

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Airbus Industrie: Docket 95–NM–208–AD.

Applicability: Model A320–111, –211, and –231 series airplanes; manufacturer's serial numbers 002 through 008 inclusive, 010 through 014 inclusive, 016 through 078 inclusive, and 080 through 098 inclusive; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD are affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent fatigue cracking on the lower forward corner of the sliding window frames at frame 4, which could result in rapid depressurization of the airplane, accomplish the following:

(a) Prior to the accumulation of 15,000 total landings, or 3 months after the effective date of this AD, whichever occurs first, perform a high frequency eddy current inspection to detect cracks around the 5 fasteners of the lower forward corners of the sliding window frames at frame 4, in accordance with the procedures of Airbus Service Bulletin A320–53–1082, Revision 1, dated November 9, 1994.

(1) If no cracks are detected, repeat the inspection thereafter at intervals not to exceed 15,000 landings.

(2) If any crack is detected, prior to further flight, repair it in accordance with a method approved by the Manager, Standardization Branch, ANM–113, FAA, Transport Airplane Directorate.

(b) Prior to the accumulation of 30,000 total landings, or 5 years after the effective date of this AD, whichever occurs later, accomplish Airbus Modification 23685P3199 for each fastener hole, in accordance with Airbus Service Bulletin A320–53–1044, dated February 8, 1994. Accomplishment of the modification constitutes terminating action for the repetitive inspection requirements of paragraph (a) of this AD.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Standardization Branch, ANM–113. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standardization Branch, ANM–113.

Note 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Standardization Branch, ANM–113.

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on March 22, 1996.

Darrell M. Pederson,
Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 96–7548 Filed 3–27–96; 8:45 am]

BILLING CODE 4910–13–P

14 CFR Part 39

[Docket No. 95–NM–252–AD]

Airworthiness Directives; Fokker Model F28 Mark 0100 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking (NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Fokker Model F28 Mark 0100 series airplanes. This proposal would require replacement of certain flexible oxygen hoses, located in the flight compartment gangway and in the consoles, with insulated hose assemblies. This proposal is prompted by reports of either insufficient or no clearance between these hoses and adjacent structure or electrical wiring. The actions specified by the proposed AD are intended to prevent chafing of the flexible oxygen hoses, which could result in an uncontrollable loss of oxygen from the flightcrew oxygen system, and could allow the presence of oxygen in areas where ignition is possible.

DATES: Comments must be received by May 6, 1996.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–103, Attention: Rules Docket No. 95–NM–252–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Fokker Aircraft USA, Inc., 1199 North Fairfax Street, Alexandria, Virginia

22314. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Tim Dulin, Aerospace Engineer, Standardization Branch, ANM–113, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055–4056; telephone (206) 227–2141; fax (206) 227–1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 95–NM–252–AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM–103, Attention: Rules Docket No. 95–NM–252–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056.

Discussion

The Rijksluchtvaartdienst (RLD), which is the airworthiness authority for the Netherlands, recently notified the FAA that an unsafe condition may exist on certain Fokker Model F28 Mark 0100 series airplanes. The RLD advises that it has received reports indicating that, on airplanes in production, no clearance was found to exist between the flexible

oxygen hose located in the flight compartment gangway and the adjacent electrical wiring. In addition, it was reported that no clearance was found between the flexible oxygen hoses installed in the left-hand (LH) and right-hand (RH) side consoles and the air conditioning ducts, and there was insufficient clearance between these hoses and the adjacent electrical wiring. These conditions, if not corrected, could result in chafing of the flexible oxygen hoses and the subsequent uncontrollable loss of oxygen from the flightcrew oxygen system. It also could result in the presence of oxygen in areas where ignition is possible.

Fokker has issued Service Bulletin SBF100-35-004, dated May 17, 1995, which describes procedures for replacement of the flexible oxygen hoses with insulated hose assemblies. The insulated hose assemblies are intended to prevent chafing and damage of the hoses. The RLD classified this service bulletin as mandatory and issued Dutch airworthiness directive BLA 1995-050 (A), dated May 31, 1995, in order to assure the continued airworthiness of these airplanes in the Netherlands.

This airplane model is manufactured in the Netherlands and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the RLD has kept the FAA informed of the situation described above. The FAA has examined the findings of the RLD, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design, the proposed AD would require replacement of flexible oxygen hoses with insulated hose assemblies. The actions would be required to be accomplished in accordance with the service bulletin described previously.

The FAA estimates that 21 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 4 work hours per airplane to accomplish the proposed actions, and that the average labor rate is \$60 per work hour. Required parts would cost approximately \$1,376 per airplane. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be \$33,936, or \$1,616 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Fokker: Docket 95-NM-252-AD.

Applicability: Model F28 Mark 0100 series airplanes, having serial numbers 11244 through 11321 inclusive, and 11323 through 11332 inclusive; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been

otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent an uncontrollable loss of oxygen from the flightcrew oxygen system due to chafing of the flexible oxygen hoses, which could result in the presence of oxygen in areas where ignition is possible; accomplish the following:

(a) Within 12 months or 3,000 flight cycles after the effective date of this AD, whichever occurs first, replace the flexible oxygen hoses having part number (P/N) A66152-407, located in the left-hand (LH) and right-hand (RH) side consoles with insulated tube assemblies having P/N D66127-401; and replace the flexible oxygen assemblies having P/N A66152-417, located in the flight compartment gangway with insulated tube assemblies having P/N D66127-403; in accordance with Fokker Service Bulletin SBF100-35-004, dated May 17, 1995.

(b) As of the effective date of this AD, no person shall install a hose assembly with P/N A66152-417 or A66152-407, on any airplane.

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standardization Branch, ANM-113.

NOTE 2: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Standardization Branch, ANM-113.

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on March 22, 1996.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

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