See the **ADDRESSES** section for a location to examine the regulatory evaluation.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

■ 2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

**2006–04–14 Boeing:** Amendment 39–14496. Docket No. FAA–2005–23282; Directorate Identifier 2005–NM–210–AD.

#### Effective Date

(a) This AD becomes effective April 3, 2006.

#### Affected ADs

(b) None.

#### **Applicability**

(c) This AD applies to airplanes identified in paragraphs (c)(1) and (c)(2) of this AD, certificated in any category.

(1) Boeing Model 757–200 series airplanes, having certain variable numbers as identified in Boeing Special Attention Service Bulletin 757–21–0106, dated March 24, 2005.

(2) Boeing Model 757–300 series airplanes, having certain variable numbers as identified in Boeing Special Attention Service Bulletin 757–21–0107, dated March 24, 2005.

#### **Unsafe Condition**

(d) This AD results from finding that the end caps of the overhead distribution ducts for the air conditioning system were not bonded to the ducts with an adhesive. We are issuing this AD to detect and correct loosened end caps, which could change the air flow balance in the airplane. During a smoke event in the cargo or main electronics compartment, the incorrect balance of air flow could change the smoke clearance air capacity and result in smoke and toxic fumes penetrating the flight deck and main cabin.

### Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

### Service Bulletin References

(f) The term "service bulletin," as used in this AD, means the Accomplishment Instructions of the following service bulletins, as applicable: (1) For Model 757–200 series airplanes: Boeing Special Attention Service Bulletin 757–21–0106, dated March 24, 2005; and

(2) For Model 757–300 series airplanes: Boeing Special Attention Service Bulletin 757–21–0107, dated March 24, 2005.

#### **Install Clamps**

(g) Within 12,000 flight hours or 36 months after the effective date of this AD, whichever is first: Install clamps on the end caps of the overhead distribution ducts of the air conditioning system at stations 864.88, 864.9, 866.6, and 875, as applicable, and before further flight do other specified and related investigative actions as applicable, by doing all of the applicable actions specified in the service bulletin.

# Alternative Methods of Compliance (AMOCs)

(h)(1) The Manager, Seattle Aircraft Certification Office, FAA, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with § 39.19 on any airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

#### Material Incorporated by Reference

(i) You must use Boeing Special Attention Service Bulletin 757-21-0106, dated March 24, 2005; or Boeing Special Attention Service Bulletin 757-21-0107, dated March 24, 2005; as applicable, to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of these documents in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Boeing Commercial Airplanes, P.O. Box 3707, Seattle, Washington 98124-2207, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street SW., room PL-401, Nassif Building, Washington, DC; on the Internet at http://dms.dot.gov; or at the National Archives and Records Administration (NARA).

For information on the availability of this material at the NARA, call (202) 741–6030, or go to http://www.archives.gov/federal\_register/code\_of\_federal\_regulations/ibr\_locations.html.

Issued in Renton, Washington, on February 15, 2006.

### Michael Zielinski,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 06–1694 Filed 2–24–06; 8:45 am]

#### BILLING CODE 4910-13-P

#### **DEPARTMENT OF TRANSPORTATION**

#### **Federal Aviation Administration**

#### 14 CFR Part 39

[Docket No. FAA-2006-23594; Directorate Identifier 2005-NE-54-AD; Amendment 39-14497; AD 2006-04-15]

#### RIN 2120-AA64

Airworthiness Directives; Turbomeca Artouste III B, Artouste III B1, and Artouste III D Turboshaft Engines

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule; request for comments.

**SUMMARY:** The FAA is adopting a new airworthiness directive (AD) for Turbomeca Artouste III B, Artouste III B1, and Artouste III D turboshaft engines. This AD requires removing certain fuel pumps from service and installing serviceable fuel pumps. This AD results from a report that an acceptance test facility used test equipment that was out of calibration, on certain fuel pumps, and those fuel pumps might have been accepted with a limitation in the maximum available fuel flow. We are issuing this AD to prevent reduced helicopter performance, subsequent loss of control of the helicopter, or accident.

**DATES:** Effective March 14, 2006.

We must receive any comments on this AD by April 28, 2006.

**ADDRESSES:** Use one of the following addresses to comment on this AD:

- DOT Docket Web site: Go to http://dms.dot.gov and follow the instructions for sending your comments electronically.
- Government-wide rulemaking Web site: Go to <a href="http://www.regulations.gov">http://www.regulations.gov</a> and follow the instructions for sending your comments electronically.
- Mail: Docket Management Facility;
  U.S. Department of Transportation, 400
  Seventh Street, SW., Nassif Building,
  Room PL-401, Washington, DC 20590-0001.
  - Fax: (202) 493–2251.
- Hand Delivery: Room PL—401 on the plaza level of the Nassif Building, 400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Turbomeca, 40220 Tarnos, France; telephone +33 05 59 74 40 00, fax +33 05 59 74 45 15, for the service information identified in this AD.

