assemblies of the same type design. We are issuing this AD to prevent propeller hub failure due to cracks in the hub, which could result in loss of control of the airplane. This AD requires:

- Within 10 flight hours (FH) time-in-service (TIS) or 20 days after the effective date of this AD, whichever occurs earlier, performing an initial ultrasonic inspection of the rear halves of propeller hubs P/N 660709201, that are installed in Type R334/4-82-F/13 propeller assemblies, and;

- Within 50 FH TIS or 60 days after the effective date of this AD, whichever occurs earlier, performing an initial ultrasonic inspection of the rear halves of propeller hubs P/N 660709201, that are installed in Type R321/4-82-F/8, R324/4-82-F/9, and R333/4-82-F/12 propeller assemblies, and;

- Within 300 FH time-since-last-inspection (TSLI) performing a repetitive ultrasonic inspection of the rear halves of propeller hubs P/N 660709201, that are installed in Type R334/4-82-F/13 propeller assemblies, and;

- Within 1,000 FH TSLI performing a repetitive ultrasonic inspection of the rear halves of propeller hubs P/N 660709201, that are installed in Type R321/4-82-F/8, R324/4-82-F/9, and R333/4-82-F/12 propeller assemblies, and;

- If not already done, performing an ultrasonic inspection of the rear halves of propeller hubs P/N 660709201, that are installed in Type R321/4-82-F/8, R324/4-82-F/9, R333/4-82-F/12, and R334/4-82-F/13 propeller assemblies that are in storage before installing the propeller assembly onto an airplane.

You must use the service information described previously to perform the actions required by this AD.

FAA's Determination of the Effective Date

Since an unsafe condition exists that requires the immediate adoption of this AD, we have found that notice and opportunity for public comment before issuing this AD are impracticable and that good cause exists for making this amendment effective in less than 30 days.

Comments Invited

This AD is a final rule that involves requirements affecting flight safety. We did not precede it by notice and an opportunity for public comment; however, we invite you to send us any written relevant data, views, or arguments regarding this AD. Send your comments to an address listed under **ADDRESSES**. Include "AD Docket No. 2001-NE-50-AD" in the subject line of your comments. If you want us to acknowledge receipt of your mailed comments, send us a self-addressed, stamped postcard with the docket number written on it; we will date-

stamp your postcard and mail it back to you. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify it. If a person contacts us verbally, and that contact relates to a substantive part of this AD, we will summarize the contact and place the summary in the docket. We will consider all comments received by the closing date and may amend the AD in light of those comments.

Examining the AD Docket

You may examine the AD Docket (including any comments and service information), by appointment, between 8 a.m. and 4:30 p.m., Monday through Friday, except Federal holidays. See **ADDRESSES** for the location.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD and placed it in the AD Docket. You may get a copy of this summary by sending a request to us at the address listed under **ADDRESSES**. Include "AD Docket No. 2001-NE-50-AD" in your request.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The FAA amends § 39.13 by removing Amendment 39-13681 (69 FR 34560, June 22, 2004), and by adding a new airworthiness directive, Amendment 39-14306, to read as follows:

2005-20-12 Dowty Aerospace Propellers:
Amendment 39-14306. Docket No. 2001-NE-50-AD.

Effective Date

(a) This airworthiness directive (AD) becomes effective October 28, 2005.

Affected ADs

(b) This AD supersedes AD 2004-13-01, Amendment 39-13681.

Applicability

(c) This AD applies to Dowty Aerospace Propellers Type R321/4-82-F/8, R324/4-82-F/9, R333/4-82-F/12, and R334/4-82-F/13 propeller assemblies with propeller hubs part number (P/N) 660709201. These propeller assemblies are installed on, but not limited to, Construcciones Aeronauticas, S.A. (CASA) 212, British Aerospace Regional Aircraft Jetstream Models 3101 and 3201, Fairchild Aircraft, Inc., Merlin IIC, and Merlin IVC/Metro III airplanes.

Unsafe Condition

(d) This AD results from a report of a hub separation on a CASA 212 airplane. We are issuing this AD to prevent propeller hub failure due to cracks in the hub, which could result in loss of control of the airplane.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified unless the actions have already been done.

Initial Ultrasonic Inspections

(f) Perform an initial ultrasonic inspection of the rear wall of the rear half of the propeller hub for cracks within the compliance time specified in the following Table 1. Use Appendix A of the applicable Dowty Mandatory Service Bulletin (MSB) listed in Table 1 of this AD.

