2231-2233, 2237, 2239); sec. 201, 88 Stat. 1242, as amended (42 U.S.C. 5841), sec. 5, Pub. L. 101-575, 104 Stat. 2835 (42 U.S.C. 2243).

Sections 110.1(b)(2) and 110.1(b)(3) also issued under Pub. L. 96-92, 93 Stat. 710 (22 U.S.C. 2403). Section 110.11 also issued under sec. 122, 68 Stat. 939 (42 U.S.C. 2152) and secs. 54c and 57d., 88 Stat. 473, 475 (42 U.S.C. 2074). Section 110.27 also issued under sec. 309(a), Pub. L. 99-440. Section 110.50(b)(3) also issued under sec. 123. 92 Stat. 142 (42 U.S.C. 2153). Section 110.51 also issued under sec. 184, 68 Stat. 954, as amended (42 U.S.C. 2234). Section 110.52 also issued under sec. 186, 68 Stat. 955 (42 U.S.C. 2236). Sections 110.80-110.113 also issued under (5 U.S.C. 552, 554). Sections 110.130-110.135 also issued under (5 U.S.C. 553). Sections 110.2 and 110.42(a)(9) also issued under sec. 903, Pub. L. 102-496 (42 U.S.C. 2151 et seq.).

66. In § 110.2 the definition of Public Document Room is removed and new definitions of NRC Public Document Room and NRC Web site are added to read as follows:

§110.2 Definitions.

NRC Public Document Room means the facility at 2120 L Street, NW., Washington, DC where certain public records of the NRC that were made available for public inspection in paper or microfiche prior to the implementation of the NRC Agencywide Documents Access and Management System, commonly referred to as ADAMS, will remain available for public inspection. It is also the place where computer terminals are available to access the Electronic Reading Room component of ADAMS on the NRC Web site, http://www.nrc.gov, where copies can be made or ordered as set forth in § 9.35. The facility is staffed with reference librarians to assist the public in identifying and locating documents and in using the NRC Web site and ADAMS. The NRC Public Document Room is open from 7:45 am to 4:15 pm, Monday through Friday, except on Federal holidays. Reference service and access to documents may also be requested by telephone (202-634-3273 or 800-397-4209) between 8:30 am and 4:15 pm, or by e-mail (PDR@nrc.gov), fax (202–634–3343), or letter (NRC Public Document Room, LL-6, Washington, DC 20555-0001).

NRC Web site, http://www.nrc.gov, is the Internet uniform resource locator name for the Internet address of the Web site where NRC will ordinarily make available its public records for inspection.

*

67. In § 110.70, paragraph (d) is removed and paragraph (a) is revised to read as follows:

§110.70 Public notice of receipt of application.

(a) The Commission will notice the receipt of each license application for an export or import for which a specific license is required by making a copy available at the NRC Web site, http:// www.nrc.gov.

68. Section 110.71 is revised to read as follows:

§110.71 Notice of withdrawal of an application.

The Commission will notice the withdrawal of an application by making a copy available at the NRC Web site, http://www.nrc.gov.

69. In § 110.72, the section heading and introductory text are revised to read as follows:

§110.72 Public availability of documents.

Unless exempt from disclosure under part 9 of this chapter, the following documents pertaining to each license and license application for an import or export requiring a specific license under this part will be made available at the NRC Web site, http://www.nrc.gov, and/ or at the NRC Public Document Room:

70. In § 110.112, paragraph (b) is revised to read as follows:

§110.112 Reporter and transcript for an oral hearing.

(b) Except for any classified portions, transcripts will be made available at the NRC Web site, http://www.nrc.gov, and/ or at the NRC Public Document Room.

71. In § 110.113, paragraph (c) is revised to read as follows:

§110.113 Commission action.

(c) If the Commission considers information not in the hearing record in reaching its licensing decision, the hearing participants will be informed and, if not classified or otherwise privileged, the information will be made available at the NRC Web site, http:// www.nrc.gov, and furnished to the participants.

Dated at Rockville, Md. this 13 day of April, 1999.

For the Nuclear Regulatory Commission. A. J. Galante,

Chief Information Officer.

