

under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

#### 98-11-27 Dornier Luftfahrt GmbH:

Amendment 39-10552. Docket 98-NM-46-AD.

**Applicability:** Model 328-100 series airplanes, serial numbers 3005 through 3095 inclusive; certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent failure of the pressure dump and relief valves in the main and auxiliary hydraulic systems, which could cause a loss in hydraulic pressure for roll control spoilers and brakes, and consequent reduced controllability of the airplane, accomplish the following:

(a) Within 8 months after the effective date of this AD, replace the existing pressure dump and relief valves having part number (P/N) ZHV29-1 with new valves having P/N ZHV29-2, in the main and auxiliary hydraulic systems, in accordance with Dornier Service Bulletin SB-328-29-205, dated February 12, 1997.

(b) As of the effective date of this AD, no person shall install on any airplane any pressure dump and relief valve having P/N ZHV29-1.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(e) The replacement shall be done in accordance with Dornier Service Bulletin SB-328-29-205, dated February 12, 1997. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from FAIRCHILD Dornier, Dornier Luftfahrt GmbH, P.O. Box 1103, D-82230 Wessling, Germany. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Note 3:** The subject of this AD is addressed in German airworthiness directive 97-072, dated March 27, 1997.

(f) This amendment becomes effective on July 8, 1998.

Issued in Renton, Washington, on May 22, 1998.

**Darrell M. Pederson,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 98-14218 Filed 6-2-98; 8:45 am]

**BILLING CODE 4910-13-U**

#### DEPARTMENT OF TRANSPORTATION

#### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 98-NM-52-AD; Amendment 39-10554; AD 98-11-29]

**RIN 2120-AA64**

#### Airworthiness Directives; British Aerospace Model BAe 146 Series Airplanes and Model Avro 146-RJ Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to certain British Aerospace Model BAe 146 series airplanes and Model Avro 146-RJ series airplanes, that requires a one-time inspection to detect corrosion of the threads of the eyebolt and piston rod on the retraction jack of the main landing gear (MLG); and repair, if necessary. This amendment is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by this AD are intended to prevent corrosion of the threads of the eyebolt and piston rod on the retraction jack of the MLG, which may cause the eyebolt to detach from the jack, and consequent unrestrained MLG deployment or inability to retract the MLG.

**DATES:** Effective July 8, 1998.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of July 8, 1998.

**ADDRESSES:** The service information referenced in this AD may be obtained from AI(R) American Support, Inc., 13850 Mclearen Road, Herndon, Virginia 20171. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

#### FOR FURTHER INFORMATION CONTACT:

Norman B. Martenson, Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain British Aerospace Model BAe 146 series airplanes and Model Avro 146-RJ series airplanes was published in the **Federal Register** on April 3, 1998 (63 FR 16447). That action proposed to require a one-time inspection to detect corrosion of the threads of the eyebolt and piston rod on the retraction jack of the main landing gear, and repair, if necessary.

#### Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

## Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

## Cost Impact

The FAA estimates that 25 airplanes of U.S. registry will be affected by this AD, that it will take approximately 1 work hour per airplane to accomplish the required inspection, and that the average labor rate is \$60 per work hour. Based on this figure, the cost impact of the AD on U.S. operators is estimated to be \$1,500, or \$60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

## Regulatory Impact

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

## List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

## Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

## PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

**98-11-29 British Aerospace Regional Aircraft (Formerly British Aerospace Regional Aircraft Limited, Avro International Aerospace Division; British Aerospace, PLC; British Aerospace Commercial Aircraft Limited):** Amendment 39-10554. Docket 98-NM-52-AD.

**Applicability:** Model BAe 146 series airplanes and Model Avro 146-RJ series airplanes, as listed in British Aerospace Service Bulletin SB.32-145, Revision 1, dated October 6, 1997, certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent corrosion of the threads of the eyebolt and piston rod on the retraction jack of the main landing gear (MLG), which may cause the eyebolt to detach from the jack, and consequent unrestrained MLG deployment or inability to retract the MLG, accomplish the following:

(a) Perform a one-time visual inspection to detect corrosion of the threads of the eyebolt and piston rod on the retraction jack of the MLG, in accordance with British Aerospace Service Bulletin SB.32-145, Revision 1, dated October 6, 1997, at the time specified in paragraph (a)(1), (a)(2), or (a)(3) of this AD, as applicable. Except as provided by paragraph (b) of this AD, if any corrosion is detected: Prior to further flight, repair in accordance with the service bulletin.

