The FAA is proposing a calendar compliance time instead of hours timein-service (TIS) because the service history on the U.S.-registered Alexander Schleicher Model ASK-21 sailplanes does not warrant a need for immediate compliance. Also, the average monthly usage of the affected sailplanes varies throughout the fleet. For example, one owner may operate the sailplane 25 hours TIS in one week, while another operator may operate the sailplane 25 hours TIS in one year. In order to ensure that all of the affected sailplanes have been inspected for displacement of the plastic S-shaped rudder tube and any displacement has been corrected within a reasonable amount of time, the FAA is proposing a compliance time of 6 calendar months.

## **Cost Impact**

The FAA estimates that 30 sailplanes in the U.S. registry would be affected by the proposed AD, that it would take approximately 2 workhours per sailplane to accomplish the proposed action, and that the average labor rate is approximately \$60 an hour. Parts cost approximately \$5 (for glue) per sailplane. Based on these figures, the total cost impact of the proposed AD on U.S. operators is estimated to be \$3,750 or \$125 per sailplane.

## **Regulatory Impact**

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a ''significant rule'' under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

## **List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Safety.

## **The Proposed Amendment**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

## § 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

**Alexander Schliecher:** Docket No. 97–CE–104–AD.

Applicability: Model ASK–21 sailplanes (serial numbers 21001 through 21345), certificated in any category.

Note 1: This AD applies to each sailplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For sailplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 6 calendar months after the effective date of this AD, unless already accomplished.

To prevent rudder control jamming, which, if not corrected, could result in loss of directional control of the sailplane, accomplish the following:

(a) Inspect the plastic S-shaped rudder pedal tube for displacement in accordance with the Actions sections 1.1, 1.2, and 1.3 of Alexander Schleicher Technical Note No. 20, dated October 16, 1987.

(b) If there is any displacement of the plastic S-shaped rudder pedal tube, prior to further flight, correct the placement in accordance with the Actions sections 1.1, 1.2, and 1.3 of Alexander Schleicher Technical Note No. 20, dated October 16, 1987.

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the sailplane to a location where the requirements of this AD can be accomplished.

(d) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Small Airplane Directorate, Aircraft Certification Service, 1201 Walnut, suite 900, Kansas City, Missouri 64106. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Small Airplane Directorate.

(e) Questions or technical information related to Alexander Schleicher Technical Note No. 20, dated October 16, 1987, should be directed to Alexander Schleicher, Segelflugzeugbau, 6416 Poppenhausen, Wasserkuppe, Federal Republic of Germany. This service information may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

**Note 3:** The subject of this AD is addressed in German AD No. 88–2 Schleicher, dated January 18, 1988.

Issued in Kansas City, Missouri, on February 5, 1998.

### John R. Colomy,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98-3519 Filed 2-11-98; 8:45 am] BILLING CODE 4910-13-P

## **DEPARTMENT OF TRANSPORTATION**

## Federal Aviation Administration

### 14 CFR Part 39

[Docket No. 97-CE-103-AD]

RIN 2120-AA64

# Airworthiness Directives; Alexander Schleicher Model ASK-21 Sailplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Notice of proposed rulemaking (NPRM).

**SUMMARY:** This document proposes to adopt a new airworthiness directive (AD) that would apply to Alexander Schleicher Model ASK-21 sailplanes that have certain modifications installed. The proposed action would require changing the sailplane flight manual's weight and balance information. The proposed AD is the result of mandatory continuing airworthiness information (MCAI) issued by the airworthiness authority for Germany. The actions specified by the proposed AD are intended to prevent the operator from using inaccurate weight and balance information provided in the sailplane flight manual (SFM), which, if not corrected, could lead to hazardous flight conditions.

**DATES:** Comments must be received on or before March 16, 1998.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97–CE–103–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

Service information that applies to the proposed AD may be obtained from Alexander Schleicher, Segelflugzeugbau, 6416 Poppenhausen, Wasserkuppe, Federal Republic of Germany; telephone 49.6658.890 or 49.6658.8920; facsimile: 49.6658.8923 or 49.6658.8940. This information also may be examined at the Rules Docket at the address above.

FOR FURTHER INFORMATION CONTACT: Mr. J. Mike Kiesov, Project Officer, Sailplanes/Gliders, Small Airplane Directorate, Aircraft Certification Service, FAA, 1201 Walnut, suite 900, Kansas City, Missouri 64106; telephone (816) 426–6932; facsimile (816) 426–2169.

