

Issued in Fort Worth, Texas, on June 20, 2005.

S. Frances Cox,

*Acting Manager, Rotorcraft Directorate,  
Aircraft Certification Service.*

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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 39

[Docket No. FAA-2005-21242; Directorate Identifier 2005-NE-09-AD]

RIN 2120-AA64

#### Airworthiness Directives; Turbomeca Arriel 1B, 1D, 1D1 and 1S1 Turboshaft Engines

**AGENCY:** Federal Aviation Administration (FAA), Department of Transportation (DOT).

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** The Federal Aviation Administration (FAA) proposes to adopt a new airworthiness directive (AD) that is applicable to certain Turbomeca Arriel 1B, 1D, 1D1 and 1S1 turboshaft engines. This proposal would require initial and repetitive position checks of the gas generator 2nd stage turbine blades on all Turbomeca Arriel 1B, 1D, 1D1 and 1S1 turboshaft engines, and replacement of 2nd stage turbines on 1B and 1D1 engines only. This proposal is prompted by the release of gas generator 2nd stage turbine blades while in service, with full containment of debris. We are proposing this AD to prevent an uncommanded engine in flight shutdown.

**DATES:** We must receive any comments on this proposed AD by August 29, 2005.

**ADDRESSES:** Use one of the following addresses to comment on this proposed AD.

- DOT Docket Web site: Go to <http://dms.dot.gov> and follow the instructions for sending your comments electronically.
- Government-wide rulemaking Web site: Go to <http://www.regulations.gov> and follow the instructions for sending your comments electronically.
- Mail: Docket Management Facility; U.S. Department of Transportation, 400 Seventh Street, SW., Nassif Building, Room PL-401, Washington, DC 20590-0001.
- Fax: (202) 493-2251.
- Hand Delivery: Room PL-401 on the plaza level of the Nassif Building,

400 Seventh Street, SW., Washington, DC, between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays.

Contact Turbomeca, 40220 Tarnos, France; telephone +33 05 59 74 40 00, fax +33 05 59 74 45 15, for the service information identified in this proposed AD.

#### FOR FURTHER INFORMATION CONTACT:

Christopher Spinney, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone (781) 238-7175, fax (781) 238-7199.

#### SUPPLEMENTARY INFORMATION:

##### Comments Invited

We invite you to send us any written relevant data, views, or arguments regarding this proposal. Send your comments to an address listed under **ADDRESSES**. Include "Docket No. FAA-2005-21242; Directorate Identifier 2005-NE-09-AD" in the subject line of your comments. We specifically invite comments on the overall regulatory, economic, environmental, and energy aspects of the proposed AD. We will consider all comments received by the closing date and may amend the proposed AD in light of those comments.

We will post all comments we receive, without change, to <http://dms.dot.gov>, including any personal information you provide. We will also post a report summarizing each substantive verbal contact with FAA personnel concerning this proposed AD. Using the search function of the DMS Web site, anyone can find and read the comments in any of our dockets, including the name of the individual who sent the comment (or signed the comment on behalf of an association, business, labor union, etc.). You may review the DOT's complete Privacy Act Statement in the **Federal Register** published on April 11, 2000 (65 FR 19477-78) or you may visit <http://dms.dot.gov>.

#### Examining the AD Docket

You may examine the docket that contains the proposal, any comments received, and any final disposition in person at the DMS Docket Offices between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Office (telephone (800) 647-5227) is on the plaza level of the Department of Transportation Nassif Building at the street address stated in **ADDRESSES**. Comments will be available in the AD docket shortly after the DMS receives them.

#### Discussion

The Direction Generale de L'Aviation Civile (DGAC), which is the airworthiness authority for France, notified the FAA that an unsafe condition might exist on Turbomeca, Arriel 1B (modified per TU 148), 1D, 1D1 and 1S1 turboshaft engines. The DGAC advises that sixteen cases of release of gas generator 2nd stage turbine blades occurred in service, with full containment of debris. These events resulted in uncommanded engine in flight shutdown. Although terminating action is still unavailable, mandatory checks of the turbine blades and replacement of the turbine are being required in order to reduce the probability of an uncommanded engine in flight shutdown.

