

## INITIAL INSPECTION AND CLEANING SCHEDULE—Continued

Engine hours time-since-new (TSN) or time-since-last-shop-visit (TSLSV)	Inspect and clean
(3) 4,000 hours or greater TSN or 4,000 hours or greater TSLSV if it can be confirmed that both the B-sump scavenge screens were cleaned and the B-sump and combustor frame (strut tubes) were removed from the engine and cleaned at that prior shop visit, or 1,000 hours or greater TSLSV if it can NOT be confirmed that both the B-sump and combustor frame (strut tubes) were removed from the engine and cleaned at that prior shop visit.	Within 500 hours time-in-service (TIS) after the effective date of this AD.

**Repetitive Inspections and Cleaning**

(b) Perform repetitive visual inspections and cleaning of scavenge screens, P/Ns 4047T95P01 and 5054T86G02, installed in the B-sump oil scavenge system, in accordance with Paragraphs 3A through 3B of the Accomplishment Instructions of GE ASB CF34-AL S/B 79-A0014, Revision 3, dated January 31, 2003; and ASB CF34-BJ S/B 79-A0015, Revision 3, dated January 31, 2003; and the following:

(1) At intervals not to exceed 200 hours time-since-last-inspection (TSLI), if no coke is found in screens during initial or any prior inspections, or

(2) At intervals not to exceed 100 hours TSLI, if coke is found in screens during initial or any prior inspections.

**Terminating Actions**

(c) Install new screenless fittings or fittings that have been reworked to remove the

screens, in the B-sump oil scavenge system, in accordance with GE ASB CF34-AL S/B 79-A0016, dated June 17, 2002; or ASB CF34-BJ S/B 79-A0017, dated June 17, 2002, and the following schedule:

(1) For engines with more than 4,000 hours TSN, within 500 hours TIS after the effective date of this AD, or within 1,000 hours TSLSV, whichever occurs first.

(2) For engines with less than or equal to 4,000 hours TSN, prior to 4,500 hours TSN.

This constitutes terminating action to the inspections required in paragraph (b) of this AD.

**Alternative Methods of Compliance**

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Engine Certification Office (ECO). Operators must submit their request through an appropriate

FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, ECO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the ECO.

**Special Flight Permits**

(e) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be done.

**Documents That Have Been Incorporated by Reference**

(f) The actions must be done in accordance with the following General Electric Company Alert Service Bulletins (ASB):

Document No.	Pages	Revision	Date
ASB CF34-AL S/B 79-A0014 Total Pages: 10	All	3	January 31, 2003.
ASB CF34-BJ S/B 79-A0015 Total Pages: 9	All	3	January 31, 2003.
ASB CF34-AL S/B 79-A0016 Total Pages: 12	All	Original	June 17, 2002.
ASB CF34-BJ S/B 79-A0017 Total Pages: 11	All	Original	June 17, 2002.

This incorporation by reference of General Electric Alert Service Bulletin (ASB) CF34-AL S/B 79-A0014, Revision 3, dated January 31, 2003; ASB CF34-BJ S/B 79-A0015, Revision 3, dated January 31, 2003; GE ASB CF34-AL S/B 79-A0016 and ASB CF34-BJ S/B 79-A0017, both dated June 17, 2002, was approved by the Director of the Federal Register on April 2, 2003, (68 FR 12086; March 18, 2003) in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from GE Aircraft Engines, 1000 Western Avenue, Lynn, MA 01910; Attention: CF34 Product Support Engineering, Mail Zone: 34017; telephone (781) 594-6323; fax (781) 594-0600. Copies may be inspected at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA; or at the Office of the Federal Register, 800 North Capitol Street, NW, suite 700, Washington, DC.

**Effective Date**

(g) This amendment becomes effective on November 18, 2003.

Issued in Burlington, Massachusetts, on October 6, 2003.