(1) For MLG retraction jacks that have accumulated more than 7 and less than 9 years since date of manufacture: Inspect within 2 years after the effective date of this AD.

(2) For MLG retraction jacks that have accumulated 9 or more years since date of manufacture: Inspect within 1 year after the effective date of this AD.

(3) For MLG retraction jacks other than those identified in paragraph (a)(1) or (a)(2) of this AD, and other than those MLG retraction jacks having Part/Type No.

104628003 with serial numbers DH/0029/93 (where "93" identifies the year of manufacture) and subsequent: Inspect within 6 years since date of manufacture, or within 2 years after the effective date of this AD, whichever occurs later.

**Note 2:** British Aerospace Service Bulletin SB.32-145, Revision 1, dated October 6, 1997, references Dowty Aerospace Hydraulics—Cheltenham Service Bulletin 146-32-507, dated August 1, 1997, as an additional source of service information to accomplish the inspection and repair.

(b) If any corrosion is detected during the inspection required by paragraph (a) of this AD in areas other than those detailed in British Aerospace Service Bulletin SB.32-145, Revision 1, dated October 6, 1997: Prior to further flight, repair in accordance with a method approved by the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate.

(c) As of the effective date of this AD, no person shall install an eyebolt or piston rod on the retraction jack of the MLG on any airplane unless it has been modified in accordance with British Aerospace Service Bulletin SB.32-145, Revision 1, dated October 6, 1997.

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM-116. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM-116.

**Note 3:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

(f) The actions shall be done in accordance with British Aerospace Service Bulletin SB.32-145, Revision 1, dated October 6, 1997, which contains the following list of effective pages:

Page number shown on page	Revision level shown on page	Date shown on page
1, 2 .....	1 .....	Oct. 6, 1997.
3-7 .....	Original ....	Sept. 18, 1997.

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from AI(R) American Support, Inc., 13850 Mclearen Road, Herndon, Virginia 20171. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at

the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Note 4:** The subject of this AD is addressed in British airworthiness directive 006-09-97.

(g) This amendment becomes effective on July 8, 1998.

Issued in Renton, Washington, on May 22, 1998.

**Darrell M. Pederson,**

*Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.*  
[FR Doc. 98-14217 Filed 6-2-98; 8:45 am]

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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. 97-NM-134-AD; Amendment 39-10551; AD 98-11-26]

RIN 2120-AA64

#### **Airworthiness Directives; Saab Model SAAB 340B and SAAB 2000 Series Airplanes**

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to certain Saab Model SAAB 340B and SAAB 2000 series airplanes, that requires modification of the check valves of the airfoil de-icing system, or replacement of the check valves with improved valves. This amendment is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by this AD are intended to prevent failure of the check valves, which could result in loss of airfoil de-icing capability during single engine operation, and consequent reduced controllability of the airplane.

**DATES:** Effective July 8, 1998.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of July 8, 1998.

**ADDRESSES:** The service information referenced in this AD may be obtained from Saab Aircraft AB, SAAB Aircraft Product Support, S-581.88, Linköping, Sweden. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

#### **FOR FURTHER INFORMATION CONTACT:**

Norman B. Martenson, Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

#### **SUPPLEMENTARY INFORMATION:**

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Saab Model SAAB 340B and SAAB 2000 series airplanes was published in the **Federal Register** on April 1, 1998 (63 FR 15797). That action proposed to require modification of the check valves of the airfoil de-icing system, or replacement of the check valves with improved valves.

#### **Comments**

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

#### **Conclusion**

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

#### **Cost Impact**

The FAA estimates that 125 Model SAAB 340B and SAAB 2000 series airplanes of U.S. registry will be affected by this AD, that it will take approximately 4 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will be provided by the manufacturer at no cost to the operators. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$30,000, or \$240 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

#### **Regulatory Impact**

The regulations adopted herein will not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

#### **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

#### **Adoption of the Amendment**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### **PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### **§ 39.13 [Amended]**

2. Section 39.13 is amended by adding the following new airworthiness directive:

#### **98-11-26 SAAB AIRCRAFT AB:**

Amendment 39-10551. Docket 97-NM-134-AD.

**Applicability:** Model SAAB 340B series airplanes, serial numbers 240 through 430 inclusive; Model SAAB 2000 series airplanes, serial numbers 002 through 050 inclusive, and 052; certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent failure of the check valves, which could result in loss of airfoil de-icing