[FR Doc. 99-11246 Filed 05-06-99: 8:45 am] BILLING CODE 7590-01-P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-NM-151-AD]

RIN 2120-AA64

Airworthiness Directives; Saab Model **SAAB 2000 Series Airplanes**

AGENCY: Federal Aviation Administration, DOT.

ACTION: Supplemental notice of proposed rulemaking; reopening of

comment period.

SUMMARY: This document revises an earlier proposed airworthiness directive (AD), applicable to certain Saab Model SAAB 2000 series airplanes, that would have required repetitive inspections for excessive wear of the aileron control cables, cable guides, and cable pulleys located at the rear wing spars; and corrective actions, if necessary. That proposal also would have required repetitive replacement of the control cables and cable guides with new or serviceable components. That proposal was prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. This new action revises the proposed rule by expanding the applicability of the proposed rule to include additional airplanes. In addition, this new action provides for optional terminating action for the repetitive inspections. The actions specified by this new proposed AD are intended to detect and correct excessive wear on the aileron control cables, cable guides, and cable pulleys located at the rear wing spars, which could result in broken aileron control cables and consequent reduced controllability of the airplane.

DATES: Comments must be received by June 1, 1999.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 97-NM-151-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Saab Aircraft AB, SAAB Aircraft Product Support, S–581.88, Linköping, Sweden. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington. FOR FURTHER INFORMATION CONTACT: Norman B. Martenson, Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055–4056; telephone (425) 227–2110; fax (425) 227–1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 97–NM–151–AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 97-NM-151-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to add an airworthiness directive (AD), applicable to certain Saab Model SAAB 2000 series airplanes,

was published as a notice of proposed rulemaking (NPRM) in the Federal **Register** on February 9, 1998 (63 FR 6499). That NPRM would have required repetitive inspections for excessive wear of the aileron control cables, cable guides, and cable pulleys located at the rear wing spars, and corrective actions, if necessary. That NPRM also would have required repetitive replacement of the control cables and cable guides with new or serviceable components. That NPRM was prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. Such excessive wear, if not corrected, could result in broken aileron control cables and consequent reduced controllability of the airplane.

Actions Since Issuance of the NPRM

Since the NPRM was issued, the manufacturer has issued Saab Service Bulletin 2000–27–033, Revision 01, dated March 27, 1998. Revision 01 introduces no new actions but revises the effectivity to include additional airplanes and specifies that the modification specified by Service Bulletin 2000–27–037, described below, eliminates the need for the repetitive inspections specified by Service Bulletin 2000–27–033, Revision 01.

Also, the manufacturer has issued Saab Service Bulletin 2000–27–037, dated March 11, 1998, which describes procedures for modification of the aileron control system. The modification involves replacement of the aileron control cables with new, improved (carbon steel) cables, and replacement of the semi-glides (cable guides) with new improved cable guides.

The Luftfartsverket (LFV), which is the civil aviation authority for Sweden, has classified Service Bulletin 2000–27–033, Revision 01, as mandatory. Further, the LFV has approved Service Bulletin 2000–27–037 as optional terminating action for the repetitive inspections specified by Service Bulletin 2000–27–033, Revision 01. In addition, the LFV has issued Swedish airworthiness directive SAD No. 1–111R1, dated March 30, 1998, in order to ensure the continued airworthiness of these airplanes in Sweden.

Explanation of New Requirements of the Supplemental NPRM

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, this supplemental NPRM would require accomplishment of the actions

specified in Saab Service Bulletin 2000–27–033 described previously. The proposed AD also would provide for optional terminating action for the repetitive inspections and repetitive replacements.

Operators should note that, in consonance with the findings of the Luftfartsverket (LFV), which is the civil aviation authority for Sweden, the FAA has determined that the repetitive actions proposed by this AD can be allowed to continue in lieu of accomplishment of a terminating action. In making this determination, the FAA considers that, in this case, long-term continued operational safety will be adequately assured by accomplishing the repetitive inspections to detect discrepancies before they represent a hazard to the airplane.

Conclusion

Since these changes expand the scope of the originally proposed rule, the FAA has determined that it is necessary to reopen the comment period to provide additional opportunity for public comment.

Cost Impact

The FAA estimates that 3 airplanes of U.S. registry would be affected by this proposed AD.