**Francis A. Favara,**

*Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.*

[FR Doc. 03-25864 Filed 10-10-03; 8:45 am]

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**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

[Docket No. 2002-CE-58-AD; Amendment 39-13335; AD 2003-21-01]

**RIN 2120-AA64**

**Airworthiness Directives; Univair Aircraft Corporation Models Alon A-2 and A2-A; ERCO 415-C, 415-CD, 415-D, 415-E, and 415-G; Forney F-1 and F-1A; and Mooney M10 Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule.

**SUMMARY:** The FAA supersedes Airworthiness Directive (AD) 94-18-04

R1, which currently applies to all Univair Aircraft Corporation (Univair) Models Alon A-2 and A2-A; ERCO 415-C, 415-CD, 415-D, 415-E, and 415-G; Forney F-1 and F-1A; and Mooney M10 airplanes. AD 94-18-04 R1 requires installing inspection openings in the outer wing panels, inspecting (one-time) the wing outer panel structural components for corrosion, and repairing any corroded wing outer panel structural component. Several reports of corrosion in the outer wing panels of the affected airplanes prompted that AD. This AD is the result of additional reports of corrosion on airplanes in compliance with AD 94-18-04 R1. This AD makes the inspection required in AD 94-18-04 R1 repetitive. We are issuing this AD to prevent wing damage caused by a corroded wing outer panel structural component, which, if not detected and corrected, could progress to the point of structural failure.

**DATES:** This AD becomes effective on December 1, 2003.

The Director of the Federal Register previously approved the incorporation by reference of Univair Aircraft Corporation Service Bulletin No. 29, Revision B, dated January 2, 1995, as of March 24, 1995 (60 FR 13626, March 14, 1995).

The Director of the Federal Register approved the incorporation by reference of Univair Aircraft Corporation Mandatory Service Bulletin No. 29, Revision C, dated July 8, 1999, as of December 1, 2003.

**ADDRESSES:** You may get the service information identified in this AD from Univair Aircraft Corporation, 2500 Himalaya Road, Aurora, Colorado 80011, telephone: (303) 375-8882; facsimile: (303) 375-8888.

You may view the AD docket at FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 2002-CE-58-AD, 901 Locust, Room 506, Kansas City, Missouri 64106. Office hours are 8 a.m. to 4 p.m., Monday through Friday, except Federal holidays.

**FOR FURTHER INFORMATION CONTACT:** Roger Caldwell, Aerospace Engineer, FAA, Denver Aircraft Certification Office, 26805 East 68th Avenue, Room 214, Denver, Colorado 80249-6361; telephone: (303) 342-1086; facsimile: (303) 342-1088.

**SUPPLEMENTARY INFORMATION:**

**Discussion**

Has FAA taken any action to this point? Maintenance inspection procedures did not allow for thorough inspection of the wing structure on Univair Models Alon A-2 and A2-A; ERCO 415-C, 415-CD, 415-D, 415-E, and 415-G; Forney F-1 and F-1A; and Mooney M10 airplanes. This caused us to issue AD 94-18-04, Amendment 39-9017 (59 FR 43727, August 25, 1994) to require installing inspection openings in the outer wing panels, inspecting (one-time) the wing outer panel structure for corrosion, and repairing any corrosion found.

After AD 94-18-04 was issued, Univair revised Service Bulletin No. 29 to the Revision B level. Univair Service Bulletin No. 29, Revision B, dated January 2, 1995, changed the dimension of one of the openings to position it symmetrically between two ribs; and clarified the dimensioning system utilized in placement of the inspection openings. Univair SB No. 29, Revision B, also presented further discussion of the service difficulties encountered on the referenced subject and clarified the intent of the preliminary inspection procedure that may be accomplished prior to the installation of the inspection openings.

This caused us to issue AD 94-18-04 R1, Amendment 39-9173 (60 FR 62321, March 14, 1995).

