

*Requested change 6:* Finally, Boeing comments that “A significant amount of the Model 747 main gear truck and axle assembly is designed by ground turn. Additionally, the axle stiffness, which is a very important parameter for brake interaction and for tire shoulder wear, could be negatively affected if the requirements are reduced. By lowering the loads below current practice, new service-related problems could result.”

*FAA response:* The special conditions require the applicant to demonstrate that the reduced ground turning load cannot be exceeded in service. If the applicant can demonstrate this and can demonstrate compliance with other regulations affecting the integrity of landing gear, brakes, and tires, we consider that the potential for new service-related problems would be minimized. Nevertheless, as with any other type design, the FAA continually monitors the safety of airplanes in the operating fleet and has the means to require mandatory corrective actions, if warranted.

Accordingly, the special conditions are adopted, as proposed, with a minor clarifying change to the text of subparagraph b.

#### Applicability

As discussed above, these special conditions are applicable to the Airbus A380–800 airplane. Should Airbus apply at a later date for a change to the type certificate to include another model incorporating the same novel or unusual design features, these special conditions would apply to that model as well under the provisions of § 21.101.

#### Conclusion

This action affects only certain novel or unusual design features of the Airbus A380–800 airplane. It is not a rule of general applicability.

#### List of Subjects in 14 CFR Part 25

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

The authority citation for these special conditions is as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

#### The Special Conditions

Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the type certification basis for the Airbus A380–800 airplane.

In lieu of the requirements of § 25.495, the following special condition applies:

a. The airplane is assumed to execute a steady turn by steering of any steerable

gear or by application of any differential power. The airplane limit vertical load factor must be 1.0, and, in the absence of a more rational analysis, the limit airplane lateral load factor must be 0.5.

b. The airplane is assumed to be in static balance, the lateral load factor being reacted by friction forces applied at the ground contact point of each tire. The lateral load must be shared between each individual tire in a rational or conservative manner. The distribution of the load among the tires must account at least for the effects of the factors specified in subparagraph c. (2) of this special condition.

c. At maximum ramp weight, a limit value of lateral center of gravity (cg) inertia load factor lower than specified in subparagraph a. but not less than 0.45g (wing axis) may be used, if it can be shown by a rational analysis that this lower value cannot be exceeded. The rational analysis must consider at least the following:

1. The maximum lateral load factor that can be reached during the full range of likely ground operations at maximum ramp weight, including ground turning, “fishtailing,” and high-speed runway exit. In each case, the full dynamic maneuver must be considered.

2. The rational analysis must include at least the following parameters:

(a) Landing gear spring curves and landing gear kinematics.

(b) Reliable tire friction characteristics.

(c) Airframe and landing gear flexibility when significant.

(d) Airplane rigid body motion.

(e) The worst combination of tire diameter, tire pressure, and runway shapes, specified in §§ 25.511(b)(2), 25.511(b)(3), and 25.511(b)(4).

d. The limit lateral load factor at maximum landing weight is 0.5.

e. Details of the analysis and any assumptions used must be agreed to by the FAA. Any assumptions made in the analysis must be based on the intrinsic characteristics of the airplane and must be independent of airfield geometry. Other influences that cannot be controlled by the airplane design must be conservatively assessed.

Issued in Renton, Washington, on November 9, 2006.

**Ali Bahrami,**

Manager, Transport Airplane Directorate, Aircraft Certification Service.

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**BILLING CODE 4910–13–P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 25

[Docket No. NM354; Special Conditions No. 25–336–SC]

#### Special Conditions: Boeing Commercial Airplane Group, Boeing Model 777 Series Airplane; Overhead Cross Aisle Stowage Compartments

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final special conditions.

**SUMMARY:** These special conditions are issued for the Boeing Model 777 series airplanes. This airplane will have novel or unusual design features associated with overhead cross aisle stowage compartments. The applicable airworthiness regulations do not contain adequate or appropriate safety standards for these design features. These special conditions contain the additional safety standards the Administrator considers necessary to establish a level of safety equivalent to that established by the existing airworthiness standards.

