

Over a period of time, the alteration of one electronic control unit (ECU) electronic component can cause a rapid uncontrolled power increase. Several occurrences have already been reported during engine start or during engine warm-up. This condition, if not corrected, could result in the loss of control of the aircraft if the pilot fails to react appropriately by switching to the mechanical backup mode.

We are issuing this AD to prevent a rapid uncontrolled power increase and possible loss of control of the airplane.

#### Actions and Compliance

(e) Unless already done, do the following actions:

(1) Before further flight, check if the ECU has a P/N listed in the Applicability section of this AD and a serial number (SN) of 131 or below, except SNs 70, 71, 83, and 88. If it does, then do not operate the engine.

(2) Remove and replace the ECU with an ECU P/N SP01160089-3, using SMA Service Bulletin (SB) No. SB-01-76-005, dated December 15, 2006.

(3) No later than 30 days after the effective date of this AD, replace all remaining affected P/N ECUs with an ECU P/N SP01160089-3, using SMA SB No. SB-01-76-005, dated December 15, 2006.

(4) After the effective date of this AD, do not install a spare ECU having a P/N listed in the Applicability section of this AD as a replacement part on any SMA SR305-230 or SR305-230-1 engine.

#### FAA AD Differences

(f) This AD differs from the Mandatory Continuing Airworthiness Information (MCAI) and/or service information as follows:

(1) EASA AD No. 2007-0033 requires compliance with the AD by March 31, 2007.

(2) This AD, written later, requires compliance within 30 days after the effective date of the AD.

#### Other FAA AD Provisions

(g) Alternative Methods of Compliance: The Manager, Engine Certification Office, FAA, has the authority to approve alternative methods of compliance for this AD, if requested using the procedures found in 14 CFR 39.19.

#### Related Information

(h) Refer to EASA AD 2007-0033, dated February 13, 2007, for related information.

(i) Contact Christopher Spinney, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803; e-mail: [Christopher.spinney@faa.gov](mailto:Christopher.spinney@faa.gov); telephone (781) 238-7175; fax (781) 238-7199 for more information about this AD.

#### Material Incorporated by Reference

(j) You must use Societe de Motorisations Aeronautiques Service Bulletin No. SB-01-76-005, dated December 15, 2006, to do the actions required by this AD, unless the AD specifies otherwise.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Societe de Motorisations Aeronautiques, 10-12 Rue Didier Daurat, F-18021 Bourges, France—Telephone +33 (0) 2 4867 5600; Fax: +33 (0) 2 4850 0141; e-mail: [customer\\_services@smasr.com](mailto:customer_services@smasr.com).

(3) You may review service information copies at the FAA, New England Region, 12 New England Executive Park, Burlington, MA 01803; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Burlington, MA, on November 15, 2007.

**Peter A. White,**

*Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.*

[FR Doc. E7-22812 Filed 11-27-07; 8:45 am]

**BILLING CODE 4910-13-P**

## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

**[Docket No. FAA-2007-28987; Directorate Identifier 2007-NM-127-AD; Amendment 39-15269; AD 2007-24-03]**

**RIN 2120-AA64**

**Airworthiness Directives; Empresa Brasileira de Aeronautica S.A. (EMBRAER) Model EMB-135ER, -135KE, -135KL, and -135LR Airplanes and Model EMB-145, -145ER, -145MR, -145LR, -145XR, -145MP, and -145EP Airplanes**

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Final rule.

**SUMMARY:** We are adopting a new airworthiness directive (AD) for the products listed above. This AD results from mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as:

It has been found the development of cracks in the forward fuselage right hand (RH) side skin during full-scale fatigue tests. Those cracks may quickly reach their critical length, reducing the aircraft structural integrity, with possible rapid decompression of the aircraft.

We are issuing this AD to require actions to correct the unsafe condition on these products.

**DATES:** This AD becomes effective January 2, 2008.

The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of January 2, 2008.

**ADDRESSES:** You may examine the AD docket on the Internet at <http://www.regulations.gov> or in person at the U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue, SE., Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2125; fax (425) 227-1149.