*What has happened since AD 94-18-04 R1 to initiate this action?* The FAA has received additional reports of corrosion damage in the wing outer panel structural components continuing to go undetected. Univair has revised Service Bulletin No. 29 to the Revision C level, dated July 8, 1999. This revision changes the one-time inspection of the wing outer panel structural components for corrosion to a repetitive inspection.

*What is the potential impact if FAA took no action?* This condition, if not detected and corrected, could cause damage to the wing outer panel structural component. Such damage could result in structural failure.

*Has FAA taken any action to this point?* We issued a proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an AD that would apply to all Univair Models Alon A-2 and A2-A; ERCO 415-C, 415-CD, 415-D, 415-E, and 415-G; Forney F-1 and F-1A, and

Mooney M10. This proposal was published in the **Federal Register** as a notice of proposed rulemaking (NPRM) on May 30, 2003 (68 FR 32437). The NPRM proposed to supersede AD 94-18-04 R1 with a new AD that would retain the actions required in AD 94-18-04 R1 and make the one-time inspection of the wing outer panel structural components for corrosion a repetitive inspection.

**Comments**

*Was the public invited to comment?* We provided the public the opportunity to participate in the development of this AD. We received no comments on the proposal or on the determination of the cost to the public.

**Conclusion**

*What is FAA's final determination on this issue?* We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed except for minor editorial corrections. We have determined that these minor corrections:

- Provide the intent that was proposed in the NPRM for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

**Changes to 14 CFR Part 39—Effect on the AD**

*How does the revision to 14 CFR part 39 affect this AD?* On July 10, 2002, the FAA published a new version of 14 CFR part 39 (67 FR 47997, July 22, 2002), which governs the FAA's AD system. This regulation now includes material that relates to altered products, special flight permits, and alternative methods of compliance. This material previously was included in each individual AD. Since this material is included in 14 CFR part 39, we will not include it in future AD actions.

**Costs of Compliance**

*How many airplanes does this AD impact?* We estimate that this AD affects 2,600 airplanes in the U.S. registry.

*What is the cost impact of this AD on owners/operators of the affected airplanes?* We estimate the following costs to accomplish the installation of the inspection openings:

	Labor cost	Parts cost	Total cost per airplane
6 workhours × \$60 per hour = \$360 .....		\$67	\$427

We estimate the following costs to accomplish the inspection:

Labor cost	Parts cost	Total cost per airplane	Total cost on U.S. operators
2 workhours $\times$ \$60 per hour = \$120 .....	Not applicable .....	\$120	\$120 $\times$ 2,600 = \$312,000

The FAA has no method of determining the number of repetitive inspections each owner/operator will incur over the life of each of the affected airplanes so the cost impact is based on the initial inspection.

The FAA has no method of determining the number of repairs or replacements each owner/operator will incur over the life of each of the affected airplanes based on the results of the inspections. We have no way of determining the number of airplanes that may need such repair. The extent of damage may vary on each airplane.

#### Compliance Time of This AD

*What is the compliance time of this AD?* The compliance time of this AD is "within the next 12 calendar months after the effective date of this AD."

*Why is the compliance time presented in calendar time instead of hours time-in-service (TIS)?* The unsafe condition specified by this AD is caused by corrosion. Corrosion can occur regardless of whether the airplane is in operation or is in storage. Therefore, to assure that the unsafe condition specified in this AD does not go undetected for a long period of time, the compliance is presented in calendar time instead of hours TIS.

#### Regulatory Findings

*Will this AD impact various entities?* We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and

responsibilities among the various levels of government.

*Will this AD involve a significant rule or regulatory action?* For the reasons discussed above, I certify that this AD:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD and placed it in the AD Docket. You may get a copy of this summary by sending a request to us at the address listed under **ADDRESSES**. Include "AD Docket No. 2002-CE-58-AD" in your request.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. FAA amends § 39.13 by removing Airworthiness Directive (AD) 94-18-04

R1, Amendment 39-9173 (60 FR 62321, March 14, 1995), and by adding a new AD to read as follows:

**2003-21-01 Univair Aircraft Corporation:** Amendment 39-13335; Docket No. 2002-CE-58-AD; Supersedes AD 94-18-04 R1, Amendment 39-9173.

