

TABLE 1.—PART NUMBERS—Continued

Door type	Door location	Goodrich slide/slide raft part number
A	2, LH	7A1539-001, -003, -005, -007, -013, -015, -017, -101, -103, -105, -107, -109, -113, -115, or -117.
A	2, RH	7A1539-002, -004, -006, -008, -014, -016, -018, -102, -104, -106, -108, -110, -114, -116, or -118.
A	3, LH	7A1510-001, -003, -005, -007, -013, -015, -017, -101, -103, -105, -107, -109, -113, -115, or -117; or 4A3934-1 or -3.
A	3, RH	7A1510-002, -004, -006, -008, -014, -016, -018, -102, -104, -106, -108, -110, -114, -116, or -118; or 4A3934-2 or -4.
1	3, LH and RH	7A1509-001, -003, -005, -101, -103, -105, -107, -109, -111, -115, or -117.
1	3, LH	4A3928-1.
1	3, RH	4A3928-2.

Parts Installation

(c) As of the effective date of this AD, no person may install a regulator valve having a part number listed in the old part number column specified in Paragraph 1.L. of the applicable service bulletin on any airplane, unless that regulator valve has been modified in accordance with paragraph (b) of this AD.

Alternative Methods of Compliance

(d) In accordance with 14 CFR 39.19, the Manager, International Branch, ANM-116, Transport Airplane Directorate, FAA, is authorized to approve alternative methods of compliance for this AD.

Note 1: The subject of this AD is addressed in French airworthiness directive F-2004-094 R1, dated February 16, 2004.

Incorporation by Reference

(e) Unless otherwise specified in this AD, the actions must be done in accordance with Airbus Service Bulletin A330-25-3225, Revision 01, dated September 30, 2004; Airbus Service Bulletin A340-25-4228, Revision 01, dated September 30, 2004; or Airbus Service Bulletin A340-25-5054, dated August 2, 2004; as applicable. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. To get copies of this service information, contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. To inspect copies of this service information, go to the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or to the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741-6030, or go to http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Effective Date

(f) This amendment becomes effective on March 24, 2006.

Issued in Renton, Washington, on February 1, 2006.

Ali Bahrami,

Manager, Transport Airplane Directorate,
Aircraft Certification Service.

[FR Doc. 06-1411 Filed 2-16-06; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2005-22411; Directorate Identifier 2005-NM-074-AD; Amendment 39-14482; AD 2006-04-01]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A300 B2 Series Airplanes; Model A300 B4 Series Airplanes; Model A300 B4-600 Series Airplanes; Model A300 B4-600R Series Airplanes; Model A300 F4 600R Series Airplanes; Model A300 C4-605R Variant F Airplanes; and Model A310-200 Series Airplanes; and Model A310-300 Series Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Airbus transport category airplanes. This AD requires replacing the existing cabin altitude indicator in the cabin pressure control panel with a new, improved cabin altitude indicator. This AD results from a report of injuries occurring on in-service airplanes when crewmembers forcibly initiated opening of passenger/crew doors against residual pressure causing the doors to rapidly open. We are issuing this AD to prevent injury to crewmembers, and subsequent damage to the airplane caused by rapid opening of the door.

DATES: This AD becomes effective March 24, 2006.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in the AD as of March 24, 2006.

ADDRESSES: You may examine the AD docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street,

SW., Nassif Building, Room PL-401, Washington, DC.

Contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France, for service information identified in this AD.

FOR FURTHER INFORMATION CONTACT: Tom Stafford, Aerospace Engineer, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1622; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Examining the Docket

You may examine the airworthiness directive (AD) docket on the Internet at <http://dms.dot.gov> or in person at the Docket Management Facility office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Management Facility office (telephone (800) 647-5227) is located on the plaza level of the Nassif Building at the street address stated in the **ADDRESSES** section.

Discussion

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to certain Airbus transport category airplanes. That NPRM was published in the **Federal Register** on September 14, 2005 (70 FR 54321). That NPRM proposed to require replacing the existing cabin altitude indicator in the cabin pressure control panel with a new, improved cabin altitude indicator.

Comments

We provided the public the opportunity to participate in the development of this AD. We have considered the comments received.

Agreement With the NPRM

One commenter states that it agrees with the proposed requirements of the NPRM.

Request To Extend the Compliance Time

One commenter requests that the compliance time of the NPRM be extended from 22 months to 34 months. The commenter points out that a compliance time of 34 months would correlate with the French airworthiness directive that mandated the service bulletins specified in the NPRM.