TABLE 1.—APPLICABLE MSB FOR PROPELLER TYPE

Propeller assembly type	Initial inspection within the earlier of . . .	Repeat inspection within . . .	Applicable MSB
(1) R334/4-82-F/13	10 flight hours (FH) time-in-service (TIS) or 20 days after the effective date of this AD.	300 FH time-since-last-inspection (TSLI).	Alert MSB No. 61-1119, Revision 4, dated September 14, 2005.
(2) R321/4-82-F/8	50 FH TIS or 60 days after the effective date of this AD.	1,000 FH TSLI	MSB No. 61-1125, Revision 1, dated October 9, 2002.
(3) R324/4-82-F/9	50 FH TIS or 60 days after the effective date of this AD.	1,000 FH TSLI	MSB No. 61-1126, Revision 1, dated October 9, 2002.
(4) R333/4-82-F/12	50 FH TIS or 60 days after the effective date of this AD.	1,000 FH TSLI	MSB No. 61-1124, Revision 1, dated October 8, 2002.

(g) For hubs and propellers in storage, perform an initial ultrasonic inspection of the rear wall of the rear half of the propeller hub for cracks, before placing in service. Use

Appendix A of the applicable Dowty MSB listed in Table 1 of this AD.

(h) Propeller hubs, P/N 660709201, already inspected using a Dowty MSB listed in Table

1 or earlier issue of those MSBs, comply with paragraph (f) of this AD.

Repetitive Ultrasonic Inspections

(i) Thereafter, perform a repetitive ultrasonic inspection of the rear wall of the rear half of the propeller hub for cracks within the compliance time specified in Table 1 of this AD. Use Appendix A of the applicable Dowty Mandatory Service Bulletin (MSB) listed in Table 1 of this AD.

Inspection Reporting Requirements

(j) Within 10 days after each inspection, record the inspection data on a copy of Appendix B of the applicable MSB listed in Table 1 of this AD. Report the findings to the Manager, Boston Aircraft Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299. The Office of Management and Budget (OMB) approved the reporting requirements and assigned OMB control number 2120-0056.

Alternative Methods of Compliance

(k) The Manager, Boston Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

Special Flight Permits

(l) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be done.

Documents That Have Been Incorporated by Reference

(m) You must use the service information specified in Table 2 to perform the inspections required by this AD. The Director of the Federal Register approved the incorporation by reference of Dowty Alert Mandatory Service Bulletin (MSB) No. 61-

1119, Revision 4, Dated September 14, 2005, listed in Table 2 of this AD, in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. The Director of the Federal Register previously approved the incorporation by reference of MSB No. 61-1124, Revision 1, Dated October 8, 2002, and MSB No. 61-1125, Revision 1, Dated October 9, 2002, and MSB 61-1126, Revision 1, Dated October 9, 2002 (69 FR 34560, June 22, 2004). Contact Dowty Propellers, Anson Business Park, Cheltenham Road East, Gloucester GL 29QN, UK; telephone 44 (0) 1452 716000; fax 44 (0) 1452 716001 for a copy of this service information. You may review copies at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call 202-741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

TABLE 2.—INCORPORATION BY REFERENCE

Service Bulletin No.	Page	Revision	Date
Alert MSB No. 61-1119	All	4	September 14, 2005.
Appendix A	1	1	November 27, 2001.
	2	Original	November 1, 2001.
	3-6	1	November 27, 2001.
Appendix B	1	Original	November 1, 2001.
Appendix C	All	Original	November 27, 2001.
Appendix D	All	Original	December 6, 2001.
Total pages	30	
MSB No. 61-1124	1	1	October 8, 2002.
	2-3	Original	May 7, 2002.
Appendix A	All	Original	May 7, 2002.
Appendix B	All	Original	May 7, 2002.
Appendix C	All	Original	May 7, 2002.
Appendix D	All	Original	May 7, 2002.
Total pages	30	
MSB No. 61-1125	1	1	October 9, 2002.
	2-3	Original	May 7, 2002.
Appendix A	All	Original	May 7, 2002.
Appendix B	All	Original	May 7, 2002.
Appendix C	All	Original	May 7, 2002.
Appendix D	All	Original	May 7, 2002.
Total pages	30	
MSB No. 61-1126	1	1	October 9, 2002.
	2-3	Original	May 7, 2002.
Appendix A	All	Original	May 7, 2002.
Appendix B	All	Original	May 7, 2002.
Appendix C	All	Original	May 7, 2002.
Appendix D	All	Original	May 7, 2002.
Total pages	30	

Related Information

(n) United Kingdom (U.K.) Civil Aviation Authority (CAA) airworthiness directives No. G-2006-0027, dated September 8, 2005; CAA UK AD No. 009-05-2002, dated April 15, 2003; CAA UK AD No. 010-05-2002, dated April 15, 2003; and CAA UK AD No. 011-05-2002, dated April 15, 2003, also addresses the subject of this AD.

Issued in Burlington, Massachusetts, on September 26, 2005.

Francis A. Favara,

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.

[FR Doc. 05-20170 Filed 10-12-05; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 71**

[Docket No. FAA-2005-21873; **Airspace Docket No. 05-ACE-27**]

Modification of Class D and Class E Airspace; Salina Municipal Airport, KS; Correction

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Direct final rule; request for comments; correction.

