determined that AD action is necessary for products of this type design that are certificated for operation in the United States

Explanation of the Provisions of the Proposed AD

Since an unsafe condition has been identified that is likely to exist or develop in other Aeromot Model AMT–200 powered gliders of the same type design registered in the United States, the FAA is proposing AD action. The proposed AD would require replacing any engine oil system hose, part number 10702, 10703, or 10704; with a hose with a larger internal diameter, part number 10706, 10707, or 10708. Accomplishment of the proposed installation would be in accordance with Aeromot SB B.S. No. 200–79–036, Issue Date: January 30, 1997.

Cost Impact

The FAA estimates that 18 powered gliders in the U.S. registry would be affected by the proposed AD, that it would take approximately 7 workhours per powered glider to accomplish the proposed replacements, and that the average labor rate is approximately \$60 an hour. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$7,560, or \$420 per powered glider.

Regulatory Impact

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

Aeromot-Industria Mecanico Metalurgica Ltda.: Docket No. 98-CE-27-AD.

Applicability: Model AMT–200 powered gliders, serial numbers 200.046 through 200.066, certificated in any category.

Note 1: This AD applies to each powered glider identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For powered gliders that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 50 hours time-in-service (TIS) after the effective date of this AD, unless already accomplished.

To prevent inefficiency of the engine lubricating system because of ineffective flexible hoses, which could result in an inflight engine shutdown with consequent loss of powered glider controllability, accomplish the following:

(a) For powered gliders with a serial number in the range of 200.046 through 200.058: Replace any engine oil system hose, part number 10702; with a hose with a larger internal diameter, part number 10706. Accomplish the replacement in accordance with Aeromot Service Bulletin B.S. No. 200–79–036, Issue Date: January 30, 1997.

(b) For powered gliders with a serial number in the range of 200.059 through 200.066: Replace any engine oil system hose, part number 10702, 10703, or 10704; with a hose with a larger diameter, part number 10706, 10707, or 10708. Accomplish the replacement in accordance with Aeromot Service Bulletin B.S. No. 200–79–036, Issue Date: January 30, 1997.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199

of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the glider to a location where the requirements of this AD can be accomplished.

(d) An alternative method of compliance or adjustment of the compliance times that provides an equivalent level of safety may be approved by the Manager, Atlanta Aircraft Certification Office (ACO), One Crown Center, 1895 Phoenix Blvd., suite 450, Atlanta, Georgia 30349. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Atlanta ACO.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Atlanta ACO.

(e) Questions or technical information related to Aeromot Service Bulletin B.S. No. 200–79–036, Issue Date: January 30, 1997, Grupo Aeromot, Aeromot-Industria Mecanico Metalurgica Ltda., Av. das Industries-1210, Bairro Anchieta, Caixa Postal 8031, 90200-Porto Alegre-RS, Brazil. This service information may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

Note 3: The subject of this AD is addressed in Brazilian AD 97–04–02, dated April 8, 1997.

Issued in Kansas City, Missouri, on April 23, 1998.

Marvin R. Nuss,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98–11439 Filed 4–29–98; 8:45 am] BILLING CODE 4910–13–U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 98-CE-28-AD]

RIN 2120-AA64

Airworthiness Directives; British Aerospace Jetstream Models 3101 and 3201 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes to adopt a new airworthiness directive (AD) that would apply to certain British Aerospace Jetstream Models 3101 and 3201 airplanes that are equipped with the ground inhibit function (Modification JM7813A (SB 27–JM7813A) or JM7813B). The proposed AD would require removing the ground inhibit time delay and the ground test relay from the stall warning and protection system. This proposed AD

also requires rewiring part of the stall warning and protection system to assure that system reliance is maintained after relay removal. The proposed AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for the United Kingdom. The actions specified by the proposed AD are intended to prevent failure of the ground inhibit relay while it is in the energized position caused by the current design, which could result in failure of the stall warning system and possible loss of control of the airplane in certain situations if the crew was not aware that the system had failed.

DATES: Comments must be received on or before June 9, 1998.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 98–CE–28–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

Service information that applies to the proposed AD may be obtained from British Aerospace Regional Aircraft, Prestwick International Airport, Ayrshire, KA9 2RW, Scotland; telephone: (01292) 479888; facsimile: (01292) 479703. This information also may be examined at the Rules Docket at the address above.

FOR FURTHER INFORMATION CONTACT: Mr. S.M. Nagarajan, Aerospace Engineer, FAA, Small Airplane Directorate, Aircraft Certification Service, 1201 Walnut, suite 900, Kansas City, Missouri 64106; telephone: (816) 426–6934; facsimile: (816) 426–2169.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 98–CE–28–AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 98–CE–28–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

Discussion

The Civil Airworthiness Authority (CAA), which is the airworthiness authority for the United Kingdom, notified the FAA that an unsafe condition may exist on certain British Aerospace Jetstream Models 3101 and 3201 airplanes that are equipped with the ground inhibit function (Modification JM7813A (SB 27-JM7813A) or JM7813B). These modifications incorporate a ground inhibit relay and a ground test relay to prevent operations of the stall warning system until the aircraft is airborne for three seconds. The CAA reports that failure of the ground inhibit relay in the energized positioned could occur.

This condition, if not corrected in a timely manner, could result in loss of control of the airplane in certain situations if the crew was not aware that the stall warning system had failed.