#### When Does This AD Become Effective?

- (a) This AD becomes effective on December 1, 2003.

#### What Other ADs Are Affected by This Action?

- (b) This AD supersedes AD 94-18-04 R1, Amendment 39-9173.

#### What Airplanes Are Affected by This AD?

- (c) This AD affects the following airplane models and serial numbers that are certificated in any category:

Models	Serial Nos.
Alon A-2 and A2-A .....	All.
ERCO 415-C, 415-CD, 415-D, 415-E, and 415-G.	All.
Forney F-1 and F-1A .....	All.
Mooney M10 .....	All.

#### What Is the Unsafe Condition Presented in This AD?

(d) This AD is the result of additional reports of corrosion on airplanes in compliance with AD 94-18-04 R1. The actions specified in this AD are intended to prevent wing damage caused by a corroded wing outer panel structural component, which, if not detected and corrected, could progress to the point of structural failure.

#### What Must I Do To Address This Problem?

- (e) To address this problem, you must accomplish the following:

Actions	Compliance	Procedures
(1) Install inspection openings in the outer wing panels and inspect the wing outer panel internal structural components for corrosion and unrepaired corrosion damage.	Within the next 12 calendar months after March 24, 1995 (the effective date of AD 94-18-04 R1), unless already accomplished.	In accordance with Univair Service Bulletin No. 29, Revision B, dated January 2, 1995, or Univair Service Bulletin No. 29, Revision C, dated July 8, 1999, and Advisory Circular 43-4A, Corrosion Control for Aircraft.
(2) If corrosion or corrosion damage is found during the inspection required in paragraph (e)(1) of this AD, repair or replace components of the wing outer panel structure.	Repair or replace prior to further flight after the inspection required in paragraph (e)(1) of this AD.	In accordance with Univair Service Bulletin No. 29, Revision B, dated January 2, 1995, or Univair Service Bulletin No. 29, Revision C, dated July 8, 1999, the applicable maintenance manual, and Advisory Circular 43-4A, Corrosion Control for Aircraft.

Actions	Compliance	Procedures
(3) Repetitively inspect the wing outer panel internal structural components for corrosion and unrepaired corrosion damage.	Initially inspect within the next 6 calendar months after December 1, 2003 (the effective date of this AD), unless the wing outer panel internal structure has been inspected for corrosion within the previous 6 calendar months immediately prior to December 1, 2003 (the effective date of this AD). Repetitively inspect thereafter at intervals not to exceed 12 calendar months after the last inspection.	In accordance with Univair Service Bulletin No. 29, Revision C, dated July 8, 1999, and Advisory Circular 43-4A, Corrosion Control for Aircraft.
(4) At any time corrosion or corrosion damage is found, repair or replace components of the wing outer panel structure.	Repair or replace prior to further flight after the inspection in which the corrosion or corrosion damage is found. Continue with the repetitive inspection requirements of this AD.	In accordance with Univair Service Bulletin No. 29, Revision B, dated January 2, 1995, or Univair Service Bulletin No. 29, Revision C, dated July 8, 1999, and Advisory Circular 43-4A, Corrosion Control for Aircraft.

**Note:** The compliance times specified in Univair Aircraft Corporation Service Bulletin No. 29, Revision B, dated January 2, 1995, or Univair Service Bulletin No. 29, Revision C, dated July 8, 1999, are different from those required by this AD. The compliance times in this AD take precedence over those in the service bulletin.

#### What About Alternative Methods of Compliance?

(f) You may request a different method of compliance or a different compliance time for this AD by following the procedures in 14 CFR 39.13.