#### Relevant Service Information

We have reviewed and approved the technical contents of the following Turbomeca Alert Service Bulletins (ASBs), all dated March 24, 2004: ASB A292 72 0807, for Arriel 1B post TU 148; ASB A292 72 0808, for Arriel 1D; ASB A292 72 0809, for Arriel 1D1; and ASB A292 72 0810, for Arriel 1S1, that describe procedures for initial and repetitive position checks of the 2nd stage turbine blades, and replacement of 2nd stage turbines on 1B and 1D1 engines only. The DGAC classified these ASBs as mandatory and issued airworthiness directive F-2004-047, dated March 31, 2004, in order to ensure the airworthiness of these Turbomeca Arriel 1B, 1D, 1D1 and 1S1 turboshaft engines in France.

#### FAA's Determination and Requirements of the Proposed AD

These engines, manufactured in France, are type-certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. In keeping with this bilateral airworthiness agreement, the DGAC kept us informed of the situation described above. We have examined the DGAC's findings, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States. For this reason, we are proposing this AD, which would require initial and repetitive position checks of the 2nd stage turbine blades on Turbomeca Arriel 1B, 1D, 1D1 and 1S1 turboshaft engines, and replacement of 2nd stage turbines on 1B and 1D1 engines only. The proposed AD would require you to

use the service information described previously to perform these actions.

#### Interim Action

These actions are interim actions and we may take further rulemaking actions in the future.

#### Costs of Compliance

There are about 2,557 Turbomeca Arriel 1B, 1D, 1D1 and 1S1 turboshaft engines of the affected design in the worldwide fleet. We estimate that this proposed AD would affect 721 engines installed on helicopters of U.S. registry. We also estimate that it would take about 2 work hours per engine to inspect all 721 engines and 40 hours per engine to replace about 571 2nd stage turbines on 1B and 1D1 engines, and that the average labor rate is \$65 per work hour. Required parts would cost about \$3,200 per engine. Based on these figures, we estimate the total cost of the proposed AD to U.S. operators to be \$3,405,530.

#### Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA's authority to issue rules on aviation safety. Subtitle I, Section 106, describes the authority of the FAA Administrator. Subtitle VII, Aviation Programs, describes in more detail the scope of the Agency's authority.

We are issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701, "General requirements." Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on

products identified in this rulemaking action.

#### Regulatory Findings

We have determined that this proposed AD would not have federalism implications under Executive Order 13132. This proposed AD would not have a substantial direct effect on the States, on the relationship between the National Government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that the proposed regulation:

1. Is not a "significant regulatory action" under Executive Order 12866;
2. Is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and
3. Would not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

We prepared a regulatory evaluation of the estimated costs to comply with this proposed AD. See the **ADDRESSES** section for a location to examine the regulatory evaluation.

#### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

#### The Proposed Amendment

Under the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend 14 CFR part 39 as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. The FAA amends § 39.13 by adding the following new airworthiness directive:

**Turbomeca:** Docket No. FAA-2005-21242; Directorate Identifier 2005-NE-09-AD.

#### Comments Due Date

(a) The Federal Aviation Administration (FAA) must receive comments on this airworthiness directive (AD) action by August 29, 2005.

#### Affected ADs

(b) None.

#### Applicability

(c) This AD applies to Turbomeca Arriel 1B engines fitted with 2nd stage turbine modification TU 148, and Arriel 1D, 1D1 and 1S1 engines. Arriel 1B engines are installed on but not limited to Eurocopter France AS-350B and AS-350A "Ecureuil" helicopters; 1D engines are installed on but not limited to Eurocopter France AS-350B1 "Ecureuil" helicopters; 1D1 engines are installed on but not limited to Eurocopter France AS-350B2 "Ecureuil" helicopters; and Arriel 1S1 engines are installed on but not limited to Sikorsky Aircraft S-76A and S-76C helicopters.

#### Unsafe Condition

(d) This AD results from the release of gas generator 2nd stage turbine blades while in service, with full containment of debris. We are issuing this AD to prevent an uncommanded engine in flight shutdown.

