

*Applicability:* Model 737-700IGW series airplanes modified by Supplemental Type Certificate (STC) ST09100AC-D, ST09104AC-D, ST09105AC-D, or ST09106AC-D; certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

*Compliance:* Required as indicated, unless accomplished previously.

To ensure that the flight crew is able to remove electrical power from the passenger entertainment system (PES) when necessary and is advised of appropriate procedures for such action, accomplish the following:

**Modification and Airplane Flight Manual Revision**

(a) Within 18 months after the effective date of this AD, do paragraphs (a)(1) and (a)(2) of this AD.

(1) Modify the PES by installing a power switch and a load-shed circuit breaker as well as associated relays and wiring, in accordance with Raytheon Service Bulletin 737IGW-24-1, dated August 11, 2000.

(2) Revise the Non-Normal Procedures Section of the FAA-approved Airplane Flight Manual (AFM) to advise the flight crew on using the power switch installed by paragraph (a)(1) to remove power from the PES in the event of an emergency related to smoke or fire, in accordance with Raytheon Airplane Flight Manual Supplement B737-700 IGW, dated October 16, 2000, which is supplied with Raytheon Service Bulletin 737IGW-24-1, dated August 11, 2000.

**Spares**

(b) As of the effective date of this AD, no person shall install a PES in accordance with STC ST09100AC-D, ST09104AC-D, ST09105AC-D, or ST09106AC-D on any airplane, unless it is modified and the AFM is revised in accordance with this AD.

**Alternative Methods of Compliance**

(c) An alternative method of compliance or adjustment of the compliance time that

provides an acceptable level of safety may be used if approved by the Manager, Fort Worth Airplane Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Fort Worth ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Fort Worth ACO.

**Special Flight Permits**

(d) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

**Incorporation by Reference**

(e) The actions shall be done in accordance with Raytheon Service Bulletin 737IGW-24-1, dated August 11, 2000; and Raytheon Airplane Flight Manual Supplement B737-700 IGW, dated October 16, 2000. Raytheon Airplane Flight Manual Supplement B737-700 IGW contains the following list of effective pages:

Page No.	Revision level shown on page	Date shown on page
List of Effective Pages: Page 1	Original .....	October 16, 2000.

This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Raytheon Systems Company, Intelligence Information and Aircraft Integration Systems, 7500 Maehre Road, Waco, Texas 76705. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at FAA, Fort Worth Airplane Certification Office, 2601 Meacham Blvd., Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Effective Date**

(f) This amendment becomes effective on August 16, 2001.

Issued in Renton, Washington, on July 3, 2001.

**Vi L. Lipski,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*  
[FR Doc. 01-17160 Filed 7-11-01; 8:45 am]

**BILLING CODE 4910-13-U**

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. 2000-NM-240-AD; Amendment 39-12322; AD 2001-14-11]

**RIN 2120-AA64**

**Airworthiness Directives; Boeing Model 747-100 and -200 Series Airplanes Modified by Supplemental Type Certificate SA8622SW**

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to all Boeing Model 747-100 and -200 series airplanes modified by Supplemental Type Certificate SA8622SW, that requires deactivation of the in-flight entertainment (IFE) system. This action is necessary to ensure that the flight crew is able to remove power from the IFE system when necessary. Inability to remove power from the IFE system during a non-normal or emergency situation could result in inability to control smoke or fumes in

the airplane flight deck or cabin. This action is intended to address the identified unsafe condition.

**DATES:** Effective August 16, 2001.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of August 16, 2001.

**ADDRESSES:** The service information referenced in this AD may be obtained from Continental Airlines, Inc., 600 Jefferson Street HQJAV, Houston, Texas 77002. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; at the FAA, Fort Worth Airplane Certification Office, 2601 Meacham Blvd., Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Ingrid Knox, Aerospace Engineer, ASW-150, FAA, Fort Worth Airplane Certification Office, 2601 Meacham Blvd., Fort Worth, Texas 76137-4298; telephone (817) 222-5139; fax (817) 222-5960.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Boeing Model 747-100 and -200 series airplanes modified by Supplemental Type Certificate SA8622SW was published in the **Federal Register** on March 2, 2001 (66 FR 13204). That action proposed to require deactivation of the in-flight entertainment system.

### Comments

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

### Conclusion

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

### Cost Impact

There are approximately 8 Model 747-100 and -200 series airplanes of the affected design in the worldwide fleet. The FAA estimates that 8 airplanes of U.S. registry will be affected by this AD, that it will take approximately 1 work hour per airplane to accomplish the required actions, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of this AD on U.S. operators is estimated to be \$480, or \$60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

### Regulatory Impact

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a

“significant regulatory action” under Executive Order 12866; (2) is not a “significant rule” under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

#### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

**2001-14-11 Boeing:** Amendment 39-12322. Docket 2000-NM-240-AD.

**Applicability:** Model 747-100 and -200 series airplanes modified by Supplemental Type Certificate (STC) SA8622SW, certificated in any category.

