List of Subjects in 10 CFR Part 55

Criminal penalties, Manpower training programs, Nuclear power plants and reactors, Reporting and recordkeeping requirements.

For the reasons set out in the preamble and under the authority of the Atomic Energy Act of 1954, as amended; the Energy Reorganization Act of 1974, as amended; and 5 U.S.C. 553; the NRC proposes to adopt the following amendments to 10 CFR part 55.

PART 55—OPERATOR'S LICENSES

1. The authority citation for part 55 continues to read as follows:

Authority: Secs. 107, 161, 182, 68 Stat. 939, 948, 953, as amended, sec. 234, 83 Stat. 444, as amended (42 U.S.C. 2137, 2201, 2232, 2282); secs. 201, as amended, 202, 88 Stat. 1242, as amended, 1244 (42 U.S.C. 5841, 5842)

Sections 55.41, 55.43, 55.45, and 55.59 also issued under sec. 306, Pub. L. 97–425, 96 Stat. 2262 (42 U.S.C. 10226). Section 55.61 also issued under secs. 186, 187, 68 Stat. 955 (42 U.S.C. 2236, 2237).

2. In § 55.8 paragraph (b) is revised to read as follows:

§ 55.8 Information collection requirements; OMB approval.

* * * * *

(b) The approved information collection requirements contained in this part appear in §§ 55.31, 55.40, 55.45, 55.53, and 55.59.

3. A new § 55.40 is added to read as follows:

§ 55.40 Implementation.

- (a) Power reactor facility licensees shall —
- Prepare the required site-specific written examinations and operating tests;
- (2) Submit the written examinations and operating tests to the Commission for review and approval; and
- (3) Proctor and grade the NRCapproved site-specific written examinations.
- (b) In lieu of requiring a specific power reactor facility licensee to prepare the examinations and tests or to proctor and grade the site-specific written examinations, the Commission may elect to perform those tasks.
- (c) The Commission will prepare and administer the written examinations and operating tests at non-power reactor facilities.

Dated at Rockville, MD. this 31st day of July, 1997.

For the Nuclear Regulatory Commission. **John C. Hoyle**,

Secretary of the Commission.
[FR Doc. 97–20645 Filed 8–6–97; 8:45 am]
BILLING CODE 7590–01–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-NM-167-AD]

RIN 2120-AA64

Airworthiness Directives; Airbus Model A320 and A321 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking

(NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to certain Airbus Model A320 and A321 series airplanes. This proposal would require a one-time inspection for discrepancies of the release cable of the forward and rear passenger doors, and replacement of any discrepant release cable with a new release cable. This proposal is prompted by the issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by the proposed AD are intended to prevent failure of the passenger door to open and consequent inability of the slide/slide raft to deploy, which could delay or impede passengers when exiting the airplane during an emergency.

DATES: Comments must be received by September 16, 1997.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–103, Attention: Rules Docket No. 97–NM–167–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Airbus Industrie, 1 Rond Point Maurice Bellonte, 31707 Blagnac Cedex, France. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Charles Huber, Aerospace Engineer,

Standardization Branch, ANM-113, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2589; fax (425) 227-1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the Rules Docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the Rules Docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the Rules Docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 97–NM–167–AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM-103, Attention: Rules Docket No. 97–NM-167–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056.

Discussion

The Direction Générale de l'Aviation Civile (DGAC), which is the airworthiness authority for France, recently notified the FAA that an unsafe condition may exist on certain Airbus Model A320 and A321 series airplanes. The DGAC advises that, during a routine deployment of the aft right-hand passenger door slide, the passenger door failed to open fully. Investigation revealed that the attachment ball nipple of the release cable detached from the cable end due to a production process error. Failure of the passenger door to open could result in the inability to deploy the slide/slide raft. This

condition, if not corrected, could delay or impede passengers when exiting the airplane during an emergency.

Explanation of Relevant Service Information

Airbus has issued All Operators Telex (AOT) 25–12, Revision 1, dated March 21, 1996, which describes procedures for a one-time inspection for discrepancies of the release cable of the forward and rear passenger doors, and replacement of any discrepant cable with a new cable. The DGAC classified this AOT as mandatory and issued French airworthiness directive 96–171–083 (B), dated August 28, 1996, in order to assure the continued airworthiness of these airplanes in France.

