• provide the flight crew with recognition cues for, and procedures for exiting from, severe icing conditions.

In addition to the proposed rule described previously, in September 1997, the FAA issued 24 other similar proposals that address the subject unsafe condition on various airplane models. These 24 proposals also were published in the **Federal Register** on September 16, 1997. Numerous comments were received on all 24 proposed actions.

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comment received for this AD. The comment received is from the manufacturer and directly relates to the I.A.M. Model Piaggio P–180 airplane. The manufacturer states that the I.A.M. Model Piaggio P–180 does not meet the criteria of the proposed action. This airplane has a bleed air and electrothermal wing anti-icing system, not a pneumatic system. The FAA concurs and has decided to withdraw the NPRM.

Withdrawal of this NPRM constitutes only such action, and does not preclude the agency from issuing future rulemaking on this issue, nor does it commit the agency to any course of action in the future.

Since this action only withdraws an NPRM, it is neither a proposed nor a final rule and therefore, is not covered under Executive Order 12866, the Regulatory Flexibility Act, or DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979).

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation Safety, Safety.

# The Withdrawal

Accordingly, the notice of proposed rulemaking, Docket No. 97–CE–52–AD, published in the **Federal Register** on September 16, 1997 (62 FR 48502), is withdrawn.

Issued in Kansas City, Missouri, on May 22, 1998.

#### Michael Gallagher,

Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98–14616 Filed 6–2–98; 8:45 am] BILLING CODE 4910–13–U

# DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 98-NM-142-AD]

RIN 2120-AA64

### Airworthiness Directives; Honeywell IC–600 Integrated Avionics Computers, as Installed in, but not Limited to, Learjet Model 45 and EMBRAER Model EMB–145 Series Airplanes

AGENCY: Federal Aviation Administration, DOT. ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Honeywell IC-600 integrated avionics computers. This proposal would require modification of the integrated avionics computers. This proposal is prompted by a report of integrated avionics computer failures, which caused a "random reset" condition of the electronic flight instrument system. The actions specified by the proposed AD are intended to prevent such "random reset" conditions, which could affect the pilot's ability to control the airplane. DATES: Comments must be received by July 20, 1998.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–114, Attention: Rules Docket No. 98–NM– 142–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Honeywell Inc., Business and Commuter Aviation Systems, Box 29000, Phoenix, Arizona 85038. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Transport Airplane Directorate, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California.

FOR FURTHER INFORMATION CONTACT: J. Kirk Baker, Aerospace Engineer, Systems and Equipment Branch, ANM– 130L, FAA, Transport Airplane Directorate, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California 90712–4137; telephone (562) 627–5345; fax (562) 627–5210.

# SUPPLEMENTARY INFORMATION:

#### **Comments Invited**

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 98–NM–142–AD." The postcard will be date stamped and returned to the commenter.

#### Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM–114, Attention: Rules Docket No. 98–NM–142–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056.

## Discussion

The FAA has received a report indicating that, during several test flights of a Cessna Model XL series airplane, the screen of the electronic flight instrument system (EFIS) turned blank. Investigation has revealed that the Honeywell IC-600 integrated avionics computer failed, which resulted in a "random reset" condition of the EFIS. During such a "random reset" condition, the following events may occur: the primary flight displays may turn blank, or display a red "X," and take 10 to 15 seconds to reboot; the flight director mode may drop; the EFIS may reset to its default state; and the autopilot and/or the yaw damper, if engaged, may disconnect. This "random reset" condition of the EFIS, if not

corrected, could affect the pilot's ability to control the airplane.

The affected computers are Honeywell IC–600 integrated avionics computers, which may be installed in, but not limited to, Learjet Model 45 and EMBRAER Model EMB–145 series airplanes.

# Explanation of Relevant Service Information

The FAA has reviewed and approved Honeywell Service Bulletin 7017000– 22–43, dated March 24, 1998, which describes procedures for modification of the integrated avionics computers. This modification involves replacing programmable array logic devices U6, U32, and U33 with new modified parts; adding two resistors; replacing resistor R7 with a new value; adding two capacitors; and adding bus wires. Accomplishment of the actions specified in the service bulletin is intended to adequately address the identified unsafe condition.

### Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other products of this same type design, the proposed AD would require accomplishment of the actions specified in the service bulletin described previously.

### **Cost Impact**

There are approximately 38 airplanes of the affected design in the worldwide fleet. The FAA estimates that 20 airplanes of U.S. registry would be affected by this proposed AD. It would take approximately 2 work hours per airplane to accomplish the proposed modification at an average labor rate of \$60 per work hour. Required parts would be supplied by the manufacturer at no cost to operators. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be \$2,400, or \$120 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

### **Regulatory Impact**

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT **Regulatory Policies and Procedures (44** FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

# List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

### **The Proposed Amendment**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### §39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Honeywell. Docket 98-NM-142-AD.

Applicability: Honeywell IC-600 integrated avionics computers having part numbers 7017000-82201, -82202, -82203, -82204, -82351, -82352, -82401, -82402, -82403, -83351, -83352, -83401, -83402, and -83403, as installed in, but not limited to, Learjet Model 45 and EMBRAER Model EMB-145 series airplanes.

Note 1: This AD applies to Honeywell IC-600 integrated avionics computers having part numbers 7017000-82201, -82202, -82203, -82204, -82351, -82352, -82401,-82402, -82403, -83351, -83352, -83401, -83402, and -83403, as installed in any airplane, regardless of whether the airplane has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of

the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

*Compliance:* Required as indicated, unless accomplished previously.

To prevent a "random reset" condition of the electronic flight instrument system, which could affect the pilot's ability to control the airplane, accomplish the following:

(a) Within 6 months after the effective date of this AD, modify the IC–600 integrated avionics computer, in accordance with Honeywell Service Bulletin 7017000–22–43, dated March 24, 1998.

(b) As of the effective date of this AD, no person shall install a Honeywell IC-600 integrated avionics computer having part number 7017000-82201, -82202, -82203, -82204, -82351, -82352, -82401, -82402, -82403, -83351, -83352, -83401, -83402, or -83403 on any airplane, unless it has been modified in accordance with Honeywell Service Bulletin 7017000-22-43, dated March 24, 1998.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Los Angeles Aircraft Certification Office (ACO), FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Los Angeles ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Los Angeles ACO.

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on May 27, 1998.

#### Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 98–14612 Filed 6–2–98; 8:45 am] BILLING CODE 4910–13–U

#### **DEPARTMENT OF TRANSPORTATION**

# **Federal Aviation Administration**

# 14 CFR Part 71

[Airspace Docket No. 98-AGL-36]

### Proposed Removal of Class D Airspace and Class E Airspace; Willoughby, OH

**AGENCY:** Federal Aviation Administration (FAA), DOT. **ACTION:** Notice of proposed rulemaking.

**SUMMARY:** This notice proposes to remove Class D airspace and Class E