

Data and Service Management, Dept. C1–L5A (D800–0024). Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Los Angeles Aircraft Certification Office, 3960 Paramount Boulevard, Lakewood, California; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741–6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Effective Date

(k) This amendment becomes effective on July 9, 2004.

Issued in Renton, Washington, on May 5, 2004.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service
[FR Doc. 04–12398 Filed 6–3–04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2002–NM–251–AD; Amendment 39–13655; AD 2004–11–09]

RIN 2120–AA64

Airworthiness Directives; Fokker Model F.28 Mark 0070 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Fokker Model F.28 Mark 0070 series airplanes, that requires inspection of cables installed on certain contactors in the electrical power center (EPC) for proper installation of wires, and reinstallation of wires if necessary. These actions are necessary to prevent a short circuit in the EPC, possibly leading to a fire in the main cabin and damage to the airplane, or injury to passengers and flightcrew. These actions are intended to address the identified unsafe condition.

DATES: Effective July 9, 2004.

The incorporation by reference of a certain publication listed in the regulations is approved by the Director of the Federal Register as of July 9, 2004.

ADDRESSES: The service information referenced in this AD may be obtained from Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at

the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741–6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

FOR FURTHER INFORMATION CONTACT: Tom Rodriguez, Aerospace Engineer; International Branch, ANM–116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055–4056; telephone (425) 227–1137; fax (425) 227–1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Fokker Model F.28 Mark 0070 series airplanes was published in the **Federal Register** on March 17, 2004 (69 FR 12580). That action proposed to require inspection of cables installed on certain contactors in the electrical power center for proper installation of wires, and reinstallation of wires if necessary.

Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

Cost Impact

The FAA estimates that 2 airplanes of U.S. registry will be affected by this AD, that it will take approximately 2 work hours per airplane to accomplish the required inspection, and that the average labor rate is \$65 per work hour. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$260, or \$130 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a “significant regulatory action” under Executive Order 12866; (2) is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

■ Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

■ 1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

■ 2. Section 39.13 is amended by adding the following new airworthiness directive:

2004–11–09 Fokker Services B.V.:

Amendment 39–13655. Docket 2002–NM–251–AD.

Applicability: Model F.28 Mark 0070 series airplanes, serial numbers 11521, and 11528 through 11585 inclusive; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent a short circuit in the electrical power center (EPC), possibly leading to a fire in the main cabin and damage to the airplane, or injury to passengers and flightcrew, accomplish the following:

Inspection, and Reinstallation if Necessary

(a) Within 6 months after the effective date of this AD, perform a general visual

inspection of the 4 contactors having part number 9124-9283 located in the EPC for proper installation of the wires; in accordance with the Accomplishment Instructions of Fokker Service Bulletin SBF100-24-035, dated May 27, 2002.

(1) If the installation is correct, no further action is required by this AD.

(2) If the installation is incorrect, prior to further flight, reinstall the wires in accordance with the Accomplishment Instructions of the service bulletin.

Note 1: For the purposes of this AD, a general visual inspection is defined as: "A visual examination of an interior or exterior area, installation, or assembly to detect obvious damage, failure, or irregularity. This level of inspection is made from within touching distance unless otherwise specified. A mirror may be necessary to enhance visual access to all exposed surfaces in the inspection area. This level of inspection is made under normally available lighting conditions such as daylight, hangar lighting, flashlight, or droplight and may require removal or opening of access panels or doors. Stands, ladders, or platforms may be required to gain proximity to the area being checked."

Exception to Service Bulletin Reporting

(b) Although Fokker Service Bulletin SBF100-24-035, dated May 27, 2002, specifies that all inspection results be reported to Fokker Services B.V., this AD does not include such a requirement.

Alternative Methods of Compliance

(c) In accordance with 14 CFR 39.19, the Manager, International Branch, ANM-116, FAA, Transport Airplane Directorate, is authorized to approve alternative methods of compliance for this AD.

