

Actions	Compliance	Procedures
<p>(5) Perform the installation in either (d)(5)(i) or (d)(5)(ii) of this AD for Model F406 airplanes:</p> <p>(i) Install improved design fuel boost pump (P/N 1C12-17 or FAA-approved equivalent P/N) and improved design wire harness (P/N 406 28 01 or FAA-approved equivalent P/N)</p> <p>(ii) Install the protective sleeving modification for boost pump lead wires that are not damaged or any lead wires that exhibit any chafing of the sleeve or outer jacket must be modified by installing a protective sleeving over the boost pump lead wires</p> <p>(iii) Installing both improved part numbers in each wing tank or protective sleeving modification for the existing fuel boost pump lead wires terminates the repetitive inspection requirements of paragraph (d)(2) of this AD</p>	<p>Within the next 400 hours TIS after June 24, 2003 (the effective date AD 2003-09-09), unless already accomplished.</p>	<p>Follow Reims/Cessna Caravan Service Bulletin No.: CAB02-8, dated June 3, 2002.</p>
<p>(6) Removing the following warnings for Model 441 airplanes after compliance with Cessna Conquest Service Bulletin No.: CQB02-1, Revision 2, dated October 7, 2002:</p> <p>(i) "PRIOR TO THE INITIAL INSPECTION: THE AIRPLANE SHOULD NOT BE OPERATED WITH LESS THAN 300 POUNDS OF FUEL IN EACH WING."</p> <p>(ii) "AFTER THE INITIAL INSPECTION: THE AIRPLANE SHOULD NOT BE OPERATED WHENEVER THE LEFT OR RIGHT LOW FUEL ANNUNCIATOR IS ILLUMINATED"</p>	<p>As of June 24, 2003 (the effective date AD 2003-09-09).</p>	<p>Not applicable.</p>
<p>(7) Install only improved design wire harnesses and fuel boost pumps as specified in paragraphs (d)(4) and (d)(5) of this AD.</p>	<p>As of June 24, 2003 (the effective date of AD 2003-09-09).</p>	<p>Not applicable.</p>

(e) *Can I comply with this AD in any other way?*

(1) You may use an alternative method of compliance or adjust the compliance time if:

(i) Your alternative method of compliance provides an equivalent level of safety; and

(ii) The Manager, Wichita Aircraft Certification Office (ACO), approves your alternative. Submit your request through an FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Wichita ACO.

(2) Alternative methods of compliance approved in accordance with AD 2003-09-09 or AD 2002-09-13 are approved as alternative methods of compliance for all inspection requirements of this AD. Regardless, you still must comply with the replacement requirements of this AD.

Note: This AD applies to each airplane identified in paragraph (a) of this AD, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (e) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if you have not eliminated the unsafe condition, specific actions you propose to address it.

(f) *Where can I get information about any already-approved alternative methods of compliance?* Contact Robert Adamson, Aerospace Engineer, FAA, Wichita Aircraft Certification Office, 1801 Airport Road, Room 100, Wichita, Kansas 67209; telephone: 316-946-4145; facsimile: 316-946-4107.

(g) *What if I need to fly the airplane to another location to comply with this AD?* The FAA can issue a special flight permit under §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate your airplane to a location where you can accomplish the requirements of this AD.

(h) *Are any service bulletins incorporated into this AD by reference?* Actions required by this AD must be done following Cessna Conquest Service Bulletin No.: CQB02-1, Revision 2, dated October 7, 2002; and Reims Cessna Caravan Service Bulletin No.: CAB02-8, dated June 3, 2002. The Director of the Federal Register previously approved this incorporation by reference under 5 U.S.C. 552(a) and 1 CFR part 51 on June 24, 2003 (68 FR 23186, May 1, 2003). You may get copies from Cessna Aircraft Company, Product Support, P.O. Box 7706, Wichita, Kansas 67277; telephone: (316) 517-5800; facsimile: (316) 942-9006. You may view copies at the FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri, or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

(i) *Does this AD action affect any existing AD actions?* This amendment revises AD 2002-09-13, Amendment 39-12746.

(j) *When does this amendment become effective?* This amendment becomes effective on January 22, 2004.

Issued in Kansas City, Missouri, on January 5, 2004.