# SUPPLEMENTARY INFORMATION:

### Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications should identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report that summarizes each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket No. 97–CE–103–AD." The postcard will be date stamped and returned to the commenter.

## **Availability of NPRMs**

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 97–CE–103–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

#### Discussion

The Luftfahrt-Bundesamt (LBA), which is the airworthiness authority for the Federal Republic of Germany, recently notified the FAA that an unsafe condition may exist on Alexander Schleicher Model ASK-21 sailplanes that are equipped with certain modification provided in Alexander Schleicher technical note (TN) 3 or TN 7. These technical notes are optional modifications that include the procedures for installing a removable ballast in the sailplane. The LBA reports that the conversion of kilograms to pounds on the trim weights of the Alexander Schleicher Model ASK-21 sailplane was miscalculated at the factory for the sailplanes with the modifications incorporated. Therefore, the U.S.-FAA version of the trim weights information provided in the SFM may lead to loading the sailplane outside the center of gravity. These conditions, if not corrected, could result in hazardous flight conditions.

# **Relevant Service Information**

Alexander Schleicher has issued Technical Note No. 13a, dated June 4, 1984, which specifies procedures for changing the sailplane flight manual (SFM) by removing page 2 (dated May 16, 1984) and page 13 (dated February 16, 1984), and inserting a new page 2 and page 13, both dated June 4, 1984.

The LBA classified this service bulletin as mandatory and issued German AD 84–32/2 Schleicher, dated June 12, 1984, in order to assure the continued airworthiness of these sailplanes in Germany.

## The FAA's Determination

The Alexander Schleicher Model ASK–21 sailplane is manufactured in Germany and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the LBA has kept the FAA informed of the situation described above.

The FAA has examined the findings of the LBA, reviewed all available information, including the service information referenced above, and determined that AD action is necessary

for products of this type design that are certificated for operation in the United States.

# **Explanation of the Provisions of the Proposed AD**

Since an unsafe condition has been identified that is likely to exist or develop in other Alexander Schleicher Model ASK–21 sailplanes of the same type design registered in the United States, the proposed AD would require changing the SFM by removing and replacing two pages referencing the trim weight information. Accomplishment of the proposed installation would be in accordance with the Action section of Alexander Schleicher Technical Note No. 13a, dated June 4, 1984.

## **Cost Impact**

The FAA estimates that 30 sailplanes in the U.S. registry would be affected by the proposed AD, that it would take approximately 1 workhour per sailplane to accomplish the proposed action, and that the average labor rate is approximately \$60 an hour. There are no parts required for the proposed action. The proposed action may be performed by the owner/operator holding at least a private pilot certificate as authorized by section 43.7 of the Federal Aviation Regulations (14 CFR 43.7), and must be entered into the aircraft records showing compliance with this AD in accordance with section 43.9 of the Federal Aviation Regulations (14 CFR 43.9). Based on these figures, there is no cost impact of the proposed AD on U.S. operators.

## **Regulatory Impact**

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action has been placed in the Rules

Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

## List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

## The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

## § 39.13 [Amended]

2. Section 39.13 is amended by adding a new airworthiness directive (AD) to read as follows:

**Alexander Schleicher:** Docket No. 97–CE–103–AD.

Applicability: Model ASK-21 sailplanes, all serial numbers, certificated in any category, that are equipped with the modifications in Alexander Schleicher Technical Note (TN) 3 or TN 7.

Note 1: This AD applies to each sailplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For sailplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (d) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required within the next 3 calendar months after the effective date of this AD, unless already accomplished.

To prevent the operator from using inaccurate weight and balance information provided in the sailplane flight manual (SFM), which, if not corrected, could lead to hazardous flight conditions, accomplish the following:

(a) Remove page 2 (dated May 16, 1984) and page 13 (dated February 16, 1984) from the Alexander Schleicher Model ASK–21 SFM, and replace these pages with new pages 2 and 13, both dated June 4, 1984, in accordance with Alexander Schleicher ASK–21 Technical Note No. 13a, dated June 4, 1984.