It would take approximately 1 work hour per airplane to accomplish the proposed inspection, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the inspection proposed by this AD on U.S. operators is estimated to be \$180, or \$60 per airplane, per inspection cycle.

It would take approximately 8 work hours per airplane to accomplish the proposed replacement, at an average labor rate of \$60 per work hour. Required parts would be supplied by the manufacturer at no cost to the operators. Based on these figures, the cost impact of the inspection proposed by this AD on U.S. operators is estimated to be \$1,440, or \$480 per airplane, per replacement cycle.

The cost impact figures discussed above are based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Should an operator elect to perform the optional terminating modification, it would take approximately 8 work hours per airplane to accomplish, at an average labor rate of \$60 per work hour. Based on these figures, the cost impact of the modification proposed by this AD

on U.S. operators is estimated to be \$1,440, or \$480 per airplane.

Regulatory Impact

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Saab Aircraft AB: Docket 97-NM-151-AD.

Applicability: Model SAAB 2000 series airplanes, serial numbers 004 through 064 inclusive; except those airplanes on which Saab Aircraft AB Modification 6093 (reference Saab Service Bulletin 2000–27–037, dated March 11, 1998) has been installed; certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability

provision, regardless of whether it has been otherwise modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To detect and correct excessive wear on the aileron control cables, cable guides, and cable pulleys located at the rear wing spars, which could result in broken aileron control cables and consequent reduced controllability of the airplane, accomplish the following:

Repetitive Inspections

- (a) Inspect to detect discrepancies of the left-and right-hand aileron control cables, cable guides, and cable pulleys at the time specified in paragraph (a)(1) or (a)(2) of this AD, as applicable, in accordance with Saab Service Bulletin 2000–27–033, dated April 29, 1997, or Revision 01, dated March 27, 1998. Repeat the inspection thereafter at intervals not to exceed 500 flight hours. If any discrepancy is found during any inspection required by this AD, prior to further flight, perform corrective action in accordance with the service bulletin.
- (1) For airplanes on which Saab Modification 5784 has been installed: Inspect at the later of the times specified in paragraphs (a)(1)(i) and (a)(1)(ii) of this AD.
- (i) Prior to the accumulation of 1,800 total flight hours; or within 1,800 flight hours after accomplishment of the modification or replacement of any control cable; whichever occurs latest. Or
- (ii) Within 200 flight hours after the effective date of this AD.
- (2) For airplanes on which Saab Modification 5784 has not been installed: Inspect at the later of the times specified in paragraphs (a)(2)(i) and (a)(2)(ii) of this AD.
- (i) Prior to the accumulation of 3,200 total flight hours, or within 3,200 flight hours after replacement of any control cable, whichever occurs later. Or
- (ii) Within 200 flight hours after the effective date of this AD.

Note 2: Although the inspection schedules of this AD apply to both left-and right-hand wing cable systems, replacement of the cable, guide, or pulley on one wing only, prior to scheduled replacement, would result in subsequent staggered inspections for the components of the left-and right-hand cable systems.

Repetitive Replacements

(b) Replace the aileron control cables, cable guides, and cable pulleys with new or serviceable parts, as applicable; at the time

- specified in paragraph (b)(1) or (b)(2) of this AD, as applicable; in accordance with Saab Service Bulletin 2000–27–033, dated April 29, 1997, or Revision 01, dated March 27, 1998.
- (1) For airplanes on which Saab Modification 5784 has been installed: Replace at the later of the times specified in paragraphs (b)(1)(i) and (b)(1)(ii) of this AD. Thereafter, repeat the inspection required by paragraph (a) of this AD at the time specified in paragraph (a)(1); and replace the control cables and cable guides thereafter prior to the accumulation of 3,200 flight hours after replacement of any control cable.
- (i) Prior to the accumulation of 3,200 total flight hours; or within 3,200 flight hours after installation of the modification, or after replacement of any control cable; whichever occurs latest. Or
- (ii) Within 200 flight hours after the effective date of this AD.
- (2) For airplanes on which Modification 5784 has not been installed: Replace at the later of the times specified in paragraphs (b)(2)(i) and (b)(2)(ii) of this AD. Thereafter, repeat the inspections required by paragraph (a) of this AD at the time specified in paragraph (a)(2); and replace the control cables and cable guides thereafter prior to the accumulation of 6,200 flight hours following replacement of any control cable.
- (i) Prior to the accumulation of 6,200 total flight hours, or within 6,200 flight hours after replacement of any control cable, whichever occurs later. Or
- (ii) Within 200 flight hours after the effective date of this AD.