FAA's Conclusions

These airplane models are manufactured in France and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DGAC has kept the FAA informed of the situation described above. The FAA has examined the findings of the DGAC, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, the proposed AD would require accomplishment of the actions specified (with one exception) in the AOT described previously.

Operators should note that, although the AOT describes procedures to declare a discrepant slide/slide raft inoperative in accordance with the Minimum Equipment List (MEL) requirements, this AD specifically requires that any discrepant cable must be replaced prior to further flight. Where there are differences between this AD and the AOT, the AD prevails.

Cost Impact

The FAA estimates that 132 airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 1 work hour per airplane to accomplish the proposed actions, and that the average labor rate is \$60 per work hour. Based on these figures, the cost impact of the proposed AD on U.S.

operators is estimated to be \$7,920, or \$60 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.

Regulatory Impact

The regulations proposed herein would not have substantial direct effects on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, in accordance with Executive Order 12612, it is determined that this proposal would not have sufficient federalism implications to warrant the preparation of a Federalism Assessment.

For the reasons discussed above, I certify that this proposed regulation (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) if promulgated, will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A copy of the draft regulatory evaluation prepared for this action is contained in the Rules Docket. A copy of it may be obtained by contacting the Rules Docket at the location provided under the caption ADDRESSES.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Safety.

The Proposed Amendment

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration proposes to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

PART 39—AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

Airbus Industrie: Docket 97-NM-167-AD.

Applicability: Model A320 and A321 series airplanes, as specified in French airworthiness directive 96–171–083 (B),

dated August 28, 1996, certificated in any category.

Note 1: This AD applies to each airplane identified in the preceding applicability provision, regardless of whether it has been modified, altered, or repaired in the area subject to the requirements of this AD. For airplanes that have been modified, altered, or repaired so that the performance of the requirements of this AD is affected, the owner/operator must request approval for an alternative method of compliance in accordance with paragraph (c) of this AD. The request should include an assessment of the effect of the modification, alteration, or repair on the unsafe condition addressed by this AD; and, if the unsafe condition has not been eliminated, the request should include specific proposed actions to address it.

Compliance: Required as indicated, unless accomplished previously.

To prevent the inability of the slide/slide raft to deploy due to a failure of the passenger door, which could delay or impede passengers when exiting the airplane during an emergency, accomplish the following:

(a) Within 500 flight hours after the effective date of this AD, perform a detailed inspection of each release cable at the left-and right-hand side of doors 1 and 4 for any discrepancy, in accordance with Airbus All Operators Telex (AOT) 25–12, Revision 1, dated March 21, 1996. If any discrepancy is found, prior to further flight, replace the release cable in accordance with the AOT.

Note 2: This AD supersedes any relief provided by the Master Minimum Equipment List (MMEL).

- (b) As of the effective date of this AD, no person shall install a release cable, part number C37103–101 or C37103–103, on any airplane unless the release cable has been inspected to detect any discrepancy in accordance with Airbus All Operators Telex (AOT) 25–12, Revision 1, dated March 21, 1996. If any discrepancy is detected in accordance with the AOT, that release cable shall not be installed.
- (c) An alternative method of compliance or adjustment of the compliance time that provides an acceptable level of safety may be used if approved by the Manager, Standardization Branch, ANM–113. Operators shall submit their requests through an appropriate FAA Principal Maintenance Inspector, who may add comments and then send it to the Manager, Standardization Branch, ANM–113.

Note 3: Information concerning the existence of approved alternative methods of compliance with this AD, if any, may be obtained from the Standardization Branch, ANM–113.

(d) Special flight permits may be issued in accordance with sections 21.197 and 21.199 of the Federal Aviation Regulations (14 CFR 21.197 and 21.199) to operate the airplane to a location where the requirements of this AD can be accomplished.

Issued in Renton, Washington, on July 31, 1997.

Darrell M. Pederson,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service. [FR Doc. 97–20730 Filed 8–6–97; 8:45 am] BILLING CODE 4910–13–P

DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. 97-NM-162-AD]

RIN 2120-AA64

Airworthiness Directives; Construcciones Aeronauticas, S.A. (CASA) Model CN–235 Series Airplanes

AGENCY: Federal Aviation Administration, DOT.

ACTION: Notice of proposed rulemaking

(NPRM).