Dorenda D. Baker,
Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-475 Filed 1-9-04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-NM-55-AD; Amendment 39-13429; AD 2004-01-15]

RIN 2120-AA64

Airworthiness Directives; McDonnell Douglas Model 717-200 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD),

applicable to certain McDonnell Douglas Model 717-200 airplanes, that requires repetitive inspections of the electric motor of the auxiliary hydraulic pump for electrical resistance, continuity, mechanical rotation, and associated wiring resistance/voltage; and corrective actions, if necessary. This action is necessary to prevent various failures of the electric motor of the auxiliary hydraulic pump and associated wiring, which could result in fire at the auxiliary hydraulic pump and consequent damage to the adjacent electrical equipment and/or structure. This action is intended to address the identified unsafe condition.

DATES: Effective February 17, 2004.

The incorporation by reference of a certain publication listed in the regulations is approved by the Director of the Federal Register as of February 17, 2004.

ADDRESSES: The service information referenced in this AD may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90846, Attention: Data and Service Management, Dept. C1-L5A (D800-0024). This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

FOR FURTHER INFORMATION CONTACT:

Albert Lam, Aerospace Engineer; Systems and Equipment Branch, ANM-130L, FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712-4137; telephone (562) 627-5346; fax (562) 627-5210.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain McDonnell Douglas Model 717-200 airplanes was published in the **Federal Register** on October 1, 2003 (68 FR 56594). That action proposed to require repetitive inspections of the electric motor of the auxiliary hydraulic pump for electrical resistance, continuity, mechanical rotation, and associated wiring resistance/voltage; and corrective actions, if necessary.

Comments

Interested persons have been afforded an opportunity to participate in the

making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Interim Action

This is considered to be interim action until final action is identified, at which time the FAA may consider further rulemaking.

Cost Impact

There are approximately 95 airplanes of the affected design in the worldwide fleet. The FAA estimates that 67 airplanes of U.S. registry will be affected by this AD, that it will take approximately 1 work hour per airplane to accomplish the required inspection, and that the average labor rate is \$65 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$4,355, or \$65 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) Is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has

been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. Section 39.13 is amended by adding the following new airworthiness directive:

2004-01-15 McDonnell Douglas:

Amendment 39-13429. Docket 2003-NM-55-AD.

Applicability: Model 717-200 airplanes, manufacturer's fuselage numbers 5002 through 5200 inclusive; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent various failures of electric motor of the auxiliary hydraulic pump and associated wiring, which could result in fire at the auxiliary hydraulic pump and consequent damage to the adjacent electrical equipment and/or structure, accomplish the following:

Service Bulletin References

(a) The term "service bulletin," as used in this AD, means the Accomplishment Instructions of Boeing Alert Service Bulletin 717-29A0005, dated July 31, 2002. Although the service bulletin referenced in this AD specifies to submit certain information to the manufacturer, this AD does not include such a requirement.

Initial Inspection and Testing

(b) Prior to the accumulation of 3,000 total flight hours, or within 12 months after the effective date of this AD, whichever occurs later, do an inspection of the electric motor of the auxiliary hydraulic pump for electrical resistance, continuity, mechanical rotation, and associated wiring resistance/voltage per the service bulletin.

Condition 1, No Failures: Repetitive Inspections

(c) If no failures are detected during any inspection required by paragraph (b) of this AD, repeat the inspection thereafter at intervals not to exceed 5,000 flight hours.

Condition 2, Failure of Any Pump Motor: Replacement and Repetitive Inspections

(d) If any pump motor fails during any inspection required by paragraph (b) of this AD, before further flight, replace the failed auxiliary hydraulic pump with a serviceable pump, per the service bulletin. Repeat the inspection required by paragraph (b) of this AD at intervals not to exceed 5,000 flight hours.

Condition 3, Failure of Any Wiring: Repair and Repetitive Inspection

(e) If any wiring fails during any inspection required by paragraph (b) of this AD, before further flight, troubleshoot and repair the failed wiring, per the service bulletin. Repeat the inspection at intervals not to exceed 5,000 flight hours.

Alternative Methods of Compliance

(f) In accordance with 14 CFR 39.19, the Manager, Los Angeles Aircraft Certification Office, FAA, is authorized to approve alternative methods of compliance (AMOCs) for this AD.

Incorporation by Reference

(g) The actions shall be done in accordance with Boeing Alert Service Bulletin 717–29A0005, dated July 31, 2002. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Boeing Commercial Aircraft Group, Long Beach Division, 3855 Lakewood Boulevard, Long Beach, California 90804, Attention: Data and Service Management, Dept. C1–L5A (D800–0024). Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Effective Date

(h) This amendment becomes effective on February 17, 2004.