We agree that the compliance time may be extended. Our intention was that the compliance time of this AD should correlate with the French airworthiness directive. We have revised the AD to specify 34 months.

Request To Revise the “Costs of Compliance” Section

This same commenter requests that the estimated cost of the NPRM be revised from \$1,071 to \$3,081, per airplane. The commenter states that it can provide the \$3,081 dollar figure based on that fact that it has accomplished the modification on several airplanes.

We do not agree that the “Costs of Compliance” section should be revised. Based on the information provided in the new Revision 2 of the service bulletins specified in this AD, there is no change in the costs that were estimated in the NPRM. The cost analysis in AD rulemaking actions typically does not include incidental costs such as the time required to gain access and close up, time necessary for planning, or time necessitated by other administrative actions. Those incidental costs, which may vary significantly among operators, are almost impossible to calculate. No change to this AD is necessary in this regard.

Request To Limit Applicability of the NPRM

One commenter, a freighter operator, requests that the FAA consider the differences in operation between passenger and cargo operations. The commenter states that its normal flightcrew procedures require that the flightcrew open the entry door, and this event occurs after engine shutdown. The commenter states that its cabin residual warning system would alert the flightcrew if a differential pressure existed within the cabin. The commenter further notes that its flightcrew ground procedures do not refer to the cabin altitude indicator for any purpose.

We infer that the commenter is requesting that we remove cargo airplanes from the applicability of the AD. We do not agree that the requirements of this AD do not apply to

cargo airplanes. We point out that cargo operator procedures may change and are not universal in application. In addition, in the event a new operator buys an airplane, the existing cargo operator procedures are not legally required to be used by the new operator. No change to this AD is necessary in this regard. However, under the provisions of paragraph (i) of the AD, we may consider requests for approval of an alternative method of compliance if sufficient data are submitted to substantiate that such a method would provide an acceptable level of safety.

Request To Address Defective Parts Manufacturer Approval (PMA) Parts

One commenter requests that the NPRM be revised to cover possible defective PMA alternative parts, rather than just a single part number, so that those defective PMA parts also are subject to the proposed AD. The commenter states that PMA manufacturers are encouraged—and in some cases, required—to identify PMA parts by alternative designations.

We concur with the commenter’s general request that, if we know that an unsafe condition also exists in PMA parts, the AD should address those parts, as well as the original parts. We are not aware of other PMA parts that have a different part number. The commenter’s remarks are timely in that the Transport Airplane Directorate currently is in the process of reviewing this issue as it applies to transport category airplanes. We acknowledge that there may be other ways of addressing this issue to ensure that unsafe PMA parts are identified and addressed. Once we have thoroughly examined all aspects of this issue, including input from industry, and have made a final determination, we will consider whether our policy regarding addressing PMA parts in ADs needs to be revised. We consider that to delay this AD action would be inappropriate, since we have determined that an unsafe condition exists and that replacement of certain parts must be accomplished to ensure continued safety. Therefore, no change has been made to the final rule in this regard.

Request To Reference Parts with PMA

The same commenter also requests that the language in the NPRM be changed to permit installation of other FAA-approved equivalent parts. The commenter states that the mandated installation of a certain part number “is at variance with FAR 21.303,” which permits the installation of other (PMA) parts.

We infer that the commenter would like the AD to permit installation of any equivalent PMA part so that it is not necessary for an operator to request approval of an alternative method of compliance (AMOC) in order to install an “equivalent” PMA part. Whether an alternative part is “equivalent” in adequately resolving the unsafe condition can only be determined on a case-by-case basis based on a complete understanding of the unsafe condition. We are not currently aware of any such parts. Our policy is that, in order for operators to replace a part with one that is not specified in the AD, they must request an AMOC. This is necessary so that we can make a specific determination that an alternative part is or is not susceptible to the same unsafe condition.

In response to the commenter’s statement regarding a “variance with FAR 21.303,” under which the FAA issues parts with a PMA, this statement appears to reflect a misunderstanding of the relationship between ADs and the certification procedural regulations of part 21 of the FARs (14 CFR part 21). Those regulations, including section 21.303 of the FARs (14 CFR part 21.303), are intended to ensure that aeronautical products comply with the applicable airworthiness standards. But ADs are issued when, notwithstanding those procedures, we become aware of unsafe conditions in these products or parts. Therefore, an AD takes precedence over design “approvals” when we identify an unsafe condition, and mandating installation of a certain part number in an AD is not at variance with section § 21.303.