#### Compliance

(e) You are responsible for having the actions required by this AD performed within the compliance times specified unless the actions have already been done.

#### Initial Relative Position Check of 2nd Stage Turbine Blades

(f) Do an initial relative position check of the 2nd stage turbine blades using the Turbomeca service bulletins (SBs) specified in the following Table 1 before reaching any of the intervals specified in Table 1 or within 50 hours time-in-service after the effective date of this AD, whichever occurs later.

TABLE 1.—INITIAL AND REPETITIVE RELATIVE POSITION CHECK INTERVALS OF 2ND STAGE TURBINE BLADE

| Turbomeca engine model           | Initial relative position check interval                                                                                                                       | Repetitive interval                                                          | Service bulletin                    |
|----------------------------------|----------------------------------------------------------------------------------------------------------------------------------------------------------------|------------------------------------------------------------------------------|-------------------------------------|
| Arriel 1B (modified per TI 148). | Within 1,200 hours time-since-new (TSN) or time-since-overhaul (TSO) or 3,500 cycles-since-new (CSN) or cycles-since-overhaul (CSO), whichever occurs earlier. | Within 200 hours time-in-service-since-last-relative-position check (TSLRPC) | A292 72 0807, dated March 24, 2004. |
| Arriel 1D .....                  | Within 1,200 hours TSN or TSO or 3,500 hours CSN or CSO, whichever occurs earlier.                                                                             | Within 200 hours TSLRPC .....                                                | A292 72 0808, dated March 24, 2004. |
| Arriel 1D1 .....                 | Within 1,200 hours TSN or TSO or 3,500 hours CSN or CSO, whichever occurs earlier.                                                                             | Within 150 hours TSLRPC .....                                                | A292 72 0809, dated March 24, 2004. |
| Arriel 1S1 .....                 | Within 1,200 hours TSN or TSO or 3,500 hours CSN or CSO, whichever occurs earlier.                                                                             | Within 150 hours TSLRPC .....                                                | A292 72 0810, dated March 24, 2004. |

**Credit for Previous Relative Position Checks**

(g) Relative position checks of 2nd stage turbine blades done using Turbomeca SB A292 72 0263, update 1, 2, 3, or 4, may be used to show compliance with the initial requirements of paragraph (f) of this AD.

**Repetitive Relative Position Check of 2nd Stage Turbine Blades**

(h) Recheck the relative position of 2nd stage turbine blades at the TSLRPC intervals specified in Table 1 of this AD, using the service bulletins indicated.

**Replace 2nd Stage Turbines on 1B and 1D1 Engines**

(i) Replace 2nd stage turbine with a new or overhauled 2nd stage turbine before accumulating 1,500 hours TSN or TSO for Arriel 1D1 engines, and 2,200 hours TSN or TSO for Arriel 1B engines, or by August 31, 2006, whichever occurs later. Overhauled Arriel 1D1 2nd stage turbines must be fitted with new blades; overhauled Arriel 1B 2nd stage turbines may be fitted with overhauled or new blades. Because this is an interim action, all turbines, including those that are new or overhauled, must continue to comply with relative position check requirements of paragraphs (f) and (h).

**Alternative Methods of Compliance**

(j) The Manager, Engine Certification Office, has the authority to approve alternative methods of compliance for this AD if requested using the procedures found in 14 CFR 39.19.

**Related Information**

(k) DGAC airworthiness directive F-2004-047, dated March 31, 2004, also addresses the subject of this AD.

Issued in Burlington, Massachusetts, on June 15, 2005.

