

**§ 39.13 [Amended]**

2. Section 39.13 is amended by adding a new airworthiness directive to read as follows:

**AD 99-22-12 Eurocopter France:**

Amendment 39-11390. Docket No. 98-SW-59-AD.

*Applicability:* Model AS332C, L, and L1 helicopters, certificated in any category.

**Note 1:** This AD applies to each helicopter identified in the preceding applicability provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For helicopters that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

*Compliance:* Required within 300 hours time-in-service (TIS) or within the next 3 calendar months, whichever occurs first, unless accomplished previously.

To prevent loss of electrical continuity, which could cause loss of critical systems, and subsequent loss of control of the helicopter, accomplish the following:

(a) Remove and replace each "CONNECTRAL" green electrical module that does not have a white dot on the face and that has a manufacturing code 95/16 through 96/21 engraved on a side, with an airworthy electrical module. Those manufacturing codes identify modules manufactured between the beginning of the 16th week of 1995 and the end of the 21st week of 1996.

**Note 1:** Eurocopter France Service Bulletin No. 01.00.51, dated May 4, 1998, pertains to the subject of this AD.

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Regulations Group, Rotorcraft Directorate, FAA. Operators shall submit their requests through an FAA Principal Maintenance Inspector, who may concur or comment and then send it to the Manager, Regulations Group.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Regulations Group.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the helicopter to a location where the requirements of this AD can be accomplished.

(d) This amendment becomes effective on November 30, 1999.

**Note 3:** The subject of this AD is addressed in Direction Generale De L'Aviation Civile (France) AD No. 98-254-070(A), dated July 1, 1998.

Issued in Fort Worth, Texas, on October 18, 1999.

**Eric Bries,**

*Acting Manager, Rotorcraft Directorate,  
Aircraft Certification Service.*

[FR Doc. 99-27791 Filed 10-25-99; 8:45 am]

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**DEPARTMENT OF TRANSPORTATION****Federal Aviation Administration****14 CFR Part 39**

[Docket No. 98-ANE-62-AD; Amendment 39-11388; AD 99-22-10]

RIN 2120-AA64

**Airworthiness Directives; General Electric Aircraft Engines CF34 Series Turbofan Engines**

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to certain General Electric Aircraft Engines CF34 series turbofan engines, that establishes new life limits for certain high pressure compressor (HPC) spools, stage 9 HPC disks, and rear HPC spools. This amendment is prompted by a cyclic life analysis using increased stress levels resulting from manufacturing discrepancies. The actions specified by this AD are intended to prevent HPC spool and disk cracking, which could result in an uncontained engine failure and damage to the aircraft.

**DATES:** Effective December 27, 1999.

**FOR FURTHER INFORMATION CONTACT:**

Eugene Triozzi, Aerospace Engineer, Engine Certification Office, FAA, Engine and Propeller Directorate, 12 New England Executive Park, Burlington, MA 01803-5299; telephone (781) 238-7148, fax (781) 238-7199.

**SUPPLEMENTARY INFORMATION:** A

proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to General Electric Aircraft Engines (GEAE) Models CF34-1A, -3A, -3A1, and -3A2 turbofan engines was published in the **Federal Register** on April 5, 1999 (64 FR 16364). That action proposed to require removal from service of forward HPC spools, part number (P/N) 6078T56P01; rear HPC spools, P/N 6087T01P03 and 6087T01P04; and stage 9 HPC disks, P/N 5087T46P01 or 5087T46P02. The affected parts must be removed prior to accumulating cycles in service beyond new, reduced cyclic life limits.

**Comment Received**

Interested persons have been afforded an opportunity to participate in the making of this amendment. Due consideration has been given to the comment received.

The commenter supports the rule as proposed.

After careful review of the available data, including the comment noted above, the FAA has determined that air safety and the public interest require the adoption of the rule as proposed.

**Economic Analysis**

There are approximately 600 engines of the affected design in the worldwide fleet. The FAA estimates that 28 engines installed on aircraft of US registry will be affected by the requirement within this AD to replace the forward spool. The FAA has calculated the prorated cost for forward spool replacements to be \$36,500 per engine, based on the estimated new part cost divided by the original life limit, multiplied by the number of cycles that will be reduced by the AD requirement. Therefore, the FAA estimates the total cost impact for replaced forward spools to be \$1,022,000.

The FAA estimates that 200 engines installed on aircraft of US registry will be affected by the requirement to replace the stage 9 disk. The FAA has calculated the prorated cost for stage 9 disk replacements to be \$3,500 per engine, based on the estimated new part cost divided by the original life limit, multiplied by the number of cycles that will be reduced by the AD requirement. The FAA estimates the total cost impact for replaced stage 9 disks to be \$700,000.