SUMMARY: This document proposes the adoption of a new airworthiness directive (AD) that is applicable to all CASA Model CN-235 series airplanes. This proposal would require installation of a contactor and relocation of the existing fuse in the battery circuit. This proposal is prompted by issuance of mandatory continuing airworthiness information by a foreign civil airworthiness authority. The actions specified by the proposed AD are intended to prevent failure of the battery circuit due to a burned fuse, and consequent inability to restart the engine using batteries during flight. DATES: Comments must be received by

DATES: Comments must be received by September 16, 1997.

ADDRESSES: Submit comments in triplicate to the Federal Aviation Administration (FAA), Transport Airplane Directorate, ANM–103, Attention: Rules Docket No. 97–NM–162–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056. Comments may be inspected at this location between 9:00 a.m. and 3:00 p.m., Monday through Friday, except Federal holidays.

The service information referenced in the proposed rule may be obtained from Construcciones Aeronauticas, S.A., Getafe, Madrid, Spain. This information may be examined at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington.

FOR FURTHER INFORMATION CONTACT: Tim Backman, Aerospace Engineer, Standardization Branch, ANM-113, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton,

Washington 98055–4056; telephone (425) 227–2797; fax (425) 227–1149.

SUPPLEMENTARY INFORMATION:

Comments Invited

Interested persons are invited to participate in the making of the proposed rule by submitting such written data, views, or arguments as they may desire. Communications shall identify the rules docket number and be submitted in triplicate to the address specified above. All communications received on or before the closing date for comments, specified above, will be considered before taking action on the proposed rule. The proposals contained in this notice may be changed in light of the comments received.

Comments are specifically invited on the overall regulatory, economic, environmental, and energy aspects of the proposed rule. All comments submitted will be available, both before and after the closing date for comments, in the rules docket for examination by interested persons. A report summarizing each FAA-public contact concerned with the substance of this proposal will be filed in the rules docket.

Commenters wishing the FAA to acknowledge receipt of their comments submitted in response to this notice must submit a self-addressed, stamped postcard on which the following statement is made: "Comments to Docket Number 97–NM–162–AD." The postcard will be date stamped and returned to the commenter.

Availability of NPRMs

Any person may obtain a copy of this NPRM by submitting a request to the FAA, Transport Airplane Directorate, ANM–103, Attention: Rules Docket No. 97–NM–162–AD, 1601 Lind Avenue, SW., Renton, Washington 98055–4056.

Discussion

The Dirección General de Aviación Civil (DGAC), which is the airworthiness authority for Spain, recently notified the FAA that an unsafe condition may exist on all CASA Model CN-235 series airplanes. The DGAC advises that during a flight test performed by the manufacturer the flight crew intentionally shut an engine down, but were unable to restart the engine by using batteries. Investigation revealed that the batteries failed because a fuse had burned. This condition, if not corrected, could result in the inability to restart the engine with the batteries during flight.

Explanation of Relevant Service Information

CASA has issued Service Bulletin SB-235-24-07M, dated June 4, 1995; and Revision 1, dated January 25, 1996, which describe procedures for installation and relocation of a contactor in the battery circuit to allow for an alternate current path in the event of a fuse failure in the battery circuit. The DGAC classified these service bulletins as mandatory and issued Spanish airworthiness directive 09/96, dated December 9, 1996, in order to assure the continued airworthiness of these airplanes in Spain.

FAA's Conclusions

This airplane model is manufactured in Spain and is type certificated for operation in the United States under the provisions of section 21.29 of the Federal Aviation Regulations (14 CFR 21.29) and the applicable bilateral airworthiness agreement. Pursuant to this bilateral airworthiness agreement, the DGAC has kept the FAA informed of the situation described above. The FAA has examined the findings of the DGAC, reviewed all available information, and determined that AD action is necessary for products of this type design that are certificated for operation in the United States.

Explanation of Requirements of Proposed Rule

Since an unsafe condition has been identified that is likely to exist or develop on other airplanes of the same type design registered in the United States, the proposed AD would require accomplishment of the actions specified in the service bulletins described previously.

Cost Impact

The FAA estimates that 2 CASA Model CN–235 series airplanes of U.S. registry would be affected by this proposed AD, that it would take approximately 58 work hours per airplane to accomplish the proposed actions, and that the average labor rate is \$60 per work hour. Required parts would cost approximately \$2,000 per airplane. Based on these figures, the cost impact of the proposed AD on U.S. operators is estimated to be \$10,960, or \$5,480 per airplane.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the proposed requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted.