wheel leg version "B", in accordance with the applicable maintenance

Cost Impact

The FAA estimates that 20 airplanes in the U.S. registry would be affected by the proposed AD, that it would take approximately 1 workhour per airplane to accomplish the proposed replacement, and that the average labor rate is approximately \$60 per hour. Parts cost approximately \$900 per airplane. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$19,200 or \$960 per airplane.

Differences Between the Austrian AD, the Service Bulletin, and This Proposed AD

Austrian AD No. 86, dated May 29, 1996, and Diamond Aircraft Industries Service Bulletin N. 20-32, dated April 5, 1996, both give the owners/operators of certain HOAC Austria Model DV-20 Katana airplanes the option of either (1) repetitively inspecting the weld of the nose wheel tappet in the NLG until cracks are found, at which time immediate modification or replacement (with parts of improved design) would be required; or (2) immediately replacing the weld of the nose wheel tappet in the NLG with parts of improved design.

The FAA's policy is to provide corrective action that will eliminate the need for repetitive inspections. The FAA has determined that long-term operational safety will be better assured by design changes that remove the source of the problem, rather than by repetitive inspections or other special

procedures.

Because replacing the nose wheel leg (with parts of improved design) eliminates the need for repetitive inspections, the proposed AD differs from the service bulletin and the Austrian AD in that it would mandate the replacement.

Regulatory Impact

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under

Executive Order 12866; (2) is not a 'significant rule'' under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS **DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

HOAC Austria: Docket No. 97-CE-84-AD.

Applicability: Model DV-20 Katana airplanes, serial numbers 20005 through 20160, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 100 hours time-in-service (TIS) after the effective date of this AD, unless already accomplished.

To prevent nose landing gear (NLG) collapse caused by cracks in the welding of the nose wheel tappet of the NLG, which could result in the inability to control the airplane during landing, takeoff, and other ground operations, accomplish the following:

(a) Replace the nose wheel leg of the NLG with a part of improved design, nose wheel

leg version "B", in accordance with the applicable maintenance manual.

Note 2: Diamond Aircraft Industries Service Bulletin No. 20-32, dated April 5, 1996, specifies the replacement required by

- (b) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.
- (c) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Small Airplane Directorate, FAA, 1201 Walnut, suite 900, Kansas City, Missouri 64106. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Small Airplane Directorate.

(d) All persons affected by this directive may obtain copies of the document referred to herein upon request to HOAC Austria Ges.m.b.H., N.A. Otto-Strabe 5, A-2700, Wiener Neustadt, Austria; or may examine this document at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

Note 4: The subject of this AD is addressed in Austrian AD No. 86, dated May 29, 1996.

Issued in Kansas City, Missouri, on October 31, 1997.

James E. Jackson,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 97-29408 Filed 11-6-97; 8:45 am] BILLING CODE 4910-13-U

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-CE-43-AD]

RIN 2120-AA64

Airworthiness Directives; SOCATA-**Groupe AEROSPATIALE Model TBM** 700 Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes to adopt a new airworthiness directive (AD) that would apply to certain SOCATA—Groupe AEROSPATIALE (Socata) Model TBM 700 airplanes. The proposed AD would require replacing the starter generator mounting adapter

with a part of improved design. The proposed AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for France. The actions specified by the proposed AD are intended to prevent loss of the starter generator caused by failure of the starter generator mounting adapter, which could result in loss of electrical power. DATES: Comments must be received on or before December 15, 1997. ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97-CE-43-AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

Service information that applies to the proposed AD may be obtained from the SOCATA-Groupe AEROSPATIALE, Socata Product Support, Aeroport Tarbes-Ossun-Lourdes, B P 930, 65009 Tarbes Cedex, France; telephone 62.41.74.26; facsimile 62.41.74.32; or the Product Support Manager, SOCATA-Groupe AEROSPATIALE, North Perry Airport, 7501 Pembroke Road, Pembroke Pines, Florida 33023; telephone (954) 964–6877; facsimile (954) 964–1668. This information also may be examined at the Rules Docket at the address above.

FOR FURTHER INFORMATION CONTACT: Mr. Karl Schletzbaum, Aerospace Engineer, FAA, Small Airplane Directorate, 1201 Walnut Street, suite 900, Kansas City, Missouri 64106; telephone (816) 426–6934; facsimile (816) 426–2169.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 97–CE–43–AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97–CE–43–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

Discussion

The Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, recently notified the FAA that an unsafe condition may exist on certain Socata Model TBM 700 airplanes. The DGAC reports four incidents of cracked starter generator mounting adapters and one incident of a broken adapter. These conditions, if not detected and corrected, could result in loss of electrical power.