The FAA estimates that 300 engines installed on aircraft of US registry will be affected by the requirement to replace the rear spool. The FAA has calculated the prorated cost for rear spool replacements to be \$8,900 per engine, based on the new part cost divided by the original life limit, multiplied by the number of cycles that will be reduced by the AD requirement. Therefore, the FAA estimates the total cost impact for replaced rear spools to be \$2,670,000.

The FAA has determined that it will take no additional work hours per engine to remove affected components, as removal would take place at available opportunities. Based on these figures, the total cost impact of the AD on US operators is estimated to be \$4,392,000.

**Regulatory Impact**

The regulations adopted herein will not have substantial direct effects on the

States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this final rule does not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption **ADDRESSES**.

#### List of Subjects in 14 CFR Part 39

Air Transportation, Aircraft, Aviation safety, Safety.

#### Adoption of the Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

#### PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

**Authority:** 49 U.S.C. 106(g), 40113, 44701.

##### § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

##### 99-22-10 General Electric Aircraft

**Engines:** Amendment 39-11388. Docket 98-ANE-62-AD.

**Applicability:** General Electric Aircraft Engines (GEAE) Models CF34-1A, -3A, -3A1, and -3A2 turbofan engines, installed on but not limited to Canadair aircraft models CL-600-2A12, -2B16, and -2B19.

**Note 1:** This airworthiness directive (AD) applies to each engine identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For engines that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (b)

of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

**Compliance:** Required as indicated, unless accomplished previously.

To prevent high pressure compressor (HPC) spool and disk cracking, which could result in an uncontained engine failure and damage to the aircraft, accomplish the following:

(a) Remove from service the following HPC spools and disks prior to accumulating cycles in service beyond new, reduced cyclic life limits, and replace with a serviceable part, as follows:

(1) For forward HPC spools, part number (P/N) 6078T56P01, which have accumulated fewer than 6,000 cycles since new (CSN) on the effective date of this AD, remove prior to accumulating 6,000 CSN.

(2) For forward HPC spools, P/N 6078T56P01, which have accumulated 6,000 or more CSN on the effective date of this AD, remove at the next shop visit after the effective date of this AD, but prior to accumulating 12,000 CSN.

(3) For the purpose of this AD, engine shop visit is defined as engine disassembly that includes separation of the compressor section from the fan section front frame and from the combustion section combustion chamber frame.

(4) For stage 9 HPC disks, P/N 6087T01P03 or 6087T01P04, remove prior to accumulating 20,000 CSN.

(5) For rear HPC spools, P/N 5087T46P01 or 5087T46P02, remove prior to accumulating 17,000 CSN.

(b) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Engine Certification Office. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Engine Certification Office.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this airworthiness directive, if any, may be obtained from the Engine Certification Office.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the aircraft to a location where the requirements of this AD can be accomplished.

(d) This amendment becomes effective on December 27, 1999.

Issued in Burlington, Massachusetts, on October 18, 1999.

**David A. Downey,**

*Assistant Manager, Engine and Propeller Directorate, Aircraft Certification Service.*

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## DEPARTMENT OF TRANSPORTATION

### Federal Aviation Administration

#### 14 CFR Part 71

[Airspace Docket No. 99-AEA-09]

#### Establishment of Class E Airspace: York County, PA

**AGENCY:** Federal Aviation Administration (FAA) DOT.

**ACTION:** Final rule.

**SUMMARY:** This action establishes Class E airspace extending upward from the surface within a 6.5 mile radius of the York Airport (THV). The increased traffic at the airport and its capacity to accept flights operating under Instrument Flight Rules via a Standard Instrument Approach Procedure (SIAP) makes it desirable to establish Class E airspace designated as a surface area.

**EFFECTIVE DATE:** 0901 UTC, Nov. 5, 1999.

**FOR FURTHER INFORMATION CONTACT:** Mr. Francis Jordan, Airspace Specialist, Airspace Branch, AEA-520, Air Traffic Division, Eastern Region, Federal Aviation Administration, Federal Building #111, John F. Kennedy International Airport, Jamaica, New York 11430, telephone: (718) 553-4521.

#### SUPPLEMENTARY INFORMATION:

#### History

On August 4, 1999 a notice proposing to amend Part 71 of the Federal Aviation Regulations (14 CFR Part 71) to establish Class E airspace at York Airport, York County, PA was published in the **Federal Register** (64 FR 42301-42302). Controlled airspace extending upward from the surface is needed to accommodate the increased traffic. The notice proposed to establish controlled airspace extending upward from surface to contain IFR operations in controlled airspace during portions of the terminal operation and while transitioning between the enroute and terminal environments.

Interested parties were invited to participate in this rulemaking proceeding by submitting written comments on the proposal to the FAA. No comments to the proposal were received. The rule is adopted as proposed.

The coordinates for this airspace docket are based on North American Datum 83. E airspace areas designations for airspace extending upward from the surface are published in paragraph 6002 of FAA Order 7400.9G, dated September 1, 1999, and effective September 16, 1999, which is incorporated by reference in 14 CFR 71.1. The Class E airspace designation listed in this