(1) Send your request to the Manager, Denver Aircraft Certification Office (ACO). For information on any already approved alternative methods of compliance, contact Roger Caldwell, Aerospace Engineer, FAA, Denver Aircraft Certification Office, 26805 East 68th Avenue, Room 214, Denver, Colorado 80249-6361; telephone: (303) 342-1086; facsimile: (303) 342-1088.

(2) Alternative methods of compliance approved for the inspection required in AD 94-18-04 R1, which is superseded by this AD, are approved as alternative methods of compliance with this AD.

#### Is There Material Incorporated by Reference?

(g) You must do the actions required by this AD per Univair Aircraft Corporation Service Bulletin No. 29, Revision B, dated January 2, 1995, and Univair Service Bulletin No. 29, Revision C, dated July 8, 1999.

(1) The Director of the Federal Register approved the incorporation by reference of Univair Aircraft Corporation Mandatory Service Bulletin No. 29, Revision C, dated July 8, 1999, in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. You may get a copy from Univair Aircraft Corporation, 2500 Himalaya Road, Aurora, Colorado 80011, telephone: (303) 375-8882; facsimile: (303) 375-8888.

(2) The Director of the Federal Register previously approved the incorporation by reference of Univair Aircraft Corporation Service Bulletin No. 29, Revision B, dated January 2, 1995, as of March 24, 1995 (60 FR 13626, March 14, 1995).

(3) You may review copies at FAA, Central Region, Office of the Regional Counsel, 901 Locust, Room 506, Kansas City, Missouri

64106; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

Issued in Kansas City, Missouri, on October 6, 2003.

**James E. Jackson,**

*Acting Manager, Small Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 03-25699 Filed 10-10-03; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 2002-SW-17-AD; Amendment 39-13330; AD 2003-20-12]

**RIN 2120-AA64**

#### Airworthiness Directives; Eurocopter France Model AS 350B3, SA-365N, N1, AS-365N2, AS 365N3, and EC 155B Helicopters

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD) for the specified Eurocopter France (Eurocopter) model helicopters that requires modifying the SIREN cargo hook and inspecting the cargo hook locking catch (locking catch) for corrosion. This amendment is prompted by the discovery of internal corrosion on a Siren locking catch that may weaken the locking catch. The actions specified by this AD are intended to detect internal corrosion of the locking catch, which can cause the locking catch to return to an incomplete locking position, undetectable by the operator, and result in an unexpected cargo load release.

**DATES:** Effective November 18, 2003.

The incorporation by reference of certain publications listed in the

regulations is approved by the Director of the Federal Register as of November 18, 2003.

**ADDRESSES:** The service information referenced in this AD may be obtained from American Eurocopter Corporation, 2701 Forum Drive, Grand Prairie, Texas 75053-4005, telephone (972) 641-3460, fax (972) 641-3527. This information may be examined at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Carroll Wright, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Regulations and Guidance Group, Fort Worth, Texas 76193-0111, telephone (817) 222-5120, fax (817) 222-5961.

**SUPPLEMENTARY INFORMATION:** A proposal to amend 14 CFR part 39 to include an AD for Eurocopter Model AS 350B3, SA-365N, N1, AS-365N2, AS 365N3, and EC 155B helicopters was published in the **Federal Register** on April 22, 2003 (68 FR 19757). That action proposed to require modifying the SIREN cargo hook and inspecting the cargo hook locking catch for corrosion.

The Direction Generale De L'Aviation Civile (DGAC), the airworthiness authority for France, notified the FAA that an unsafe condition may exist on Eurocopter Model AS 350 B3, AS-365N, N1, AS 365N2, AS 365N3, and EC 155 B helicopters fitted with Siren cargo hooks, part number (P/N) AS-21-5-7. The DGAC advises that corrosion was discovered on a locking catch, which might lead to untimely load release.

Eurocopter has issued Alert Telexes No. 05.00.39, for Model AS 350B3 helicopters; No. 05.00.41, for Model AS 365N, N1, AS 365N2, and AS 365N3 helicopters; and No. 05A002, for Model EC 155B helicopters; all dated December