Issued in Renton, Washington, on December 31, 2003.

Michael J. Kaszycki,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04–424 Filed 1–9–04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION**Federal Aviation Administration****14 CFR Part 71**

[Docket No. FAA–2003–16749; Airspace Docket No. 03–ACE–93]

Modification of Class E Airspace; Beloit, KS

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Direct final rule; request for comments.

SUMMARY: This action modifies the Class E airspace area at Beloit, KS by changing the name of the airport in the legal description from Beloit Municipal Airport to Moritz Memorial Airport. It also modifies the dimensions of this Class E airspace area. A review of controlled airspace for Moritz Memorial Airport indicates it does not comply with the criteria for 700 feet Above Ground Level (AGL) airspace required for diverse departures. The area is modified and enlarged to conform to the criteria in FAA Orders.

DATES: This direct final rule is effective on 0901 UTC, April 15, 2004. Comments for inclusion in the Rules Docket must be received on or before February 23, 2004.

ADDRESSES: Send comments on this proposal to the Docket Management System, U.S. Department of Transportation, Room Plaza 401, 400 Seventh Street, SW., Washington, DC 20590–0001. You must identify the docket number FAA–2003–16749/ Airspace Docket No. 03–ACE–93, at the beginning of your comments. You may also submit comments on the Internet at <http://dms.dot.gov>. You may review the public docket containing the proposal, any comments received, and any final disposition in person in the Dockets Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Office (telephone 1–800–647–5527) is on the plaza level of the Department of Transportation NASSIF Building at the above address.

FOR FURTHER INFORMATION CONTACT: Kathy Randolph, Air Traffic Division, Airspace Branch, ACE–520C, DOT Municipal Headquarters Building, Federal Aviation Administration, 901 Locust, Kansas City, MO 64106; telephone: (816) 329–2525.

SUPPLEMENTARY INFORMATION: This amendment to 14 CFR part 71 modifies the Class E airspace area extending upward from 700 feet above the surface at Beloit, KS. The name of the airport at Beloit, KS has been changed from Beloit Municipal Airport to Moritz Memorial Airport. An examination of controlled airspace for Moritz Memorial Airport reveals it does not meet the criteria for 700 feet AGL airspace required for diverse departures as specified in FAA Order 7400.2E, Procedures for Handling Airspace Matters. The criteria in FAA Order 7400.2E for an aircraft to each 1200 feet AGL is based on a standard climb gradient of 200 feet per mile plus the distance from the Airport Reference Point (ARP) to the end of the outermost

runway. Any fractional part of a mile is converted to the next higher tenth of a mile. This action changes the name of the airport in the legal description, increases the dimensions of the Class E airspace area from a 6-mile radius to a 6.5-mile radius of the airport and brings the legal description of the Beloit, KS Class E airspace area into compliance with FAA Order 7400.2E. This area will be depicted on appropriate aeronautical charts. Class E airspace areas extending upward from 700 feet or more above the surface of the earth are published in paragraph 6005 of FAA Order 7400.9L, dated September 2, 2003, and effective September 16, 2003, which is incorporated by reference in 14 CFR 71.1. The Class E airspace designation listed in this document will be published subsequently in the Order.

The Direct Final Rule Procedure

The FAA anticipates that this regulation will not result in adverse or negative comment and, therefore, is issuing it as a direct final rule. Previous actions of this nature have not been controversial and have not resulted in adverse comments or objections. Unless a written adverse or negative comment, or a written notice of intent to submit an adverse or negative comment is received within the comment period, the regulation will become effective on the date specified above. After the close of the comment period, the FAA will publish a document in the **Federal Register** indicating that no adverse or negative comments were received and confirming the date on which the final rule will become effective. If the FAA does receive, within the comment period, an adverse or negative comment, or written notice of intent to submit such a comment, a document withdrawing the direct final rule will be published in the **Federal Register**, and a notice of proposed rulemaking may be published with a new comment period.

Comments Invited

Interested parties are invited to participate in this rulemaking by submitting such written data, views, or arguments, as they may desire. Comments that provide the factual basis supporting the views and suggestions presented are particularly helpful in developing reasoned regulatory decisions on the proposal. Comments are specifically invited on the overall regulatory, aeronautical, economic, environmental, and energy-related aspects of the proposal. Communications should identify both docket numbers and be submitted in triplicate to the address listed above. Commenters wishing the FAA to