#### SUPPLEMENTARY INFORMATION:

##### Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to include an AD that would apply to the specified products. That NPRM was published in the **Federal Register** on August 16, 2007 (72 FR 45963). That NPRM proposed to correct an unsafe condition for the specified products. The MCAI states:

It has been found the development of cracks in the forward fuselage right hand (RH) side skin during full-scale fatigue tests. Those cracks may quickly reach their critical length, reducing the aircraft structural integrity, with possible rapid decompression of the aircraft.

The corrective action includes rework of the aircraft structure on the forward fuselage LH (left-hand) and RH sides. You may obtain further information by examining the MCAI in the AD docket.

#### Relevant Service Information

EMBRAER has issued Service Bulletin 145-53-0067, Revision 02, dated August 28, 2007. We referred to EMBRAER Service Bulletin 145-53-0067, Revision 01, dated February 27, 2007, as the appropriate source of service information for doing the actions specified in the NPRM. The procedures in Revision 02 of the service bulletin are essentially the same as those procedures in Revision 01. Revision 02 revises the illustrations and makes editorial changes. We have revised paragraph (f)(1) and Table 1 of this AD to also refer to Revision 02 of the service bulletin.

#### Comments

We gave the public the opportunity to participate in developing this AD. We considered the comment received.

#### Request To Extend Grace Period

American Eagle Airlines requests that we extend the grace period specified in

the NPRM. The commenter states that the compliance time of “prior to the accumulation of 22,000 total flight cycles or within 6 months after the effective date of this AD, whichever is later” would impose an excessive strain on the operator due to labor requirements and time out of service. The commenter notes that a number of its aircraft are near the 22,000 total flight cycle threshold and suggests that we change the grace period to within 2,000 flight cycles after the effective date of this AD.

We do not agree with the commenter’s request to extend the grace period. In developing an appropriate compliance time for this action, we considered the urgency associated with the subject unsafe condition, the availability of required parts, and the practical aspect of accomplishing the required modification within a period of time that corresponds to the normal scheduled maintenance for most affected operators. However, according to the provisions of paragraph (g) of the final rule, we may approve requests to adjust the compliance time if the request includes data that prove that the new compliance time would provide an acceptable level of safety. We have not changed this final rule in this regard.

### Conclusion

We reviewed the available data, including the comment received, and determined that air safety and the public interest require adopting the AD with the change described previously. We determined that this change will not increase the economic burden on any operator or increase the scope of the AD.

### Differences Between This AD and the MCAI or Service Information

We have reviewed the MCAI and related service information and, in general, agree with their substance. But we might have found it necessary to use different words from those in the MCAI to ensure the AD is clear for U.S. operators and is enforceable. In making these changes, we do not intend to differ substantively from the information provided in the MCAI and related service information.

We might also have required different actions in this AD from those in the MCAI in order to follow our FAA policies. Any such differences are highlighted in a Note within the AD.

### Costs of Compliance

We estimate that this AD will affect 624 products of U.S. registry. We also estimate that it will take about 60 work-hours per product to comply with the basic requirements of this AD. The

average labor rate is \$80 per work-hour. Required parts will cost about \$1,210 per product. Where the service information lists required parts costs that are covered under warranty, we have assumed that there will be no charge for these parts. As we do not control warranty coverage for affected parties, some parties may incur costs higher than estimated here. Based on these figures, we estimate the cost of this AD to the U.S. operators to be \$3,750,240, or \$6,010 per product.

### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

### Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify this AD:

1. Is not a “significant regulatory action” under Executive Order 12866;
2. Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this AD and placed it in the AD docket.

### Examining the AD Docket

You may examine the AD docket on the Internet at <http://www.regulations.gov>; or in person at the

Docket Operations office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Operations office (telephone (800) 647-5527) is in the **ADDRESSES** section. Comments will be available in the AD docket shortly after receipt.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### Adoption of the Amendment

- Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

- 1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

- 2. The FAA amends § 39.13 by adding the following new AD:

**2007-24-03 Empresa Brasileira de Aeronautica S.A. (EMBRAER):**  
Amendment 39-15269. Docket No. FAA-2007-28987; Directorate Identifier 2007-NM-127-AD.

#### Effective Date

- (a) This airworthiness directive (AD) becomes effective January 2, 2008.

#### Affected ADs

- (b) None.

#### Applicability

- (c) This AD applies to all EMBRAER Model EMB-135ER, -135KE, -135KL, and -135LR airplanes; and Model EMB-145, -145ER, -145MR, -145LR, -145XR, -145MP, and -145EP airplanes; certificated in any category.

#### Subject

- (d) Air Transport Association (ATA) of America Code 53: Fuselage.

#### Reason

- (e) The mandatory continuing airworthiness information (MCAI) states:

It has been found the development of cracks in the forward fuselage right-hand (RH) side skin during full-scale fatigue tests. Those cracks may quickly reach their critical length, reducing the aircraft structural integrity, with possible rapid decompression of the aircraft.

The corrective action includes rework of the aircraft structure on the forward fuselage LH (left-hand) and RH sides.

**Actions and Compliance**

(f) Prior to the accumulation of 22,000 total flight cycles, or within 6 months after the effective date of this AD, whichever is later, unless already done, do the following actions:

(1) Add two reinforcements to the forward fuselage skin on the LH and RH sides between frames 9 to 10 and 10 to 11, and stringers 12 to 15. Install supports to the reinforcements and stringers as well as new fasteners to the reinforcements and supports, and reroute the electrical wiring on the affected area. Do all actions in accordance with EMBRAER Service Bulletin 145-53-0067, Revision 01, dated February 27, 2007; or Revision 02, dated August 28, 2007.

(2) Accomplishing the detailed instructions and procedures described in the EMBRAER Service Bulletin 145-53-0051, dated July 15, 2004; or EMBRAER Service Bulletin 145-53-0051, Revision 01, dated February 7, 2006; is considered acceptable for compliance with the actions specified in this AD.

**FAA AD Differences**

**Note:** This AD differs from the MCAI and/or service information as follows: No differences.

**Other FAA AD Provisions**

(g) The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, International Branch, ANM-116, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98057-3356; telephone (425) 227-2125; fax (425) 227-1149. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the

FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.

(3) Reporting Requirements: For any reporting requirement in this AD, under the provisions of the Paperwork Reduction Act, the Office of Management and Budget (OMB) has approved the information collection requirements and has assigned OMB Control Number 2120-0056.

**Related Information**

(h) Refer to MCAI Brazilian Airworthiness Directive 2007-05-01R1, effective July 4, 2007, and the service bulletins listed in Table 1 of this AD, for related information.

**TABLE 1.—SERVICE BULLETINS**

EMBRAER Service Bulletin	Revision level	Date
145-53-0051 .....	Original .....	July 15, 2004.
145-53-0051 .....	01 .....	February 7, 2006.
145-53-0067 .....	01 .....	February 27, 2007.
145-53-0067 .....	02 .....	August 28, 2007.

**Material Incorporated by Reference**

(i) You must use the service information specified in Table 2 of this AD to do the

actions required by this AD, unless the AD specifies otherwise.

**TABLE 2.—MATERIAL INCORPORATED BY REFERENCE**

EMBRAER Service Bulletin	Revision level	Date
145-53-0051 .....	Original .....	July 15, 2004.
145-53-0051 .....	01 .....	February 7, 2006.
145-53-0067 .....	01 .....	February 27, 2007.
145-53-0067 .....	02 .....	August 28, 2007.

EMBRAER Service Bulletin 145-53-0051, Revision 01, dated February 7, 2006, has the following effective pages:

Page No.	Revision level shown on page	Date shown on page
1, 2 .....	01 .....	February 7, 2006.
3-129 .....	Original .....	July 15, 2004.

(1) The Director of the Federal Register approved the incorporation by reference of this service information under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) For service information identified in this AD, contact Empresa Brasileira de Aeronautica S.A. (EMBRAER), P.O. Box 343—CEP 12.225, Sao Jose dos Campos—SP, Brazil.

(3) You may review copies at the FAA, Transport Airplane Directorate, 1601 Lind Avenue SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call

(202) 741-6030, or go to: <http://www.archives.gov/federal-register/cfr/ibr-locations.html>.

Issued in Renton, Washington, on November 13, 2007.

**Ali Bahrami,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

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**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 97**

[Docket No. 30581; Amdt. No. 3246]

**Standard Instrument Approach Procedures; Miscellaneous Amendments**

**AGENCY:** Federal Aviation Administration (FAA), DOT.