**EFFECTIVE DATE:** November 15, 2006.

**FOR FURTHER INFORMATION CONTACT:** Jayson Claar, FAA, Airframe/Cabin Branch, ANM–115, Transport Airplane Directorate, Aircraft Certification Service, 1601 Lind Avenue SW., Renton, Washington, 98057–3356; telephone (425) 227–2194; facsimile (425) 227–1232.

#### SUPPLEMENTARY INFORMATION:

##### Background

On April 20, 2005, Boeing Commercial Airplane Group, Seattle, Washington, applied for a supplemental type certificate to permit installation of overhead cross aisle stowage compartments in Boeing 777 series airplanes. The Boeing Model 777 series airplanes are large twin engine airplanes with four or five pairs of Type A exits. The Boeing 777 airplanes can be configured with various passenger capacities and ranges.

The regulations do not address the novel and unusual design features associated with the installation of overhead cross aisle stowage compartments installed on the Boeing Model 777, making these special conditions necessary. Generally, the requirements for overhead stowage compartments are similar to stowage compartments in remote crew rest compartments (i.e., located on lower lobe, main deck or overhead) already in use on Boeing Model 777 and 747 series

airplanes. Remote crew rest compartments have been previously installed and certified in the main passenger cabin area, above the main passenger area, and below the passenger cabin area adjacent to the cargo compartment of the Boeing Model 777–200, and –300 series airplanes.

#### **Type Certification Basis**

Under the provisions of § 21.101, Boeing Commercial Airplane Group must show that the Boeing Model 777, as changed, continues to meet the applicable provisions of the regulations incorporated by reference in Type Certificate No. T00001SE or the applicable regulations in effect on the date of application for the change. The regulations incorporated by reference in the type certificate are commonly referred to as the “original type certification basis.” The regulations incorporated by reference in Type Certificate No. T00001SE for the Boeing Model 777 series airplanes include Title 14 Code of Federal Regulations (CFR), part 25, as amended by Amendments 25–1 through 25–100, with exceptions, for various models. Refer to Type Certificate No. T00001SE, as applicable, for a complete description of the certification basis for this model, including certain special conditions that are not relevant to these special conditions.

If the Administrator finds the applicable airworthiness regulations (part 25 as amended) do not contain adequate or appropriate safety standards for the Boeing Model 777 because of a novel or unusual design feature, special conditions are prescribed under the provisions of § 21.16.

In addition to the applicable airworthiness regulations and special conditions, the Boeing Model 777 must comply with the fuel vent and exhaust emission requirements of 14 CFR part 34 and the noise certification requirements of 14 CFR part 36.

The FAA issues special conditions, as defined in § 11.19, under § 11.38, and they become part of the type certification basis under § 21.101.

Special conditions are initially applicable to the model for which they are issued. Should the applicant apply for a change to modify any other model included on the same type certificate to incorporate the same or similar novel or unusual design feature, the special conditions would also apply to the other model under § 21.101.

#### **Novel or Unusual Design Features**

The Boeing Model 777 will incorporate the following novel or unusual design features: the installation

of powered lift-enabled stowage compartments that rise into the overhead area and lower into the cabin.

The overhead cross aisle stowage compartments are configured to allow stowage of galley type standard containers as well as coats, bags, and other items typically stowed in closets or bins. These stowage compartments may be located above the emergency exit cross aisles of Boeing Model 777 series airplanes. Because the compartment is lowered into the main cabin, it could affect egress if it cannot be raised again. The overhead compartment may lower into a cross aisle as defined in § 25.813, but it may also lower into other potential egress paths. For the purposes of these special conditions, the same criteria apply, whether or not the egress path is required by § 25.813. Therefore, as used in these special conditions, the term “overhead cross aisle stowage compartment” addresses all such compartments.