Incorporation by Reference

(d) The actions shall be done in accordance with Fokker Service Bulletin SBF100-24-035, dated May 27, 2002. This incorporation by reference is approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Fokker Services B.V., P.O. Box 231, 2150 AE Nieuw-Vennep, the Netherlands. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, call (202) 741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

Note 2: The subject of this AD is addressed in Dutch airworthiness directive 2002-112, dated July 31, 2002.

Effective Date

(e) This amendment becomes effective on July 9, 2004.

Issued in Renton, Washington, on May 20, 2004.

Ali Bahrami,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-12397 Filed 6-3-04; 8:45 am]

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DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 2003-SW-32-AD; Amendment 39-13652; AD 2004-11-06]

RIN 2120-AA64

Airworthiness Directives; Agusta S.p.A. Model A109E Helicopters

AGENCY: Federal Aviation Administration, DOT.

ACTION: Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD) for the specified Agusta S.p.A. (Agusta) model helicopters that requires disabling certain windshield wipers and thereafter modifying the electrical system of the windshield wipers by installing a new resistor and condenser, eliminating incompatibility problems with the relays, and replacing the timed relay for certain windshield wiper kits. This amendment is prompted by testing that revealed overheating of the electrical resistor on the electrical system of the windshield wipers due to a system overload because of a partial incompatibility of new timed relays with the configuration of the windshield wiper electrical system. The actions specified by this AD are intended to prevent the incompatibility of certain relays with the windshield wiper electrical system, overheating of the resistor due to system overload, and an electrical fire.

DATES: Effective July 9, 2004.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of July 9, 2004.

ADDRESSES: The service information referenced in this AD may be obtained from Agusta, 21017 Cascina Costa di Samarate (VA) Italy, Via Giovanni Agusta 520, telephone 39 (0331) 229111, fax 39 (0331) 229605-222595. This information may be examined at the FAA, Office of the Regional Counsel, Southwest Region, 2601 Meacham Blvd., Room 663, Fort Worth, Texas; or at the National Archives and Records Administration (NARA). For information on the availability of this

material at NARA, call 202-741-6030, or go to: http://www.archives.gov/federal_register/code_of_federal_regulations/ibr_locations.html.

FOR FURTHER INFORMATION CONTACT:

Carroll Wright, Aviation Safety Engineer, FAA, Rotorcraft Directorate, Regulations and Guidance Group, Fort Worth, Texas 76193-0111, telephone (817) 222-5120, fax (817) 222-5961.

SUPPLEMENTARY INFORMATION:

A proposal to amend 14 CFR part 39 to include an AD for the specified model helicopters was published in the **Federal Register** on January 8, 2004 (69 FR 1274). That action proposed to require disabling certain windshield wipers and thereafter modifying the electrical system of the windshield wipers by installing a new resistor and condenser, eliminating incompatibility problems with the relays, and replacing the timed relay for certain windshield wiper kits.

Ente Nazionale per l'Aviazione Civile (ENAC), the airworthiness authority for Italy, notified the FAA that an unsafe condition may exist on Agusta Model 109E helicopters. ENAC advises modifying the electrical installation of some windshield wiper kits as stated in the manufacturer's service information.

Agusta has issued Alert Bollettino Tecnico No. 109EP-27, Revision A, dated February 7, 2003 (ABT), which specifies modifying the electrical installation of windshield wiper kit, part number (P/N) 109-0741-65, by installing kit, P/N 109-0823-13, to replace the existing resistor and condenser to eliminate functional malfunction when timed relays, P/N TDH-8070-1001P or T412-2006, are installed. During a ground functional test, overheating of the electrical resistor was found in the windshield wiper electrical system due to a system overload. An investigation revealed that the source of the overheating was a functional malfunction caused by a partial incompatibility of new timed relays with the actual configuration of the windshield wiper electrical system. ENAC classified the ABT as mandatory and issued AD No. 2003-032, dated February 10, 2003, to ensure the continued airworthiness of these helicopters in Italy.

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were received on the proposal or the FAA's determination of the cost to the public. With the exception of changing the ABT No. from 109SP-27 as shown in the "Discussion" section of the notice to the correct No.