#### FOR FURTHER INFORMATION CONTACT:

Christopher Spinney, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803–5299; telephone (781) 238–7175; fax (781) 238–7199.

SUPPLEMENTARY INFORMATION: The Direction Generale de L'Aviation Civile (DGAC), which is the airworthiness authority for France, notified the FAA that an unsafe condition may exist on Turbomeca Artouste III B, Artouste III B1, and Artouste III D turboshaft engines. The DGAC advises that an acceptance test facility used test equipment that was out of calibration, on 102 fuel pumps, and those fuel pumps might have been accepted with a limitation in the maximum available fuel flow. This condition causes fuel flow limitation and therefore reduces the maximum available engine power over a portion of the helicopter flight envelope. These pumps may be installed on Eurocopter France Alouette III SE.3160, SA.316B, SA.315B, and SA.316C helicopters registered in the U.S. Turbomeca issued Mandatory Service Bulletin No. 218 73 0802, dated November 17, 2005, to address the 102 suspect fuel pumps. We cannot confirm that these fuel pumps have been removed from service and retested or replaced. The DGAC issued AD No. F-2005–201, dated December 7, 2005, in order to ensure the airworthiness of these engines in France.

## **Bilateral Airworthiness Agreement**

These Turbomeca Artouste III series turboshaft engines are manufactured in France and are type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Under this bilateral airworthiness agreement, the DGAC kept the FAA informed of the situation described above. We have examined the findings of the DGAC, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United

# FAA's Determination and Requirements of This AD

The unsafe condition described previously is likely to exist or develop on other Turbomeca Artouste III B, Artouste III B1, and Artouste III D turboshaft engines of the same type design. We are issuing this AD to prevent reduced helicopter performance, subsequent loss of control of the helicopter, or accident. This AD requires removing affected fuel pumps from service and installing serviceable fuel pumps, within 30 days or 80

operating hours after receipt of a serviceable fuel pump, whichever occurs first, but no later than March 15, 2006.

## FAA's Determination of the Effective Date

Since an unsafe condition exists that requires the immediate adoption of this AD, we have found that notice and opportunity for public comment before issuing this AD are impracticable, and that good cause exists for making this amendment effective within 30 days.

## **Comments Invited**

This AD is a final rule that involves requirements affecting flight safety and was not preceded by notice and an opportunity for public comment; however, we invite you to send us any written relevant data, views, or arguments regarding this AD. Send your comments to an address listed under ADDRESSES. Include "AD Docket No. FAA-2006-23594; Directorate Identifier 2005-NE-54-AD" in the subject line of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the rule that might suggest a need to modify it.

We will post all comments we receive, without change, to http:// dms.dot.gov, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this AD. Using the search function of the DMS Web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review the DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78) or you may visit http://dms.dot.gov.

## **Examining the AD Docket**

You may examine the docket that contains the AD, any comments received, and any final disposition in person at the Docket Management Facility Docket Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Office (telephone (800) 647–5227) is located on the plaza level of the Department of Transportation Nassif Building at the street address stated in ADDRESSES. Comments will be available in the AD docket shortly after the DMS receives them.

### **Authority for This Rulemaking**

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in subtitle VII, part A, subpart III, section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

## **Regulatory Findings**

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the regulation:

- 1. Is not a "significant regulatory action" under Executive Order 12866;
- 2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- 3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a summary of the costs to comply with this AD and placed it in the AD Docket. You may get a copy of this summary by sending a request to us at the address listed under **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

## Adoption of the Amendment

■ Under the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

■ 2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**2006–04–15 Turbomeca:** Amendment 39–14497. Docket No. FAA–2006–23594; Directorate Identifier 2005–NE–54–AD.

#### **Effective Date**

(a) This airworthiness directive (AD) becomes effective March 14, 2006.

#### Affected ADs

(b) None.

#### **Applicability**

(c) This AD applies to Turbomeca Artouste III B, Artouste III B1, and Artouste III D turboshaft engines. These engines are installed on, but not limited to, Eurocopter France Alouette III SE.3160, SA.316B, SA.315B, and SA.316C helicopters.

### **Unsafe Condition**

(d) This AD results from a report that an acceptance test facility used test equipment that was out of calibration, on certain fuel pumps, and those fuel pumps might have been accepted with a limitation in the maximum available fuel flow. We are issuing this AD to prevent reduced helicopter performance, subsequent loss of control of the helicopter, or accident.

#### Compliance

(e) You are responsible for having the actions required by this AD performed within 30 days or 80 operating hours after the receipt of a serviceable fuel pump, whichever occurs first, but no later than March 15, 2006, unless the actions have already been done.