The AD provides a means of compliance for operators to ensure that the identified unsafe condition is addressed appropriately. For an unsafe condition attributable to a part, the AD normally identifies the replacement parts necessary to obtain that compliance. As stated in section 39.7 of the FARs (14 CFR 39.7), “Anyone who operates a product that does not meet the requirements of an applicable airworthiness directive is in violation of this section.” Unless an operator obtains approval for an AMOC, replacing a part with one not specified by the AD would make the operator subject to an enforcement action and result in a civil penalty. No change to the AD is necessary in this regard.

Request To Require Latest Revision of the Service Bulletin

One commenter, the manufacturer, advises that it has issued new revisions to each of the service bulletins referenced in the NPRM. The

commenter also states that the new revision level was issued to inform operators that the service information was mandatory. The commenter requests that the NPRM be revised to reflect the new revision level of each of the service bulletins.

We agree that the AD should reflect the new revision level of the service bulletins and have revised the AD accordingly.

Request To Give "Credit" for a Service Bulletin

The same commenter, the manufacturer, notes that paragraph (g) of the NPRM gives credit for accomplishing the actions of Airbus Service Bulletin A310-21-2063 and A300-21-0131, both dated September 9, 2004. The commenter requests that credit also be given for accomplishing the actions specified in Airbus Service Bulletin A310-21-0131, dated September 9, 2004.

We agree to revise paragraph (g) of the AD. We have added a new Table 2, which specifies not only the requested additional service bulletin revision, but also Revision 01 for each of the service bulletins.

Clarification of Alternative Method of Compliance (AMOC) Paragraph

We have revised this action to clarify the appropriate procedure for notifying the principal inspector before using any approved AMOC on any airplane to which the AMOC applies.

Conclusion

We have carefully reviewed the available data, including the comments received, and determined that air safety and the public interest require adopting the AD with the changes described previously. We have determined that these changes will neither increase the economic burden on any operator nor increase the scope of the AD.

Costs of Compliance

This AD will affect about 194 airplanes of U.S. registry. The required actions will take about 7 work hours per airplane, at an average labor rate of \$65 per work hour. Required parts will cost about \$1,246 per airplane. Based on these figures, the estimated cost of this AD for U.S. operators is \$329,994, or \$1,701 per airplane.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We have determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

- (1) Is not a "significant regulatory action" under Executive Order 12866;
- (2) Is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
- (3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. The Federal Aviation Administration (FAA) amends § 39.13 by adding the following new airworthiness directive (AD):

2006-04-01 Airbus: Amendment 39-14482. Docket No. FAA-2005-22411; Directorate Identifier 2005-NM-074-AD.

Effective Date

(a) This AD becomes effective March 24, 2006.

Affected ADs

(b) None.

Applicability

(c) This AD applies to Airbus Model A300 B2-1A, B2-1C, B2K-3C, and B2-203 airplanes; Model A300 B4-2C, B4-103, and B4-203 airplanes; Model A300 B4-601, B4-603, B4-620, and B4-622 airplanes; Model A300 B4-605R and B4-622R airplanes; Model A300 F4-605R and F4-622R airplanes; Model A300 C4-605R Variant F airplanes; Model A310-203, -204, -221, and -222 airplanes; and Model A310-304, -322, -324, and -325 airplanes; certificated in any category; except for those airplanes on which Airbus Modification 12857 has been incorporated in production.

Unsafe Condition

(d) This AD was prompted by a report of injuries occurring on in-service airplanes when crewmembers forcibly initiated opening of passenger/crew doors against residual pressure causing the doors to rapidly open. We are issuing this AD to prevent injury to crewmembers, and subsequent damage to the airplane caused by the rapid opening of the door.

Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified, unless the actions have already been done.

Replacing the Cabin Altitude Indicator

(f) Within 34 months after the effective date of this AD, replace the cabin altitude indicator (Part Number (P/N) 37000-3) in the cabin pressure control panel with a new improved cabin altitude indicator (P/N 37000-3-01), in accordance with the service bulletins specified in Table 1 of this AD, as applicable.

TABLE 1.—AIRBUS SERVICE BULLETINS

Model	Service bulletin and revision number	Date
Model A300 B2–1A, B2–1C, B2K–3C, and B2–203 airplanes; Model A300 B4–2C, B4–103, and B4–203 airplanes; Model A300 B4–601, B4–603, B4–620, and B4–622 airplanes.	A300–21–0131, Revision 02	April 20, 2005.
Model A300 B4–605R and B4–622R airplanes; Model A300 F4–605R and F4–622R airplanes; Model A300 C4–605R Variant F airplanes.	A300–21–6050, Revision 02	April 20, 2005.
Model A310–304, –322, –324, and –325 airplanes	A310–21–2063, Revision 02	April 20, 2005.

