### **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

### 14 CFR Part 39

[Docket No. 2001-NM-08-AD; Amendment 39-12341; AD 2001-15-07]

RIN 2120-AA64

Airworthiness Directives; Aerospatiale Model ATR42–200, –300, and –320 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT. **ACTION:** Final rule.

SUMMARY: This amendment adopts a new airworthiness directive (AD), applicable to certain Aerospatiale Model ATR42–200, –300, and –320 series airplanes, that requires modifying the wiring of the starting rotary switch. The actions specified by this AD are intended to prevent the loss of electrical power supply of the direct current (DC) emergency and standby buses, which could result in the loss of some electrical loads and the consequent display of erroneous information to the flight crew. This action is intended to address the identified unsafe condition.

DATES: Effective August 29, 2001.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of August 29, 2001

ADDRESSES: The service information referenced in this AD may be obtained from Aerospatiale, 316 Route de Bayonne, 31060 Toulouse, Cedex 03, France. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

# FOR FURTHER INFORMATION CONTACT:

Todd Thompson, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2110; fax (425) 227-1149.

# SUPPLEMENTARY INFORMATION: A

proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain Aerospatiale Model ATR42–200, –300, and –320 series airplanes was published in the **Federal Register** on April 25, 2001 (66 FR 20768). That action proposed to require modifying the wiring of the starting rotary switch.

### Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

### Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

## **Cost Impact**

The FAA estimates that 60 Model ATR42–200, –300, and –320 series airplanes of U.S. registry will be affected by this AD, that it will take approximately 4 work hours per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Required parts will be provided by the manufacturer at no cost to operators. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$14,400, or \$240 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

# **Regulatory Impact**

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy

of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

# List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

# Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

### §39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

**2001–15–07 Aerospatiale:** Amendment 39–12341. Docket 2001–NM–08–AD.

Applicability: Model ATR42–200, –300, and –320 airplanes, certificated in any category, except those modified in accordance with Modification 3047 or Avions de Transport Regional Service Bulletin ATR42–80–0001.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent the loss of electrical power supply of the direct current (DC) emergency and standby buses, which could result in the loss of some electrical loads and the consequent display of erroneous information to the flight crew, accomplish the following:

## Modification

(a) Within 6 months after the effective date of this AD, modify the wiring of the starting rotary switch, in accordance with Avions de Transport Regional Service Bulletin ATR42–80–0001, Revision 2, dated November 15, 2000.

# **Alternative Methods of Compliance**

(b) An alternative method of compliance or adjustment of the compliance time that

provides an acceptable level of safety may be used if approved by the Manager,

International Branch, ANM–116, Transport Airplane Directorate, FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM–116.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM-116.

### **Special Flight Permits**

(c) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

### Incorporation by Reference

(d) The modification shall be done in accordance with Avions de Transport Regional Service Bulletin ATR42–80–0001, Revision 2, dated November 15, 2000, which includes the specified list of effective pages in Table 1:

TABLE 1.—LIST OF EFFECTIVE PAGES

Page No.	Revision level shown on page	Date shown on page
1–6	2	November 15, 2000.
7, 9–13	Original	March 19, 1992.
8	1	May 11, 1993.

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Aerospatiale, 316 Route de Bayonne, 31060 Toulouse, Cedex 03, France. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Note 3:** The subject of this AD is addressed in French airworthiness directive 2000–454–081(B), dated November 15, 2000.

### **Effective Date**

(e) This amendment becomes effective on August 29, 2001.

Issued in Renton, Washington, on July 16, 2001.

# Donald L. Riggin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 01–18258 Filed 7–24–01; 8:45 am]

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### **DEPARTMENT OF TRANSPORTATION**

### **Federal Aviation Administration**

### 14 CFR Part 39

[Docket No. 2000-NM-366-AD; Amendment 39-12338; AD 2001-15-04]

RIN 2120-AA64

# Airworthiness Directives; Dornier Model 328–100 Series Airplanes

AGENCY: Federal Aviation Administration, DOT. ACTION: Final rule.

**SUMMARY:** This amendment supersedes an existing airworthiness directive (AD), applicable to all Dornier Model 328–100 series airplanes, that currently requires repetitive inspections of the left and right roll spoiler actuators to check for signs of leakage and deformation of the housing, repetitive inspections of the gap between the left roll spoiler actuator housing cap and the actuator housing, repetitive torque checks of the left roll spoiler actuator housing cap attachment screws, and corrective action, if necessary. This amendment requires replacement of the double shuttle valves in the roll spoiler actuators with new improved valves, which constitutes terminating action for the requirements of this AD. This amendment is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by this AD are intended to prevent oil leakage from the roll spoiler actuators, which could result in incorrect roll spoiler operation and reduced controllability of the airplane.

DATES: Effective August 29, 2001.

The incorporation by reference of
Dornier Service Bulletin SB-328-27310, dated June 10, 2000, as listed in the
regulations, is approved by the Director
of the Federal Register as of August 29,
2001.

The incorporation by reference of Dornier Alert Service Bulletin ASB–328–27–025, Revision 1, dated September 22, 1999, as listed in the regulations, was previously approved by the Director of the Federal Register as of December 7, 1999 (64 FR 59117, November 2, 1999).

ADDRESSES: The service information referenced in this AD may be obtained from Fairchild Dornier, Dornier Luftfahrt GmbH, P.O. Box 1103, D—82230 Wessling, Germany. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW.,

Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC. FOR FURTHER INFORMATION CONTACT: Tom

Groves, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-1503; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION: A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) by superseding AD 99-22-15, amendment 39-11393 (64 FR 59117, November 2, 1999), which is applicable to all Dornier Model 328-100 series airplanes, was published in the Federal **Register** on May 4, 2001 (66 FR 22486). The action proposed to continue to require repetitive inspections of the left and right roll spoiler actuators to check for signs of leakage and deformation of the housing, repetitive inspections of the gap between the left roll spoiler actuator housing cap and the actuator housing, repetitive torque checks of the left roll spoiler actuator housing cap attachment screws, and corrective action, if necessary. The action also proposed to add a requirement to replace the double shuttle valves in the roll spoiler actuators with new improved valves, which would constitute terminating action for the requirements of this AD.

### Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the single comment received.

The commenter, an operator of an airline, indicates that it has already met the proposed requirements.

### Conclusion

After careful review of the available data, including the comment noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

### **Cost Impact**

There are approximately 50 Dornier Model 328–100 series airplanes of U.S. registry that will be affected by this AD.

The repetitive inspections and checks that are currently required by AD 99–22–15 take approximately 3 work hours per airplane to accomplish, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of these inspections and checks on U.S. operators is estimated to be \$9,000, or \$180 per airplane.

The replacement that is added by this new AD will take approximately 1 work