(f) Remove from service the fuel pumps listed by serial number (SN) in the following Table 1, and install a serviceable fuel pump.

TABLE 1.—AFFECTED FUEL PUMP SNS

A59B	F504B	2827
A82B	F506B	2828
A91B	F537B	2830
B14B	F561B	2838
B29B B42B	F589B F596B	2854
C27B	F607B	2867 2868
C6B	F630B	2884
C92B	F643B	2944
D16B	F706B	3078
D18B	F724B	3175
D20B	F743B	3230
D80B	F745B	3259
D99B	F748B	3282
E49B	F759B	3343
E77B	F760B	3376
E90B	F762B	3383
F112B	F957B	3385
F131B	808	3397
F176B	1725	3458
F220B F243B	1766	3515
F243B F253B	1770 1897	3548 3660
F262B	1897	3746
F293B	2154	3756
F317B	2155	3757
F320B	2233	3783
		2. 30

# TABLE 1.—AFFECTED FUEL PUMP SNS—Continued

F357B	2512	3792
F368B	2620	3826
F420B	2729	3858
F464B	2759	3888
F466B	2763	3894
F477B	2786	3979
F47B	2787	4066

#### Definition

- (g) For the purpose of this AD, a serviceable fuel pump is:
- (1) A fuel pump that is not listed in Table 1 of this AD; or
- (2) A fuel pump that is listed in Table 1 of this AD that has passed a repeat of the original production acceptance test.

### **Alternative Methods of Compliance**

(h) The Manager, Engine Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

#### **Related Information**

- (i) Direction Generale de L'Aviation Civile AD No. F-2005-201, dated December 7, 2005, also addresses the subject of this AD.
- (j) Turbomeca Mandatory Service Bulletin No. 218 73 0802, dated November 17, 2005, pertains to the subject of this AD.

Issued in Burlington, Massachusetts, on February 17, 2006.

#### Peter A. White.

Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service. [FR Doc. 06–1728 Filed 2–24–06; 8:45 am] BILLING CODE 4910–13–P

### **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

### 14 CFR Part 71

[Docket No. FAA-2005-22398; Airspace Docket No. 05-ASO-7]

RIN 2120-AA66

## Establishment of High Altitude Area Navigation Routes; South Central United States

AGENCY: Federal Aviation Administration (FAA), DOT. ACTION: Final rule; withdrawal.

SUMMARY: This action withdraws a final rule published in the Federal Register on February 15, 2006 (71 FR 7845), Docket No. FAA-2005-22398; Airspace Docket No. 05-ASO-7. This was an incorrect copy inadvertently sent to the Federal Register. The incorrect final rule is being withdrawn as a result of this error. The correct final rule was published February 13, 2006 (71 FR 7409), establishing 16 high altitude area

navigation routes in the South Central United States.

**DATES:** Effective Date: 0901 UTC, February 27, 2006.

FOR FURTHER INFORMATION CONTACT: Paul Gallant, Airspace and Rules, Office of System Operations and Safety, Federal Aviation Administration, 800 Independence Avenue, SW., Washington, DC 20591; telephone: (202) 267–8783.

SUPPLEMENTARY INFORMATION: On February 13, 2006, the FAA published in the Federal Register (71 FR 7409) a final rule establishing 16 high altitude area navigation routes in the South Central United States. On February 15, 2006, the FAA inadvertently published in the Federal Register an obsolete version of the final rule, which contained outdated fix names (71 FR 7845). This action withdraws the incorrect final rule published in error on February 15, 2006. The rule published on February 13, 2006 (71 FR 7409) contains the correct information.

#### Withdrawal of Final Rule

Accordingly, pursuant to the authority delegated to me, Docket No. FAA–2005–22398; Airspace Docket No. 05–ASO–7; as published in the **Federal Register** February 15, 2006 (71 FR 7845), is hereby withdrawn.

Issued in Washington, DC, on February 17, 2006.

#### Edith V. Parish,

Manager, Airspace and Rules. [FR Doc. 06–1760 Filed 2–24–06; 8:45 am] BILLING CODE 4910–13–P

#### **DEPARTMENT OF TRANSPORTATION**

#### **Federal Aviation Administration**

### 14 CFR Part 71

[Docket No. FAA-2005-22509; Airspace Docket No. 03-AWA-2]

RIN 2120-AA66

# Modification of the St. Louis Class B Airspace Area; MO

**AGENCY:** Federal Aviation Administration (FAA), DOT. **ACTION:** Final rule; correction.

**SUMMARY:** This action corrects a final rule published in the **Federal Register** on February 15, 2006 (71 FR 7848), Airspace Docket No. 03–AWA–2, FAA Docket No. FAA–2005–22509. In that rule, inadvertent errors were made in the airport description of the St. Louis Class B airspace area. This action corrects those errors.