SUMMARY: This action corrects errors in the legal description of Class D airspace in a direct final rule, request for comments that was published in the **Federal Register** on Friday July 29, 2005 (70 FR 43742).

DATES: This direct final rule is effective on 0901 UTC, October 27, 2005.

FOR FURTHER INFORMATION CONTACT: Brenda Mumper, Air Traffic Division, Airspace Branch, ACE-520A, DOT Regional Headquarters Building, Federal Aviation Administration, 901 Locust, Kansas City, MO 64106; telephone: (816) 329-2524.

SUPPLEMENTARY INFORMATION:**History**

Federal Register Document 2005-21873 published on Friday July 29, 2005 (70 FR 43742), modified Class D and Class E Airspace at Salina Municipal Airport, KS. The legal descriptions of these airspace areas used outdated and incorrect information. This action corrects those errors.

■ Accordingly, pursuant to the authority delegated to me, the errors in the legal descriptions of airspace at Salina Municipal Airport, KS as published in the **Federal Register** Friday July 29, 2005 (70 FR 43742), (FR Doc. 2005-21873), are corrected as follows:

§ 71.1 [Amended]

■ 2. The incorporation by reference in 14 CFR 71.1 of Federal Aviation Administration Order 7400.9N, dated September 1, 2005, and effective September 15, 2005, is amended as follows:

Paragraph 5000 Class D Airspace.

* * * * *

ACE KS D Salina, KS

Salina Municipal Airport, KS

(Lat. 38°47'27" N., long. 97°39'08" W.)

That airspace extending upward from the surface to and including 3,800 feet MSL within a 5.4-mile radius of Salina Municipal Airport. This Class D airspace area is effective during the specific dates and times established in advance by a Notice to Airmen. The effective date and time will thereafter be continuously published in the Airport/Facility Directory.

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Paragraph 6002 Class E Airspace Designated as Surface Areas.

* * * * *

ACE KS E2 Salina, KS

Salina Municipal Airport, KS

(Lat. 38°47'27" N., long. 97°39'08" W.)

Airmen. The effective date and time will thereafter be continuously published in the Airport/Facility Directory.

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Paragraph 6004 Class E Airspace Designated as an Extension to a Class D or Class E Surface Area.

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ACE KS E4 Salina, KS

Salina Municipal Airport, KS

(Lat. 38°47'27" N., long. 97°39'08" W.)

Salina VORTAC

(Lat. 38°55'31" N., long. 97°37'17" W.)

That airspace extending upward surface within 1.5 miles each side of the Salina VORTAC 190° radial extending from the 5.4-mile radius of Salina Municipal Airport to 2 miles south of the VORTAC.

* * * * *

Paragraph 6005 Class E airspace areas extending upward from 700 feet or more above the surface of the earth.

* * * * *

ACE KS E5 Salina, KS

Salina Municipal Airport, KS

(Lat. 38°47'27" N., long. 97°39'08" W.)

That airspace extending upward from 700 feet above the surface within an 8.4-mile radius of Salina Municipal Airport.

Issued in Kansas City, MO, on September 20, 2005.

Elizabeth S. Wallis,

Acting Area Director, Western Flight Services Operations.

[FR Doc. 05-20462 Filed 10-12-05; 8:45 am]

BILLING CODE 4910-13-M

DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 71**

[Docket FAA 2005-21523; **Airspace Docket No. 05-AWP-07**]

Establishment of Class E3 Airspace, Riverside March Field, CA

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule.

SUMMARY: This action establishes a Class E3 airspace area at Riverside March Field, CA.

EFFECTIVE DATE: 0901 UTC December 22, 2005.

FOR FURTHER INFORMATION CONTACT:

Debra Trindle, Western Terminal Service Area, Airspace Specialist, AWP-520, Federal Aviation Administration, 15000 Aviation Boulevard, Lawndale, California 90261, telephone (310) 725-6613.

SUPPLEMENTARY INFORMATION:**History**

On August 2, 2005, the FAA proposed to establish Class E3 airspace at Riverside March Field, CA. Class E3 airspace areas are designated as arrival extensions to a Class C surface area. Class E arrival extensions are primarily designated to provide additional controlled airspace ancillary to a surface area to protect instrument operations for the primary airport, without imposing additional communications burdens on airspace users. This action is necessary at Riverside March Field to provide controlled airspace for Category E aircraft conducting circling maneuvers in conjunction with published Standard Instrument Approach Procedures. Generally, Category E aircraft are very large and/or high performance. These aircraft require additional airspace when conducting circling maneuvers. This action will establish the Class E3 airspace at Riverside March Field.

Interested parties were invited to participate in this rule making proceeding by submitting written comments on the proposal to the FAA. No comments were received. Class E3 airspace is published in Paragraph 6003 of FAA Order 7400.9N dated September 1, 2005, and effective September 16, 2005, which is incorporated by reference in 14 CFR 71.1. The Class E Surface Area airspace designation listed in this document would be published subsequently in this Order.