Each stowage compartment is accessed from the main deck by a powered lift that lowers and raises the stowage compartment between the overhead and the main deck. In addition, the lift can be hand cranked down and up in the event of a power or lift motor failure. A smoke detection system will be provided in the overhead cross aisle stowage compartments.

#### **Discussion of the Special Conditions**

In general, the requirements listed in these special conditions for overhead cross aisle stowage compartments are similar to those previously approved for overhead crew rest compartments in earlier certification programs, such as for the Boeing Model 777 and Model 747 series airplanes. These special conditions establish compartment access, power lift, electrical power, smoke/fire detection, fire extinguisher, fire containment, smoke penetration, and compartment design criteria for the overhead cross aisle stowage compartments. The overhead stowage compartments are not a direct analogy to stowage compartments in remote crew rest compartments installed and certified for Boeing Model 777 series airplanes, but the safety issues raised are similar. Features similar to those considered in the development of previous special conditions for fire protection will be included here also. The requirements provide an equivalent level of safety to that provided by other Boeing Model 777 series airplanes with similar overhead compartments.

#### **Operational Evaluations and Approval**

The FAA's Aircraft Certification Service will administer these special conditions, which specify requirements for design approvals (that is, type design changes and supplemental type certificates) of overhead cross aisle stowage compartments.

The Aircraft Evaluation Group of the FAA's Flight Standards Service must evaluate and approve the operational use of overhead cross aisle stowage compartments prior to use. The Aircraft Evaluation Group must receive all instructions for continued airworthiness, including service bulletins, prior to the FAA accepting and issuing approval of the modification.

#### **Special Condition No. 1, Compartment Access and Placards**

Appropriate placards, or other means, are required to address door access and locking to prohibit or prevent passenger access, and operation of the overhead storage compartment. There must also be a means to preclude anyone from being trapped inside the stowage compartment, if it is large enough for a person to enter. If there is more than one door providing access, each door must be equipped with these means.

#### **Special Condition No. 2, Power Lift**

The power lift must be designed so the overhead stowage compartment will not jam in the down position, even if lowered on top of a hard structure. The lift must operate at a speed, and stop above the floor at such a height, that allows anyone underneath the compartment to move clear without injury. The lift controls must be placed clear of the compartment door and must be pressed continuously for lift operation. Training on power lift operation procedures must be added to appropriate manuals.

#### **Special Condition No. 3, Manual Operation**

There must be a means to manually operate the lift that is independent of the electrical drive system. The lift must be operable by a range of occupants, including a fifth percentile female. The manual means must be capable of lowering the overhead stowage compartment quickly to the main deck to fight a fire. The manual system must be capable of raising the compartment quickly so the cross aisle or other egress path (if applicable) is not blocked in an emergency. If electrical or manual power is removed, there must be a means, such as a brake, to prevent the compartment from unrestricted movement, i.e., falling. Training on

manual operation procedures must be added to appropriate manuals.

*Special Condition No. 4, Handheld Fire Extinguisher*

For compartments larger than 25 cubic feet, a handheld fire extinguisher appropriate to fight the kinds of fire likely to occur in the overhead stowage compartment must be provided. This handheld fire extinguisher must be adjacent to the overhead compartment. This extinguisher must be in addition to those required for the passenger cabin.

*Special Condition No. 5, Fire Containment*

This special condition requires either the installation of a manually activated fire extinguishing system that is accessible from outside the overhead stowage compartment, or a demonstration that the crew could satisfactorily perform the function of extinguishing a fire under the prescribed conditions. A manually activated built-in fire extinguishing system would be required only if a crewmember could not successfully locate and get access to the fire during a demonstration where the crewmember is responding to the alarm. For the duration of the flight, the system must have adequate capacity to suppress any fire occurring in the stowage compartment considering the fire threat, volume of the compartment, and the ventilation rate.

*Special Condition No. 6, Smoke Penetration*

The design of the compartment must provide means to exclude hazardous quantities of smoke or extinguishing agent originating in the compartment from entering other occupied areas. The means must take into account the time period during which the compartment may be accessed to manually fight a fire, if applicable.