**Note 1:** This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To ensure that the flight crew is able to remove power from the in-flight entertainment (IFE) system when necessary; which, if not done during a non-normal or emergency situation, could result in inability to control smoke or fumes in the airplane

flight deck or cabin; accomplish the following:

### Deactivation

(a) Within 18 months after the effective date of this AD, deactivate the IFE system, in accordance with Continental Airlines Engineering Change/Repair Authorization 2330-02321, 2330-02322, 2330-02323, 2330-02324, 2330-02325, 2330-02326, 2330-02327, or 2330-02328; all dated August 29, 2000; as applicable.

### Spares

(b) As of the effective date of this AD, no person shall install an IFE system in accordance with STC SA8622SW on any airplane.

### Alternative Methods of Compliance

(c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Fort Worth Airplane Certification Office (ACO), FAA. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Fort Worth ACO.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Fort Worth ACO.

### Special Flight Permits

(d) Special flight permits may be issued in accordance with §§ 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

### Incorporation by Reference

(e) The actions shall be done in accordance with Continental Airlines Engineering Change/Repair Authorization 2330-02321, dated August 29, 2000; Continental Airlines Engineering Change/Repair Authorization 2330-02322, dated August 29, 2000; Continental Airlines Engineering Change/Repair Authorization 2330-02323, dated August 29, 2000; Continental Airlines Engineering Change/Repair Authorization 2330-02324, dated August 29, 2000; Continental Airlines Engineering Change/Repair Authorization 2330-02325, dated August 29, 2000; Continental Airlines Engineering Change/Repair Authorization 2330-02326, dated August 29, 2000; Continental Airlines Engineering Change/Repair Authorization 2330-02327, dated August 29, 2000; or Continental Airlines Engineering Change/Repair Authorization 2330-02328, dated August 29, 2000; as applicable. (Only the first page of these documents contains the document date; no other page contains this information.) This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Continental Airlines, Inc., 600 Jefferson Street HQJAV, Houston, Texas 77002. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; at the FAA, Fort

Worth Airplane Certification Office, 2601 Meacham Blvd., Fort Worth, Texas; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Effective Date**

(f) This amendment becomes effective on August 16, 2001.

Issued in Renton, Washington, on July 3, 2001.

**Vi L. Lipski,**

*Manager, Transport Airplane Directorate, Aircraft Certification Service.*

[FR Doc. 01-17159 Filed 7-11-01; 8:45 am]

BILLING CODE 4910-13-P

**DEPARTMENT OF TRANSPORTATION**

**Federal Aviation Administration**

**14 CFR Part 39**

[Docket No. 2000-NM-237-AD; Amendment 39-12321; AD 2001-14-10]

RIN 2120-AA64

**Airworthiness Directives; Boeing Model 737-300 Series Airplanes Modified by Supplemental Type Certificate ST00171SE**

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to all Boeing Model 737-300 series airplanes modified by Supplemental Type Certificate ST00171SE, that requires installation of a master switch to apply and remove power from the in-flight entertainment (IFE) system, and revision of the Airplane Flight Manual. This action is necessary to ensure that the flight crew is able to remove electrical power from the IFE system when necessary and is advised of appropriate procedures for such action. Inability to remove power from the IFE system during a non-normal or emergency situation could result in inability to control smoke or fumes in the airplane flight deck or cabin. This action is intended to address the identified unsafe condition.

**DATES:** Effective August 16, 2001.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of August 16, 2001.

**ADDRESSES:** The service information referenced in this AD may be obtained from Flight Structures, Inc., 4407 172nd Street NE., Arlington, Washington 98223. This information may be

examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:**

Stephen S. Oshiro, Aerospace Engineer, Systems and Equipment Branch, ANM-130S, FAA, Seattle Aircraft Certification Office, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2793; fax (425) 227-1181.

**SUPPLEMENTARY INFORMATION:**

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to all Boeing Model 737-300 series airplanes modified by Supplemental Type Certificate (STC) ST00171SE was published in the **Federal Register** on March 2, 2001 (66 FR 13198). That action proposed to require installation of a master switch to apply and remove power from the in-flight entertainment (IFE) system, and revision of the Airplane Flight Manual.

**Comments**

Interested persons have been afforded an opportunity to participate in the making of this amendment. No comments were submitted in response to the proposal or the FAA's determination of the cost to the public.

**Conclusion**

The FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

**Cost Impact**

None of the Model 737-300 series airplanes modified by STC ST00171SE are on the U.S. Register. The single airplane included in the applicability of this rule currently is operated by a non-U.S. operator under foreign registry; therefore, it is not directly affected by this AD action. However, the FAA considers that this rule is necessary to ensure that the unsafe condition is addressed in the event that the subject airplane is imported and placed on the U.S. Register in the future.

Should the affected airplane be imported and placed on the U.S. Register in the future, it will take approximately 24 work hours per airplane to accomplish the required actions, at an average labor rate of \$60 per work hour. Required parts will cost approximately \$1,500 per airplane. Based on these figures, the cost impact of this AD would be \$2,940 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of

the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

**Regulatory Impact**

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

**PART 39—AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

**§ 39.13 [Amended]**

2. Section 39.13 is amended by adding the following new airworthiness directive:

**2001-14-10 Boeing:** Amendment 39-12321. Docket 2000-NM-237-AD.