Optional Terminating Action

(c) Accomplishment of the modification of the aileron control system in accordance with Saab Service Bulletin 2000–27–037, dated March 11, 1998, constitutes terminating action for the requirements of this AD.

Alternative Methods of Compliance

(d) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, International Branch, ANM–116, FAA, Transport Airplane Directorate. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, International Branch, ANM–116.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the International Branch, ANM–116.

Special Flight Permits

(e) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Note 4: The subject of this AD is addressed in Swedish airworthiness directive SAD No. 1–111R1, dated March 30, 1998.

Issued in Renton, Washington, on April 30, 1999.

D. L. Riggin,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 99–11469 Filed 5–6–99; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 99-NM-98-AD]

RIN 2120-AA64

Airworthiness Directives; Empresa Brasileira de Aeronautica S.A. (EMBRAER) Model EMB-145 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking

(NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain EMBRAER Model EMB-145 series airplanes. This proposal would require a one-time ultrasonic inspection of the maneuvering actuator piston rod of the main landing gear (MLG) to ensure adequate wall thickness of the piston rods; and replacement of any discrepant piston rod with a new piston rod. This proposal is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by the proposed AD are intended to prevent failure of the maneuvering actuator piston rod of the MLG, which would impede retraction of the MLG and consequent reduced controllability of the airplane.

DATES: Comments must be received by June 7, 1999.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 99-NM-98-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Empresa Brasileira de Aeronautica S.A. (EMBRAER), P.O. Box 343—CEP 12.225, Sao Jose dos Campos—SP, Brazil. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the FAA, Small Airplane Directorate, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Boulevard, suite 450, Atlanta, Georgia.

FOR FURTHER INFORMATION CONTACT: Curtis A. Jackson, Aerospace Engineer, Airframe and Propulsion Branch, ACE– 117A, FAA, Small Airplane Directorate, Atlanta Aircraft Certification Office, One Crown Center, 1895 Phoenix Boulevard, suite 450, Atlanta, Georgia 30349; telephone (770) 703–6083; fax (770) 703–6097.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 99–NM–98–AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-114, Attention: Rules Docket No. 99-NM-98-AD, 1601 Lind Avenue, SW., Renton, Washington 98055-4056.

Discussion

The Departmento de Aviacao Civil (DAC), which is the airworthiness authority for Brazil, recently notified the FAA that an unsafe condition may exist on certain EMBRAER Model EMB–145

series airplanes. The DAC advises that that it has received a report indicating that, due to a deviation in a manufacturing process, certain maneuvering actuator piston rods for the main landing gear (MLG) may have been delivered with reduced wall thickness. This condition, if not corrected, would result in failure of the maneuvering actuator piston rod of the MLG, which would impede retraction of the MLG and consequent reduced controllability of the airplane.

Explanation of Relevant Service Information

EMBRAER has issued Service Bulletin 145–32–0031, Change No. 01, dated December 8, 1998, and Change No. 02, dated February 12, 1999, which describe procedures for a one-time ultrasonic inspection of the maneuvering actuator piston rods of the MLG to ensure adequate wall thickness of the piston rods; and replacement of any discrepant piston rod with a new piston rod.

The DAC classified this service bulletin as mandatory and issued Brazilian airworthiness directive 98–09–01 R1, dated March 15, 1999, in order to assure the continued airworthiness of these airplanes in Brazil.

FAA's Conclusions

This airplane model is manufactured in Brazil and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DAC has kept the FAA informed of the situation described above. The FAA has examined the findings of the DAC, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, the proposed AD would require a one-time ultrasonic inspection of the maneuvering actuator piston rod of the MLG to ensure adequate wall thickness of the piston rods; and replacement of any discrepant piston rod with a new piston rod. The actions would be required to be accomplished in accordance with the service bulletin described previously.