Smoke entering any other compartment occupied by crewmembers or passengers, when access to the stowage compartment is opened to manually fight a fire, must dissipate within five minutes after the access to the stowage compartment is closed.

During the one-minute smoke detection time (see Special Condition No. 7), penetration of a small quantity of smoke (one that would dissipate within 3 minutes under normal ventilation conditions) from this overhead stowage compartment into an occupied area on this airplane configuration would be acceptable based on the limitations placed in this and other associated special conditions. These special conditions place

sufficient restrictions in the quantity and type of material allowed in the overhead stowage compartment that threat from a fire in this remote area would be equivalent to that experienced on the main cabin.

If a built-in fire extinguishing system is used in lieu of manual fire fighting, then the fire extinguishing system must be designed so that no hazardous quantities of extinguishing agent will enter other compartments occupied by passengers or crew.

*Special Condition No. 7, Compartment Design Criteria*

The material used to construct the overhead stowage compartment must meet the flammability requirements for compartment interiors in § 25.853 and be fire resistant. Depending on the size of the compartment, certain fire protection features of Class B cargo compartments are also required. Enclosed stowage compartments equal to or exceeding 25 ft<sup>3</sup> in interior volume must be provided with a smoke or fire detection system to ensure that a fire can be detected within a one-minute detection time. This is the same requirement as has been applied to remote crew rest compartments.

Enclosed stowage compartments equal to or greater than 57 ft<sup>3</sup> in interior volume but less than or equal to 200 ft<sup>3</sup>, must have a liner that meets the requirements of § 25.855 for a Class B cargo compartment. The overhead stowage compartment may not be greater than 200 ft<sup>3</sup> in interior volume. The in-flight accessibility of very large enclosed stowage compartments and the subsequent impact on the crewmember's ability to effectively reach any part of the compartment with the contents of a handheld fire extinguisher would require additional fire protection considerations similar to those required for inaccessible compartments such as Class C cargo compartments.

The overhead stowage compartment smoke or fire detection and fire suppression systems (including airflow management features which prevent hazardous quantities of smoke or fire extinguishing agent from entering any other compartment occupied by crewmembers or passengers) is considered complex in terms of paragraph 6d of Advisory Circular (AC) 25.1309-1A, "System Design and Analysis." The FAA considers failure of the overhead stowage compartment fire protection system (that is, smoke or fire detection and fire suppression systems) in conjunction with an overhead stowage fire to be a catastrophic event. Based on the "Depth of Analysis

Flowchart" shown in Figure 2 of AC 25.1309-1A, the depth of analysis should include both qualitative and quantitative assessments (reference paragraphs 8d, 9, and 10 of AC 25.1309-1A).

The requirements to enable crewmember(s) quick access to the overhead stowage compartment and to locate a fire source inherently places limits on the amount of baggage stowed and the size of the overhead stowage compartment. The overhead stowage compartment is limited to stowage of galley type standard containers as well as coats, bags, and other items typically stowed in closets or bins. It is not intended to be used for the stowage of other items. The design of such a system to include other items may require additional special conditions to ensure safe operation.

**Discussion of Comments**

Notice of proposed special conditions No. 25-06-09-SC for the Boeing Model 777-200 series airplanes was published in the **Federal Register** on October 18, 2006 (71 FR 61432). An amended proposed notice of special conditions No. SC-06-29A-SC for the Boeing Model 777 series airplanes was published in the **Federal Register** on November 2, 2006 (71 FR 64478). No comments were received, and the special conditions are adopted as proposed, except for clarifying changes.

**Applicability**

These special conditions are applicable to the Boeing Model 777 series airplanes with overhead cross aisle stowage compartments. Should Boeing Commercial Airplane Group apply later for a change to Type Certificate No. T00001SE to include another model on the same type certificate incorporating the same novel or unusual design feature, the special conditions would apply to that model as well.