**Francis A. Favara,**

*Acting Manager, Engine and Propeller Directorate, Aircraft Certification Service.*

[FR Doc. 05-12692 Filed 6-27-05; 8:45 am]

**BILLING CODE 4910-13-P**

**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 71**

[Docket No. FAA-2005-21607; Airspace; Docket No. 05-ACE-17]

**Proposed Establishment of Class E Airspace; Gardner, KS**

**AGENCY:** Federal Aviation Administration (FAA), DOT.

**ACTION:** Notice of proposed rulemaking.

**SUMMARY:** This notice proposes to establish Class E5 airspace at Gardner, KS.

**DATES:** Comments for inclusion in Rules Docket must be received on or before July 29, 2005.

**ADDRESSES:** Send comments on this proposal to the Docket Management System, U.S. Department of Transportation, Room Plaza 401, 400 Seventh Street, SW., Washington, DC 20590-0001. You must identify the docket number FAA-2005-21607/Airspace Docket No. 05-ACE-17, at the beginning of your comments. You may also submit comments on the Internet at <http://dms.dot.gov>. You may review the public docket containing the proposal, any comments received, and any final disposition in person in the Dockets Office between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The Docket Office (telephone 1-800-647-5527) is on the plaza level of the Department of Transportation NASSIF Building at the above address.

**FOR FURTHER INFORMATION CONTACT:**

Brenda Mumper, Air Traffic Division, Airspace Branch, ACE-520, DOT Regional Headquarters Building, Federal Aviation Administration, 901 Locust, Kansas City, MO 64106; telephone: (816) 329-2524.

**SUPPLEMENTARY INFORMATION:****Comments Invited**

Interested parties are invited to participate in this proposed rulemaking by submitting such written data, views, or arguments, as they may desire. Comments that provide the factual basis supporting the views and suggestions presented are particularly helpful in developing reasoned regulatory decisions on the proposal. Comments are specifically invited on the overall regulatory, aeronautical, economic, environmental, and energy-related aspects of the proposal. Communications should identify both docket numbers and be submitted in triplicate to the address listed above. Commenters wishing the FAA to acknowledge receipt of their comments on this notice must submit with those comments a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. FAA-2005-21607/Airspace Docket No. 05-ACE-17." The postcard will be date/time stamped and returned to the commenter.

**Availability of NPRM's**

An electronic copy of this document may be downloaded through the Internet at <http://dms.dot.gov>. Recently published rulemaking documents can also be accessed through the FAA's Web page at <http://www.faa.gov> or the Superintendent of Document's Web page at <http://www.access.gpo.gov/nara>.

Additionally, any person may obtain a copy of this notice by submitting a

request to the Federal Aviation Administration (FAA), Office of Air Traffic Airspace Management, ATA-400, 800 Independence Avenue, SW., Washington, DC 20591, or by calling (202) 267-8783. Communications must identify both docket numbers for this notice. Persons interested in being placed on a mailing list for future NPRM's should contact the FAA's Office of Rulemaking (202) 267-9677, to request a copy of Advisory Circular No. 11-2A, Notice of Proposed Rulemaking Distribution System, which describes the application procedure.

**The Proposal**

This notice proposes to amend Part 71 of the Federal Aviation Regulations (14 CFR part 71) by establishing a Class E airspace area extending upward from 700 feet above the surface at Gardner Municipal Airport, KS. A Class E airspace area overlies Gardner Municipal Airport, KS, however, its purpose and description are relative to Olathe, New Century Aircenter, KS and does not fully enclose the NDB or GPS-D Instrument Approach Procedures to Gardner Municipal Airport, KS. This proposal would correct this discrepancy by establishing a Class E airspace area extending upward from 700 feet above the surface within a 6.4 mile-radius of Gardner Municipal Airport, KS excluding that airspace within the Olathe, New Century Aircenter, KS Class D airspace. This will define airspace of appropriate dimensions to protect aircraft departing and executing instrument approach procedures to Gardner Municipal Airport and bring the airspace area into compliance with FAA directives. The area would be depicted on appropriate aeronautical charts.

Class E airspace areas extending upward from 700 feet or more above the surface of the earth are published in Paragraph 6005 of FAA Order 7400.9M, dated August 30, 2004, and effective September 16, 2004, which is incorporated by reference in 14 CFR 71.1. The Class E airspace designations listed in this document would be published subsequently in the Order.

The FAA has determined that this proposed regulation only involves an established body of technical regulations for which frequent and routine amendments are necessary to keep them operationally current. It, therefore, (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034; February 26, 1979); and (3) does not warrant preparation of a Regulatory Evaluation