Note 1: The service bulletins specified in Table 1 of paragraph (f) of this AD describe installation of an in-service modification equivalent to production Modification 12857.

Additional Source of Service Information

Note 2: Each of the service bulletins specified in Table 1 of paragraph (f) of this

AD refers to Thales Service Bulletin 37000–3–21–001, dated October 8, 2004, as an additional source of service information.

Actions Accomplished in Accordance With Previous Service Information

(g) Replacement of the cabin altitude indicator with a new, improved indicator, in

accordance with the Airbus service bulletins specified in Table 2 of this AD, as applicable, before the effective date of this AD, is acceptable for compliance with the requirements of paragraph (f) of this AD.

TABLE 2.—AIRBUS SERVICE BULLETINS ACCEPTABLE FOR COMPLIANCE

Model	Service bulletin and revision number	Date
Model A300 B2–1A, B2–1C, B2K–3C, and B2–203 airplanes; Model A300 B4–2C, B4–103, and B4–203 airplanes; Model A300 B4–601, B4–603, B4–620, and B4–622 airplanes.	A300–21–0131, (original)	September 9, 2004.
	A300–21–0131, Revision 01	January 6, 2005.
Model A300 B4–605R and B4–622R airplanes; Model A300 F4–605R and F4–622R airplanes; Model A300 C4–605R Variant F airplanes.	A300–21–6050, (original)	September 9, 2004.
	A300–21–6050, Revision 01	December 17, 2004.
Model A310–304, –322, –324, and –325 airplanes	A310–21–2063, (original)	September 9, 2004.
	A310–21–2063, Revision 01	January 6, 2005.

Parts Installation

(h) After the effective date of this AD, no person may install a Thales cabin altitude indicator having P/N 37000–3 on any airplane.

Alternative Methods of Compliance (AMOCs)

(i)(1) The Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate, has the authority to approve AMOCs for this AD, if requested in accordance with the procedures found in 14 CFR 39.19.

(2) Before using any AMOC approved in accordance with 14 CFR 39.19 on any

airplane to which the AMOC applies, notify the appropriate principal inspector in the FAA Flight Standards Certificate Holding District Office.

Related Information

(j) French airworthiness directive F–2005–027, dated February 16, 2005, also addresses the subject of this AD.

Material Incorporated by Reference

(k) You must use the service information specified in Table 3 of this AD to perform the actions that are required by this AD, unless the AD specifies otherwise. The Director of the Federal Register approved the incorporation by reference of these

documents in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Contact Airbus, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France, for a copy of this service information. You may review copies at the Docket Management Facility, U.S. Department of Transportation, 400 Seventh Street, SW., Room PL–401, Nassif Building, Washington, DC; on the Internet at <http://dms.dot.gov>; or at the National Archives and Records Administration (NARA). For information on the availability of this material at the NARA, call (202) 741–6030, or go to http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

TABLE 3.—MATERIAL INCORPORATED BY REFERENCE

Airbus service bulletin	Revision level	Date
A300–21–0131	02	April 20, 2005.
A300–21–6050	02	April 20, 2005.
A310–21–2063	02	April 20, 2005.

Issued in Renton, Washington, on February 1, 2006.

Ali Bahrami,

*Manager, Transport Airplane Directorate,
Aircraft Certification Service.*

[FR Doc. 06-1412 Filed 2-16-06; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Parts 47 and 49

[Docket No.: FAA-2004-19944; Amendment Nos. 47-27 and 49-10]

RIN 2120-AI48

Cape Town Treaty Implementation

AGENCY: Federal Aviation Administration (FAA), DOT.

ACTION: Final rule; confirmation of effective date; approval for collection of public information.

SUMMARY: This document confirms the effective date of the January 3, 2005, final rule amending 14 CFR parts 47 and 49 to comply with the Cape Town Treaty Implementation Act of 2004. This document also confirms the approval by the Office of Management and Budget (OMB) for the collection of public information contained in the final rule.

EFFECTIVE DATE: The effective date for the final rule amending 14 CFR parts 47 and 49 published at 70 FR 240, January 3, 2005, is March 1, 2006.

FOR FURTHER INFORMATION CONTACT: Mr. Mark D. Lash, Civil Aviation Registry, Mike Monroney Aeronautical Center, 6500 South MacArthur Blvd., Oklahoma City, OK 73169, telephone (405) 954-4331.