Under standard practice, the effective date of final special conditions would be 30 days after the date of publication in the **Federal Register**, however, as the certification date for the Boeing 777 series is imminent, the FAA finds that good cause exists for make these special conditions effective upon issuance.

**Conclusion**

This action affects only certain novel or unusual design features on the Boeing Model 777 series airplanes. It is not a rule of general applicability and affects only the applicant who applied to the FAA for approval of these features on the airplane.

**List of Subjects in 14 CFR Part 25**

Aircraft, Aviation safety, Reporting and recordkeeping requirements.

The authority citation for these special conditions is as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701, 44702, 44704.

**The Special Conditions**

Accordingly, pursuant to the authority delegated to me by the Administrator, the following special conditions are issued as part of the type certification basis for Boeing Model 777 series airplanes. Each overhead cross aisle stowage compartment and the adjacent area, including the structural frame, mechanical system and drive motor, must meet the following requirements:

1. *Compartment Access and Placards.* There must be a means to prohibit or prevent passengers from entering or operating the overhead cross aisle stowage compartment. Placards prohibiting access are acceptable. If a compartment is large enough for a person to enter, there must be a means to preclude anyone from being trapped inside the stowage compartment. If a latching/locking mechanism is installed, the door must be capable of being opened from the outside without the aid of special tools. The mechanism must not prevent opening from the inside of the stowage at any time.

2. *Power Lift.* There must be a means such as a load or force limiter to protect the overhead cross aisle stowage compartment power lift from failure or jamming in the down position in the event it is lowered on top of a hard structure such as a galley cart.

(a) The lift controls must be placed so the operator is clear of the lift and designed such that the controls must be pressed continuously for lift operation.

(b) The lift must raise and lower the stowage compartment at a slow enough rate, and stop above the floor at such a height, that anyone underneath can easily move clear without injury.

(c) Stowage compartment operation training procedures must be added to the appropriate flight attendant manuals.

3. *Manual Lift.* There must be a means in the event of failure of the aircraft's main power system, or of the powered overhead cross aisle stowage

compartment lift system, for manually activating the lift system.

(a) This manual means must be independent of the electrical drive system

(b) The manual means must be accessible and operable by a range of occupants, including a fifth percentile female.

(c) The manual means must be capable of lowering the stowage compartment to the main deck quickly enough to fight a fire in the stowage compartment before overhead cross aisle stowage compartment fire containment is compromised.

(d) The manual means must be capable of quickly raising the stowage compartment such that the cross aisle, or other egress path is not blocked in the event of an emergency.

(e) Stowage compartment firefighting training procedures must be added to the appropriate manuals.

(f) The lift system must include a means, such as a brake, to retain the overhead cross aisle stowage compartment in any position of travel when the manual or electric drive force is removed.

4. *Fire Extinguisher.* The means to manually fight a fire in the overhead cross aisle stowage compartment must consider the additional stowage volume and time required to manually lower the compartment after indication. For compartments larger than 25 ft<sup>3</sup> the following equipment must be provided directly adjacent to each overhead cross aisle stowage compartment: at least one approved handheld fire extinguisher, in addition to the fire extinguisher requirements of § 25.851 and § 121.309, appropriate for the kinds of fires likely to occur within the overhead stowage compartment.

5. *Fire Containment.* Fires originating within the overhead cross aisle stowage compartment must be controlled for the duration of the flight without a crewmember having to access the compartment. Alternatively, the design of the access provisions must allow crewmembers equipped for firefighting to have unrestricted access to the compartment. If the latter approach is elected it must be demonstrated that a crewmember has sufficient access to enable them to extinguish a fire. The time for a crewmember on the main deck to react to the fire alarm, (and, if applicable, to don the firefighting

equipment and to open the compartment) must not exceed the flammability and fire containment capabilities of the stowage compartment.