(b) Incorporating the SFM revisions, as required by this AD, may be performed by the owner/operator holding at least a private pilot certificate as authorized by section 43.7

of the Federal Aviation Regulations (14 CFR 43.7), and must be entered into the aircraft records showing compliance with this AD in accordance with section 43.9 of the Federal Aviation Regulations (14 CFR 43.9).

(c) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the sailplane to a location where the requirements of this AD can be accomplished.

(d) An alternative method of compliance or adjustment of the compliance time that provides an equivalent level of safety may be approved by the Manager, Small Airplane Directorate, Aircraft Certification Service, 1201 Walnut, suite 900, Kansas City, Missouri 64106. The request shall be forwarded through an appropriate FAA Maintenance Inspector, who may add comments and then send it to the Manager, Small Airplane Directorate.

**Note 2:** Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Small Airplane Directorate.

(e) Questions or technical information related to Alexander Schleicher Technical Note No. 13a, dated June 4, 1984, should be directed to Alexander Schleicher, Segelflugzeugbau, 6416 Poppenhausen, Wasserkuppe, Federal Republic of Germany; telephone 49.6658.890 or 49.6658.8920; facsimile: 49.6658.8923 or 49.6658.8940. This service information may be examined at the FAA, Central Region, Office of the Regional Counsel, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106.

**Note 3:** The subject of this AD is addressed in German AD No. 84–32/2 Schleicher, dated June 12, 1984.

Issued in Kansas City, Missouri, on February 4, 1998.

### John R. Colomy,

Acting Manager, Small Airplane Directorate, Aircraft Certification Service.

[FR Doc. 98–3518 Filed 2–11–98; 8:45 am] BILLING CODE 4910–13–P

## **DEPARTMENT OF TRANSPORTATION**

## **Federal Aviation Administration**

14 CFR Part 39

[Docket No. 95-CE-35-AD]

RIN 2120-AA64

Airworthiness Directives; The New Piper Aircraft, Inc. (Formerly Piper Aircraft Corporation) Models PA-23 (PA-23-150), PA-23-160, PA-23-235, and PA-23-250 Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Supplemental notice of proposed rulemaking (NPRM); reopening of the comment period.

**SUMMARY:** This document proposes to revise an earlier proposed airworthiness

directive (AD) that would have required installing external fuel ramp assemblies on The New Piper Aircraft, Inc. (Piper) Models PA-23 (PA-23-150), PA-23-160, PA-23-235, and PA-23-250 airplanes, and incorporating pilots' operating handbook (POH) revisions for Piper Models PA-23 (PA-23-150), and PA-23-160 airplanes. That proposed AD would have superseded AD 92-13-04, which currently requires preflight draining procedures on Piper Models PA-23 (PA-23-150) and PA-23-160 airplanes. The proposed AD was the result of reports of water-in-the-fuel on the affected airplanes, even on those where the airplane owners/operators had accomplished preflight draining procedures. The actions specified in the proposed AD are intended to assist in eliminating water in the fuel tanks, which could result in rough engine operation or complete loss of engine power. Comments received on the proposal specify an additional alternative to the proposed AD, and the Federal Aviation Administration (FAA) has determined that this alternative should be added to the proposal. Based upon these comments on the original proposal and the amount of time that has elapsed since issuance of this proposal, the FAA has determined that the comment period for the proposal should be reopened and the public should have additional time to comment.

**DATES:** Comments must be received on or before April 13, 1998.

ADDRESSES: Submit comments in triplicate to the FAA, Central Region, Office of the Regional Counsel, Attention: Rules Docket No. 95–CE–35–AD, Room 1558, 601 E. 12th Street, Kansas City, Missouri 64106. Comments may be inspected at this location between 8 a.m. and 4 p.m., Monday through Friday, holidays excepted.

FFC Engineering Specification 2810– 002, Revision A, dated March 21, 1995, may be obtained from Floats & Fuel Cells, 4010 Pilot Drive, suite 3, Memphis, Tennessee 38118. Piper Service Bulletin (SB) No. 827A, dated November 4, 1988, may be obtained from The New Piper Aircraft, Inc., Customer Services, 2926 Piper Drive, Vero Beach, Florida 32960. The instructions included with Transamerican Enterprises, Inc. TAE102688 Piper PA 23 Fuel Cell Drain Installation, dated September 30, 1996, may be obtained from Transamerican Enterprises, Inc., 6778 Skyline Drive, Delray Beach, Florida 33446. This information also may be examined at the Rules Docket at the address above.