SUPPLEMENTARY INFORMATION:

Confirmation of Effective Date

The FAA published a final rule, with a request for comment on the information collection requirements, in the **Federal Register** on January 3, 2005 (70 FR 240). This final rule revised the regulations concerning registering aircraft and recording security documents as directed by Section 4 of the Cape Town Treaty Implementation Act of 2004. The Cape Town Treaty (the Treaty) establishes a new International Registry for registering interests against certain aircraft and aircraft engines. These amendments enable persons to send information to the International Registry concerning certain aircraft and aircraft engines by making the FAA Aircraft Registry the U.S. authorizing entry point to the International Registry.

When published, the final rule indicated that these amendments become effective concurrent with the date the Treaty enters into force with respect to the United States. Under the terms of the Treaty, it enters into force three months after the eighth country deposits formal instruments with the International Institute for the Unification of Private Law depositary in Rome. FAA advised that it would publish a document announcing the effective date of this final rule. As of January 1, 2006, eight countries, including the United States, have deposited instruments of ratification. Thus, the Treaty enters into force with respect to the United States and the final rule becomes effective on March 1, 2006.

OMB Approval for the Collection of Public Information

When published, the final rule indicated that the FAA had submitted the information requirements associated with this final rule to OMB with a request for clearance. The effective date for the collection of this public information should be concurrent with the date the Treaty enters into force with respect to the United States, once approved by OMB. An agency may not conduct or sponsor, and a person is not required to respond to, a collection of information unless it displays a currently valid OMB control number. The OMB control number for this collection is 2120-0697. Notwithstanding any other provisions of law, no person is subject to any penalty for failing to comply with a collection of information subject to the Paperwork Reduction Act (PRA) that does not display a valid control number.

OMB Control Number: 2120-0697.

OMB Approval Date: 03/29/05.

OMB Expiration Date: 03/31/08.

Title: FAA Entry Point Filing Form—International Registry.

Form Number: AC Form 8050-135.

Respondents: Law firms, technical level personnel (paralegals), and the public.

Number of Respondents: 15,000.

Estimated Time per Response: 30 minutes.

Needs and Uses: Public Law 108-297 designates the FAA Aircraft Registry as the U.S. entry point for authorizing the transmission of information relating to civil aircraft of the United States, aircraft for which a United States identification has been assigned (but only with respect to notices of prospective assignments, interests, and sales), and aircraft engines, to the International Registry. It also provides for the filing of notices of prospective

interests. To transmit certain types of interests or prospective interests to the International Registry, interested parties must file a completed FAA Entry Point Filing Form—International Registry, AC Form 8050-135, with the FAA Civil Aviation Registry. Upon receipt of the completed form, the FAA Civil Aviation Registry will issue the unique authorization code. The FAA did not receive any adverse comments on the proposed information collection, and it was approved by OMB.

Issued in Washington, DC on February 10, 2006.

Anthony F. Fazio,

Director, Office of Rulemaking.

[FR Doc. 06-1484 Filed 2-16-06; 8:45 am]

BILLING CODE 4910-13-P

DEPARTMENT OF HEALTH AND HUMAN SERVICES

Food and Drug Administration

21 CFR Part 522

Implantation or Injectable Dosage Form New Animal Drugs; Estradiol Benzoate

AGENCY: Food and Drug Administration, HHS.

ACTION: Final rule.

SUMMARY: The Food and Drug Administration (FDA) is amending the animal drug regulations to reflect approval of a supplemental new animal drug application (NADA) filed by PR Pharmaceuticals, Inc. The supplemental NADA provides for subcutaneous injection, in the ear only, of a suspension implant of estradiol benzoate microspheres for increased rate of weight gain in suckling beef calves. It also adds the indication for use for increased rate of weight gain in steers fed in confinement for slaughter, previously approved at a lower dose, to the higher approved dose level.

DATES: This rule is effective February 17, 2006.

FOR FURTHER INFORMATION CONTACT: Eric S. Dubbin, Center for Veterinary Medicine (HFV-126), Food and Drug Administration, 7500 Standish Pl., Rockville, MD 20855, 301-827-0232, e-mail: eric.dubbin@fda.hhs.gov.

SUPPLEMENTARY INFORMATION: PR Pharmaceuticals, Inc., 1716 Heath Pkwy., Fort Collins, CO 80524, filed a supplement to NADA 141-040 that provides for use of DURALEASE (estradiol benzoate) Microencapsulated Suspension Implant by subcutaneous injection in the ear for increased rate of