6. *Smoke Penetration.* There must be a means provided to exclude hazardous quantities of smoke or extinguishing agent originating in the overhead cross aisle stowage compartment from entering any other compartment occupied by crewmembers or passengers. If access is required to comply with Special Condition No. 5, this means must include the time period when accessing the stowage compartment to manually fight a fire. Smoke entering any other compartment occupied by crewmembers or passengers, when access to the stowage compartment is opened to manually fight a fire, must dissipate within five minutes after the access to the stowage compartment is closed. Prior to the one minute smoke detection time (reference note 2 in paragraph (7)) penetration of a small quantity of smoke (one that would dissipate within 3 minutes under normal ventilation conditions) from the stowage compartment into an occupied area is acceptable. Flight tests must be conducted to show compliance with this requirement.

7. *Compartment Design Criteria.* The overhead cross aisle stowage compartment must be designed to minimize the hazards to the airplane in the event of a fire originating in the stowage compartment.

(a) *Fire Extinguishing System.* If a built-in fire extinguishing system is used in lieu of manual firefighting, then the fire extinguishing system must be designed so no hazardous quantities of extinguishing agent will enter other compartments occupied by passengers or crew. The system must have adequate capacity to suppress any fire occurring in the stowage compartment, considering the fire threat, volume of the compartment, and the ventilation rate.

(b) *Compartment Size.* All overhead cross aisle stowage compartments must meet the design criteria given in the table below. As indicated by the table below, enclosed stowage compartments greater than 200 ft<sup>3</sup> in interior volume are not addressed by this special condition.

STOWAGE COMPARTMENT INTERIOR VOLUMES

Fire protection features	Less than 25 ft <sup>3</sup>	25 ft <sup>3</sup> to 57 ft <sup>3</sup>	57 ft <sup>3</sup> to 200 ft <sup>3</sup>
Materials of Construction <sup>1</sup> .....	Yes .....	Yes .....	Yes.
Detectors <sup>2</sup> .....	No .....	Yes .....	Yes.

## STOWAGE COMPARTMENT INTERIOR VOLUMES—Continued

Fire protection features	Less than 25 ft <sup>3</sup>	25 ft <sup>3</sup> to 57 ft <sup>3</sup>	57 ft <sup>3</sup> to 200 ft <sup>3</sup>
Liner <sup>3</sup> .....	No .....	Yes .....	Yes.

<sup>1</sup> *Material*

The material used to construct each enclosed stowage compartment must be at least fire resistant and must meet the flammability standards established for interior components (that is, 14 CFR Part 25 Appendix F, Parts I, IV, and V) per the requirements of § 25.853. For compartments less than 25 ft<sup>3</sup> in total interior volume, the design must ensure the ability to contain a fire likely to occur within the compartment under normal use.

<sup>2</sup> *Detectors*

Enclosed stowage compartments equal to or exceeding 25 ft<sup>3</sup> in total interior volume must be provided with a smoke or fire detection system to ensure that a fire can be detected within one minute. Flight tests must be conducted to show compliance with this requirement. Each system (or systems) must provide:

- (a) A visual indication in the flight deck within one minute after the start of a fire;
- (b) A warning in the main passenger cabin. This warning must be readily detectable by a flight attendant, taking into consideration the positioning of flight attendants throughout the main passenger compartment during various phases of flight.

<sup>3</sup> *Liner*

If it can be shown the material used to construct the stowage compartment meets the flammability requirements of a liner for a Class B cargo compartment (that is, § 25.855 at Amendment 25–93 and Appendix F, part I, paragraph (a)(2)(ii)), in addition to the above <sup>1</sup> *Material requirement*, then no liner would be required for enclosed stowage compartments equal to or greater than 25 ft<sup>3</sup> in total interior volume but less than 57 ft<sup>3</sup> in total interior volume. For all enclosed stowage compartments equal to or greater than 57 ft<sup>3</sup> in total interior volume but less than or equal to 200 ft<sup>3</sup>, a liner must be provided that meets the requirements of § 25.855 for a Class B cargo compartment.