Relevant Service Information

British Aerospace has issued Jetstream Alert Service Bulletin 27–A– JM7847, dated December 24, 1997, which specifies procedures for the following:

- removing the ground inhibit time delay and the ground test relay from the stall warning and protection system; and
- —rewiring part of the stall warning and protection system to assure that system reliance is maintained after relay removal.

The CAA classified this service bulletin as mandatory in order to assure the continued airworthiness of these airplanes in the United Kingdom. The CAA classifying a service bulletin as mandatory is the same in the United Kingdom as the FAA issuing an AD in the United States.

The FAA's Determination

This airplane model is manufactured in the United Kingdom and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the CAA has kept the FAA informed of the situation described above.

The FAA has examined the findings of the CAA; reviewed all available information, including the service information referenced above; and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Explanation of the Provisions of the Proposed AD

Since an unsafe condition has been identified that is likely to exist or develop in other British Aerospace Jetstream Models 3101 and 3102 airplanes of the same type design that are registered in the United States and are equipped with the ground inhibit function (Modification JM7813A (SB 27-JM7813A) or JM7813B), the FAA is proposing AD action. The proposed AD would require removing the ground inhibit time delay and the ground test relay from the stall warning and protection system. This proposed AD also requires rewiring part of the stall warning and protection system to assure that system reliance is maintained after relay removal. Accomplishment of the proposed actions would be in accordance with British Aerospace Jetstream Alert Service Bulletin 27-A-JM7847, dated December 24, 1997.

Cost Impact

The FAA estimates that 301 airplanes in the U.S. registry would be affected by the proposed AD, that it would take approximately 6 workhours per airplane to accomplish the proposed action, and that the average labor rate is approximately \$60 an hour. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$108,360, or \$360 per airplane.

Regulatory Impact

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

British Aerospace: Docket No. 98-CE-28-

Applicability: Jetstream Models 3101 and 3201 airplanes, all serial numbers, certificated in any category, that are equipped with the ground inhibit function (Modification JM7813A (SB 27–JM7813A) or IM7813B)

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by

this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within 100 hours time-in-service (TIS) after the effective date of this AD, unless already accomplished.

To prevent failure of the ground inhibit relay while it is in the energized position caused by the current design, which could result in failure of the stall warning system and possible loss of control of the airplane in certain situations if the crew was not aware that the system had failed, accomplish the following:

- (a) Remove the ground inhibit time delay and the ground test relay from the stall warning and protection system, and rewire part of the stall warning and protection system to assure that system reliance is maintained after relay removal. Accomplish these actions in accordance with the ACCOMPLISHMENT INSTRUCTIONS section of British Aerospace Jetstream Alert Service Bulletin 27–A–JM7847, dated December 24, 1997.
- (b) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.
- (c) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Small Airplane Directorate, Aircraft Certification Service, 1201 Walnut, suite 900, Kansas City, Missouri 64106. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Small Airplane Directorate.

(d) Questions or technical information related to British Aerospace Jetstream Alert Service Bulletin 27–A–JM7847, dated December 24, 1997, should be directed to British Aerospace Regional Aircraft, Prestwick International Airport, Ayrshire, KA9 2RW, Scotland; telephone: (01292) 479888; facsimile: (01292) 479703. This service information may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

Note 3: The subject of this AD is addressed in British Aerospace Jetstream Alert Service Bulletin 27–A–JM7847, dated December 24, 1997. This service bulletin is classified as mandatory by the United Kingdom Civil Aviation Authority (CAA).

Issued in Kansas City, Missouri, on April 23, 1998.

Marvin R. Nuss,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98–11438 Filed 4–29–98; 8:45 am] BILLING CODE 4910–13–U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 98-ANE-06-AD]

Airworthiness Directives; Rolls-Royce, plc Viper Series Turbojet Engines

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to Rolls-Royce, plc Viper series turbojet engines. This proposal would require a one-time visual inspection of the barometric flow control unit (BFCU) augmentor and bypass valve joint washer for joint washer integrity, and replacement, if necessary, with serviceable parts. This proposal is prompted by a report of a high pressure fuel leak at the BFCU augmentor and bypass valve assembly joint, washer interface. The actions specified by the proposed AD are intended to prevent a high pressure fuel leak, which could result in an engine nacelle fire and damage to the aircraft.

DATES: Comments must be received by June 1, 1998.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), New England Region, Office of the Regional Counsel, Attention: Rules Docket No. 98–ANE–06–AD, 12 New England Executive Park, Burlington, MA 01803–5299. Comments may also be submitted to the Rules Docket by using the following Internet address: "9-adengineprop@faa.dot.gov". Comments may be inspected at this location between 8:00 a.m. and 4:30 p.m., Monday through Friday, except Federal

The service information referenced in the proposed rule may be obtained from Rolls-Royce, plc, Technical Publications Department CLS-4, P.O. Box 3, Filton, Bristol, BS34 7QE England; telephone 117–979–1234, fax 117–979–7575. This information may be examined at the FAA, New England Region, Office of the Regional Counsel, 12 New England Executive Park, Burlington, MA.

holidays.

FOR FURTHER INFORMATION CONTACT: James Lawrence, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803–5299; telephone (781) 238–7176, fax (781) 238–7199.