Relevant Service Information

Socata has issued Service Bulletin No. SB 70–072, dated January 1996, which specifies procedures for replacing the starter generator mounting adapter with a part of new design. The parts necessary for this replacement are included in Socata Kit No. OPT70K0058–24.

The DGAC classified this service bulletin as mandatory and issued French AD 95–242(B)R1, dated February 28, 1996, in order to assure the continued airworthiness of these airplanes in France.

The FAA's Determination

This airplane model is manufactured in France and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DGAC has kept the FAA informed of the situation described above.

The FAA has examined the findings of the DGAC; reviewed all available information, including the service information referenced above; and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Explanation of the Provisions of the Proposed AD

Since an unsafe condition has been identified that is likely to exist or develop in other Socata Model TBM 700 airplanes of the same type design registered in the United States, the FAA is proposing AD action. The proposed AD would require replacing the starter generator mounting adapter with a part of improved design by incorporating Socata Kit No. OPT70K0058–24. Accomplishment of the proposed replacement would be in accordance with Socata Service Bulletin No. SB 70–072, dated January 1996.

Cost Impact

The FAA estimates that 55 airplanes in the U.S. registry would be affected by the proposed AD.

The proposed replacement would take approximately 2 workhours per airplane to accomplish, at an average labor rate of approximately \$60 an hour. Parts to accomplish the proposed AD will be provided by the manufacturer at no cost to the owners/operators of the affected airplanes. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$6,600 or \$120 per airplane.

Differences Between the French AD, the Service Bulletin, and This Proposed AD

French AD 95-242(B)R1, dated February 28, 1996, and Socata Service Bulletin No. SB 70-072, dated January 1996, both give the owners/operators of certain Model TBM 700 airplanes the option of replacing the starter generator mounting adapter immediately or inspecting this adapter for cracks every 25 hours time-in-service (TIS) up to 100 hours TIS, at which time the replacement is mandatory. This allows the owners/operators the option of having their airplanes inspected up to three times before mandatory replacement, provided no cracked adapters were found, which, if found cracked, would require immediate replacement.

The FAA has determined that, since the parts for the replacement are free; the parts are available; and the action takes less than 2 workhours to accomplish, 25 hours TIS would be adequate time to incorporate the replacement. If followed with a final rule, the proposed AD would require replacing the starter generator mounting adapter within 25 hours TIS, and would not allow the option of repetitively

inspecting every 25 hours TIS up to 100 hours TIS.

Regulatory Impact

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a 'significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSÊS.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

SOCATA—Groupe Aerospatiale: Docket No. 97–CE–43–AD.

Applicability: Model TBM 700 airplanes, serial numbers 1 through 109, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the

requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 25 hours time-in-service (TIS), unless already accomplished.

To prevent loss of the starter generator caused by failure of the starter generator mounting adapter, which could result in loss of electrical power, accomplish the following:

- (a) Replace the starter generator mounting adapter with a part of improved design by incorporating Socata Kit No. OPT70K0058–24. This replacement shall be accomplished in accordance with the ACCOMPLISHMENT INSTRUCTIONS section of Socata Service Bulletin No. SB 70–072, dated January 1996.
- (b) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.
- (c) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Small Airplane Directorate, FAA, 1201 Walnut, suite 900, Kansas City, Missouri 64106. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Small Airplane Directorate.

(d) All persons affected by this directive may obtain copies of the document(s) referred to herein upon request to SOCATA—Groupe AEROSPATIALE, Socata Product Support, Aeroport Tarbes-Ossun-Lourdes, B P 930, 65009 Tarbes Cedex, France; or Perry Airport, 7501 Pembroke Road, Pembroke Pines, Florida 33023. These documents may also be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

Note 3: The subject of this AD is addressed in French AD 95–242(B)R1, dated February 28, 1996.

Issued in Kansas City, Missouri, on October 31, 1997.

James E. Jackson,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 97–29411 Filed 11–6–97; 8:45 am]

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-NM-109-AD] RIN 2120-AA64

Airworthiness Directives; Dornier Model 328–100 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Dornier Model 328-100 series airplanes. This proposal would require replacement of the main landing gear (MLG) uplocks with new or modified MLG uplocks. This proposal is prompted by the issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by the proposed AD are intended to prevent failure of the MLG to lock in the stowed position due to ice accumulation on the uplock hook and roller assembly, which could result in the inadvertent deployment of the MLG during flight.

DATES: Comments must be received by December 8, 1997.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–103, Attention: Rules Docket No. 97–NM–109–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Dornier Luftfahrt GmbH, P.O. Box 1103, D–82230 Wessling, Germany. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as