Issued in Renton, Washington, on November 15, 2006.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate,  
Aircraft Certification Service.*

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**BILLING CODE 4910–13–P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 43**

[Docket No.: FAA–2004–17683]

**RIN 2120–A119**

**Implementing the Maintenance Provisions of Bilateral Agreements**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Final rule; notice of effective date.

**SUMMARY:** The FAA is announcing the effective date of the final rule, published July 14, 2005, that amended the regulations governing maintenance, preventive maintenance, and alterations performed on U.S. aeronautical products by certain Canadian persons. That revision removes specific regulatory references and other requirements and requires that the maintenance, preventive maintenance, and alterations be performed in accordance with a Bilateral Aviation Safety Agreement (BASA) between the United States and Canada and associated Maintenance Implementation Procedures (MIP). When the rule was published, the FAA announced the amendments would become effective concurrent with the date the MIP entered into force. The MIP was signed

and entered into force on August 31, 2006; accordingly, the amendments became effective on that date.

**DATES:** The effective date of § 43.17 is August 31, 2006.

**FOR FURTHER INFORMATION CONTACT:** William D. Scott, Flight Standards, Aircraft Maintenance Division, AFS–300, Federal Aviation Administration, 800 Independence Avenue, SW., Washington, DC 20591; telephone (502) 753–4202; facsimile (502) 753–4232, e-mail: [william.d.scott@faa.gov](mailto:william.d.scott@faa.gov).

**SUPPLEMENTARY INFORMATION:****The Final Rule**

On July 14, 2005, the FAA issued a final rule amending § 43.17 of Title 14 of the Code of Federal Regulations (CFR), Maintenance, preventive maintenance, and alterations performed on U.S. aeronautical products by certain Canadian persons. (70 FR 40872). The United States and Canada had entered into an international agreement called a Bilateral Aviation Safety Agreement (BASA) that was in line with BASAs negotiated with other countries. The FAA and its Canadian counterpart, Transport Canada Civil Aviation (TCCA) of the Department of Transport, were negotiating Maintenance Implementation Procedures (MIP) to accompany the BASA. The amendment to § 43.17 removes specific regulatory references that if not removed would have constrained development of a standardized MIP.

The amendment also makes other minor changes and requires that all maintenance, preventive maintenance, and alterations performed by Canadian Approved Maintenance Organizations (AMOs) and Aviation Maintenance Engineers (AMEs) on U.S. aeronautical products be done in accordance with a

BASA between the United States and Canada and the associated MIP.

The MIP has been finalized. It was signed on August 31, 2006, and became effective immediately upon signing. In the preamble to the final rule the FAA stated, “These amendments become effective concurrent with the date the MIP accompanying the BASA between the United States and Canada enters into force.” Since the MIP is now final and entered into force on August 31, 2006, the FAA now sets the effective date for the above-referenced amendment to § 43.17 to be August 31, 2006.

The FAA has also prepared guidance material to assist maintenance providers in complying with the MIP. This guidance is contained in Advisory Circular (AC) AC 43–10B. A copy of the AC may be obtained by accessing the FAA’s Regulatory and Guidance Library Web page at [http://www.airweb.faa.gov/Regulatory\\_and\\_Guidance\\_Library/rqWebcomponents.nsf/HomeFrame?OpenFrameSet](http://www.airweb.faa.gov/Regulatory_and_Guidance_Library/rqWebcomponents.nsf/HomeFrame?OpenFrameSet).

**International Compatibility**

In keeping with U.S. obligations under the Convention on International Civil Aviation, it is FAA policy to comply with International Civil Aviation Organization (ICAO) Standards and Recommended Practices to the maximum extent practicable. The FAA has reviewed the corresponding ICAO Standards and Recommended Practices and has identified no differences with these regulations.

**Good Cause for Immediate Adoption**

In accordance with 5 U.S.C. 553(b)(3)(B), FAA finds good cause for issuing this rule without prior notice and comment. Seeking public comment